



National Aeronautics and
Space Administration

Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California



Radar Measurements for Earth Surface Change

October 29, 2009

Paul A Rosen

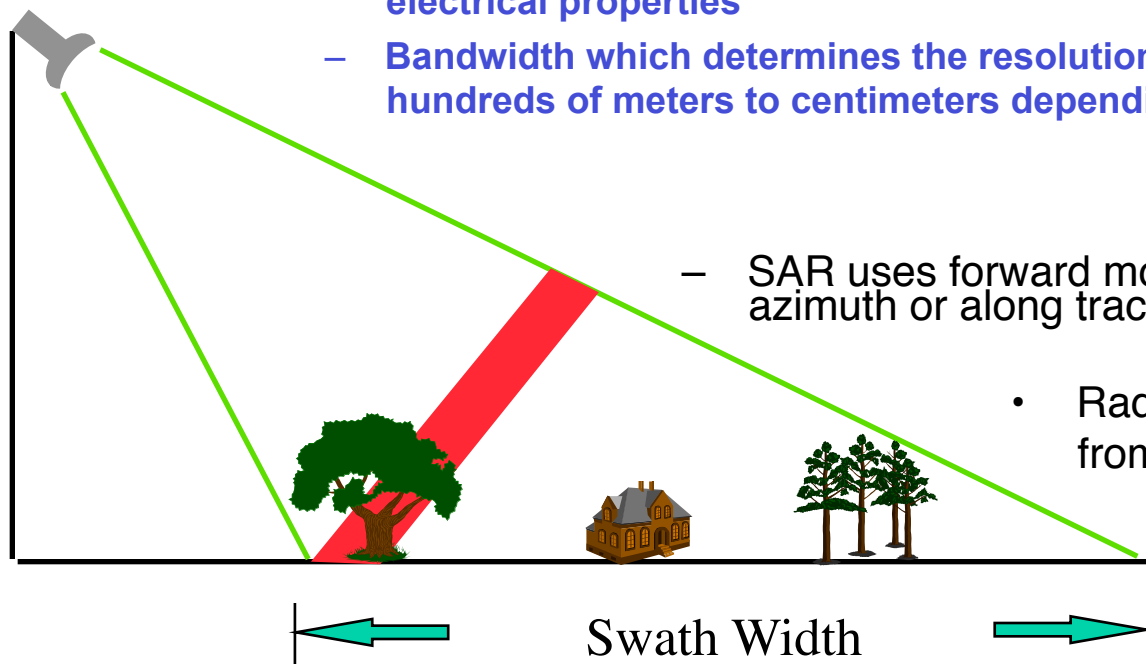
Radar Science and Engineering Section 334
Jet Propulsion Laboratory
California Institute of Technology



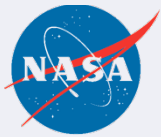


Radar Remote Sensing

- Radar works by emitting microwave radiation from an antenna and recording the energy reflected from objects in the field of view
- The returned radar signal depends on several radar parameters
 - Frequency (wavelength) — radar is most sensitive to objects larger than 1/10 of the radar wavelength. Electrical properties of objects change depending on their material composition and on the wavelength
 - Polarization data which is sensitive to the orientation of objects as well as their electrical properties
 - Bandwidth which determines the resolution of the imagery (it varies from hundreds of meters to centimeters depending on the radar system)



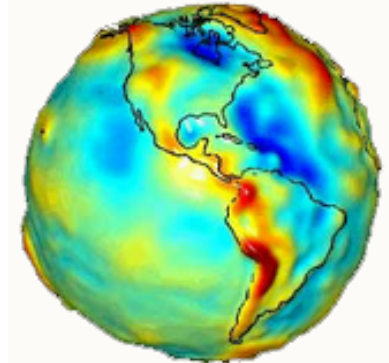
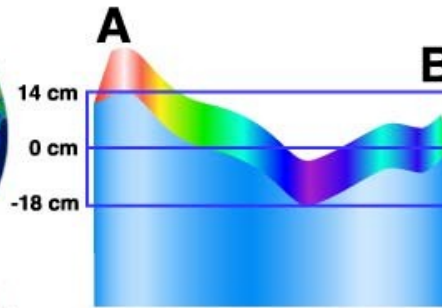
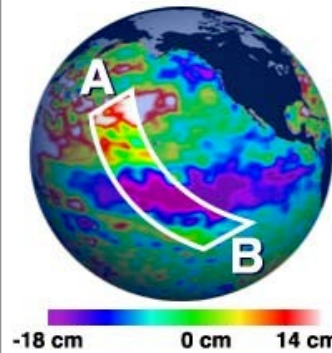
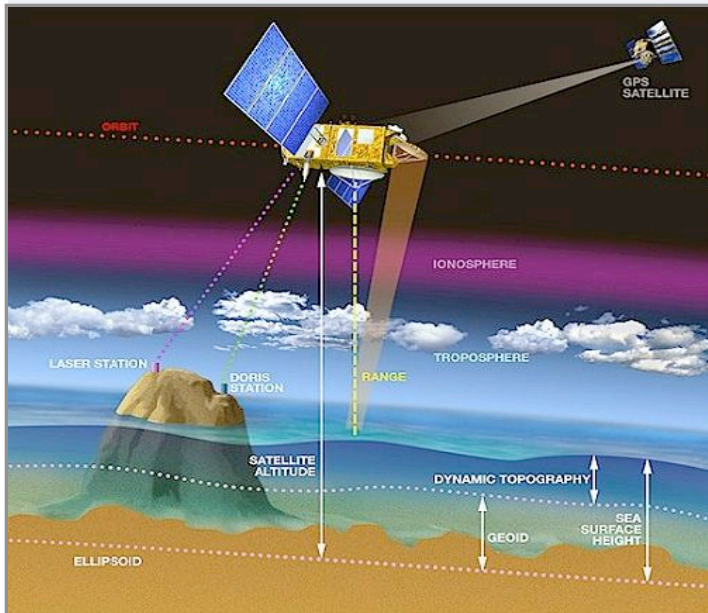
- Radar imagery may be obtained either from spaceborne or airborne platforms



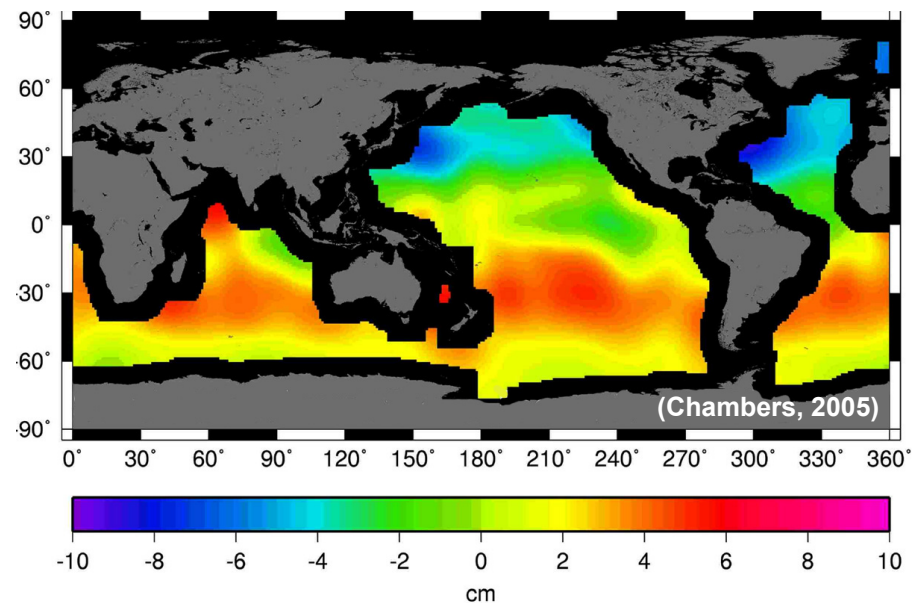
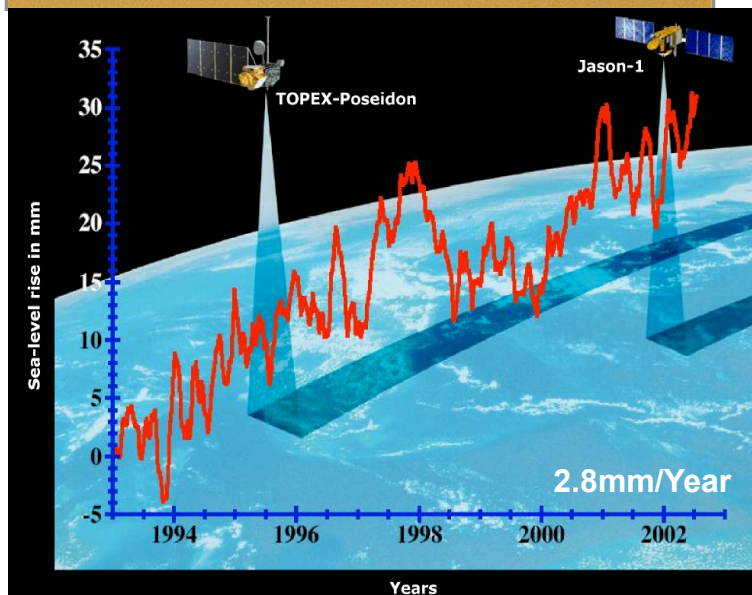
National Aeronautics and Space Administration

Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

Sea Level Rise using GRACE and Jason/Topex data



Topex/Jason topography (left)- GRACE mass change (right) = thermal expansion of the ocean (below)



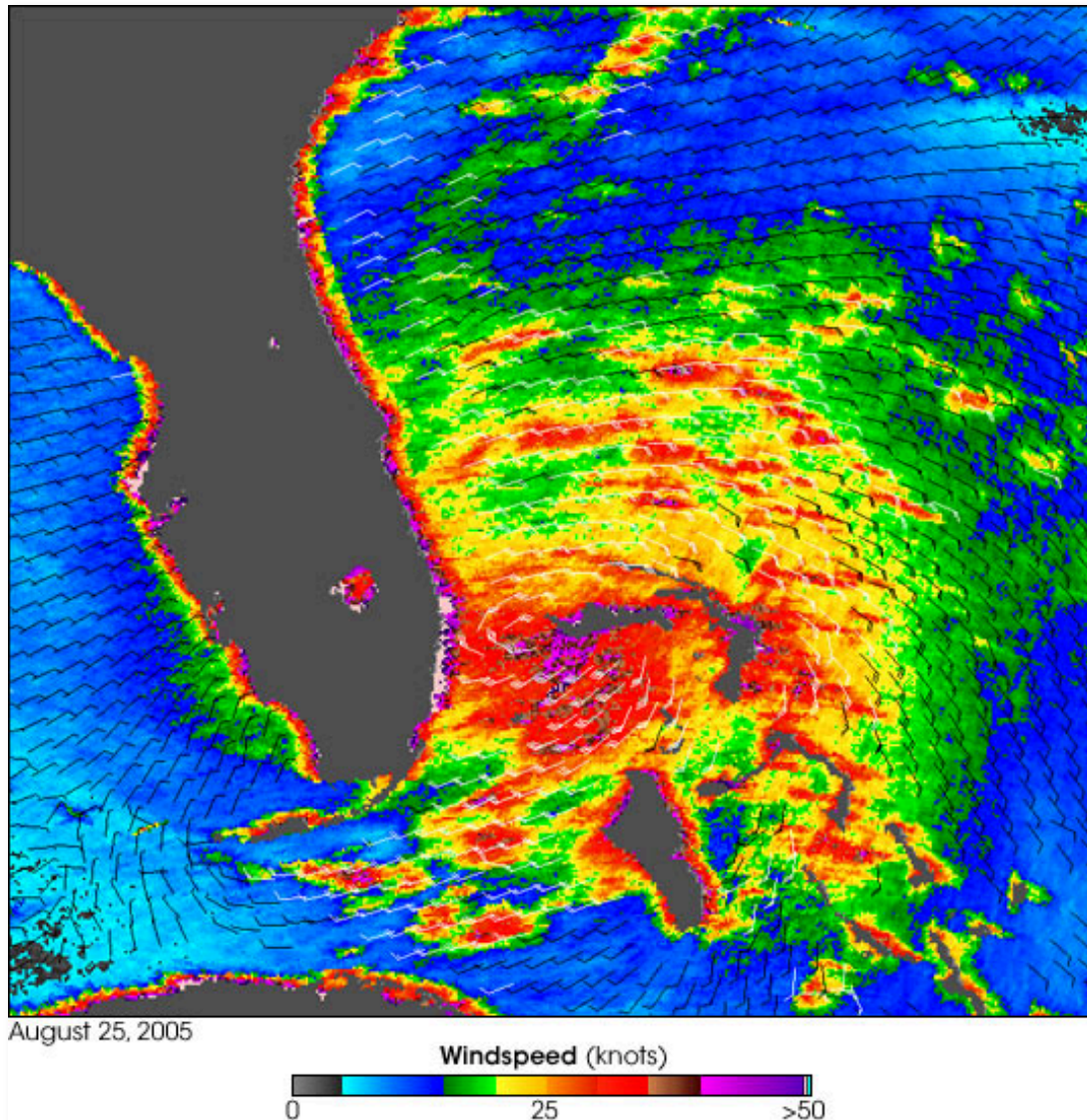


National Aeronautics and
Space Administration

Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California



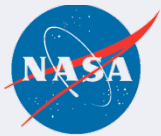
Radar scatterometer sees hurricane Katrina near Florida



- Scatterometers

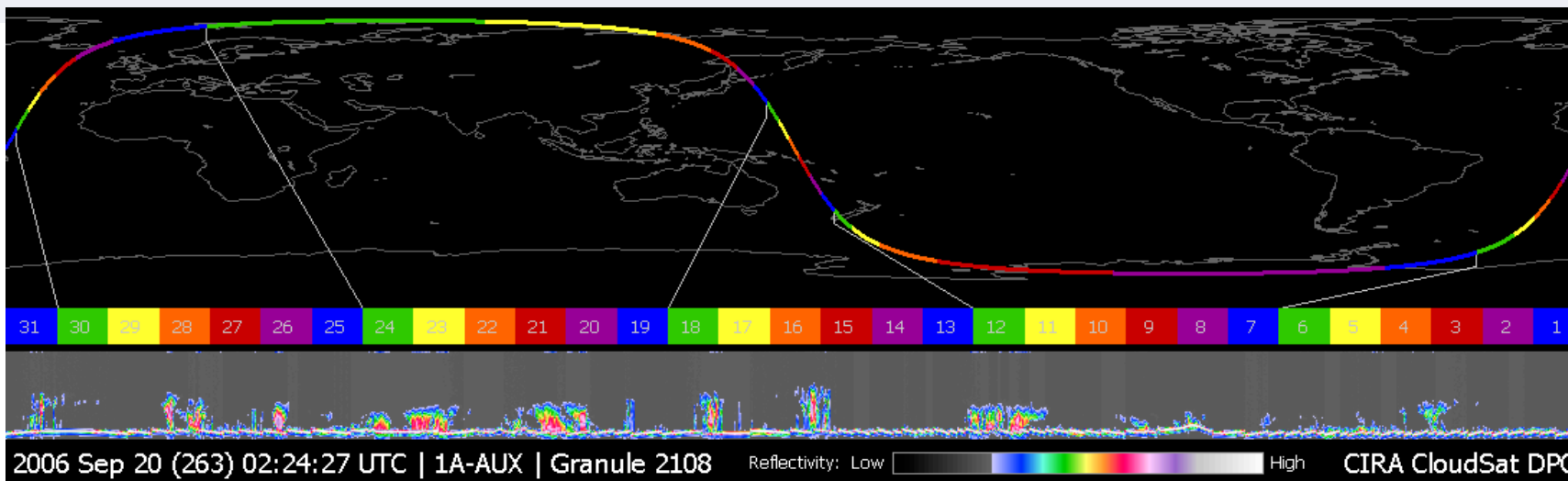
- Obtain global wind vectors on a daily basis
 - Research, climatology, weather operations
- Other applications
 - Ice edge detection, land change detection, snow cover, freeze/thaw detection, flood detection



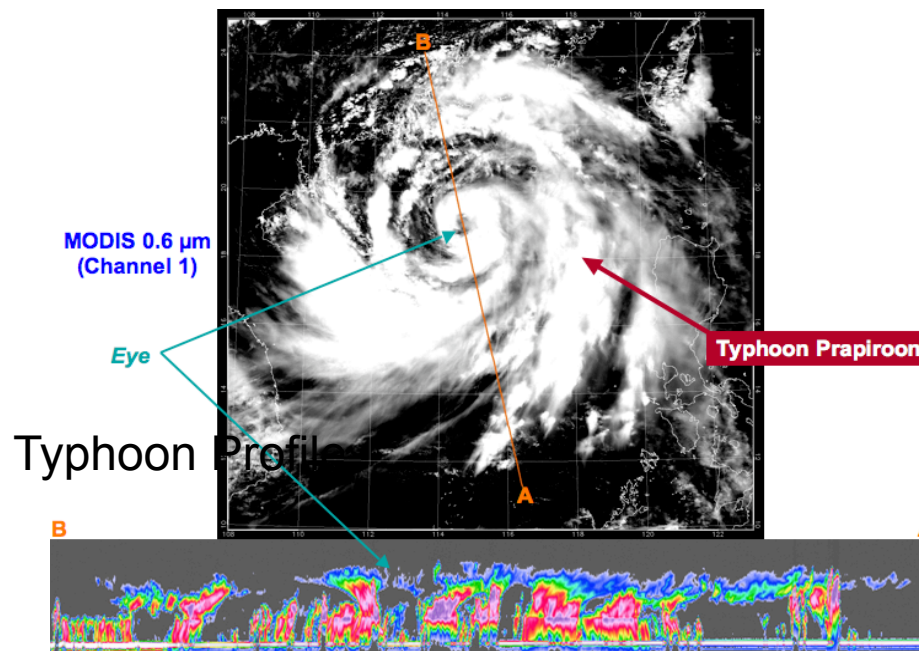


National Aeronautics and Space Administration
Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

CloudSat 94 GHz Cloud Profiling Radar - On-orbit Quick-look Data



Typical Orbital Profile

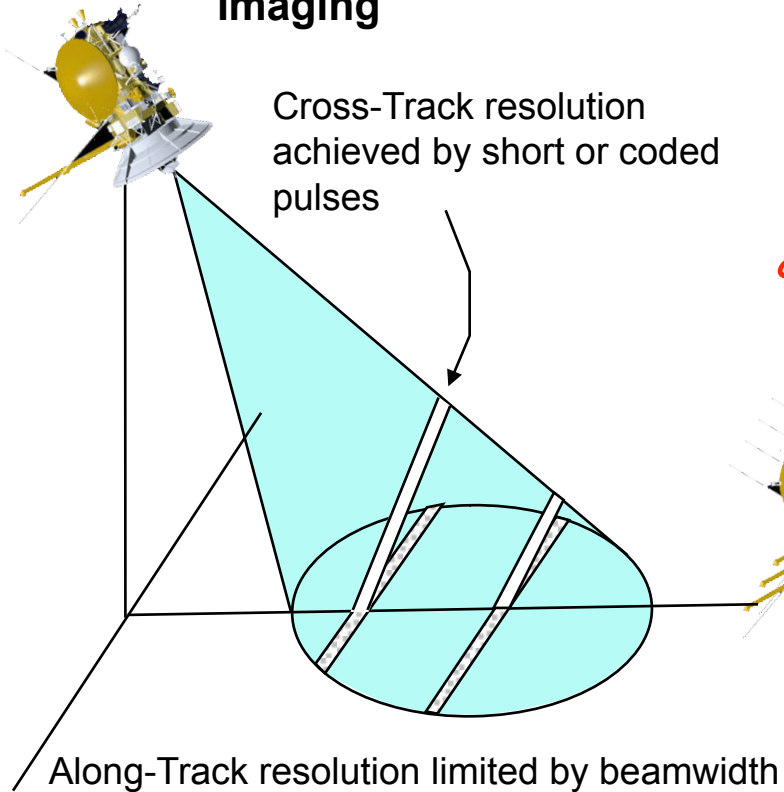




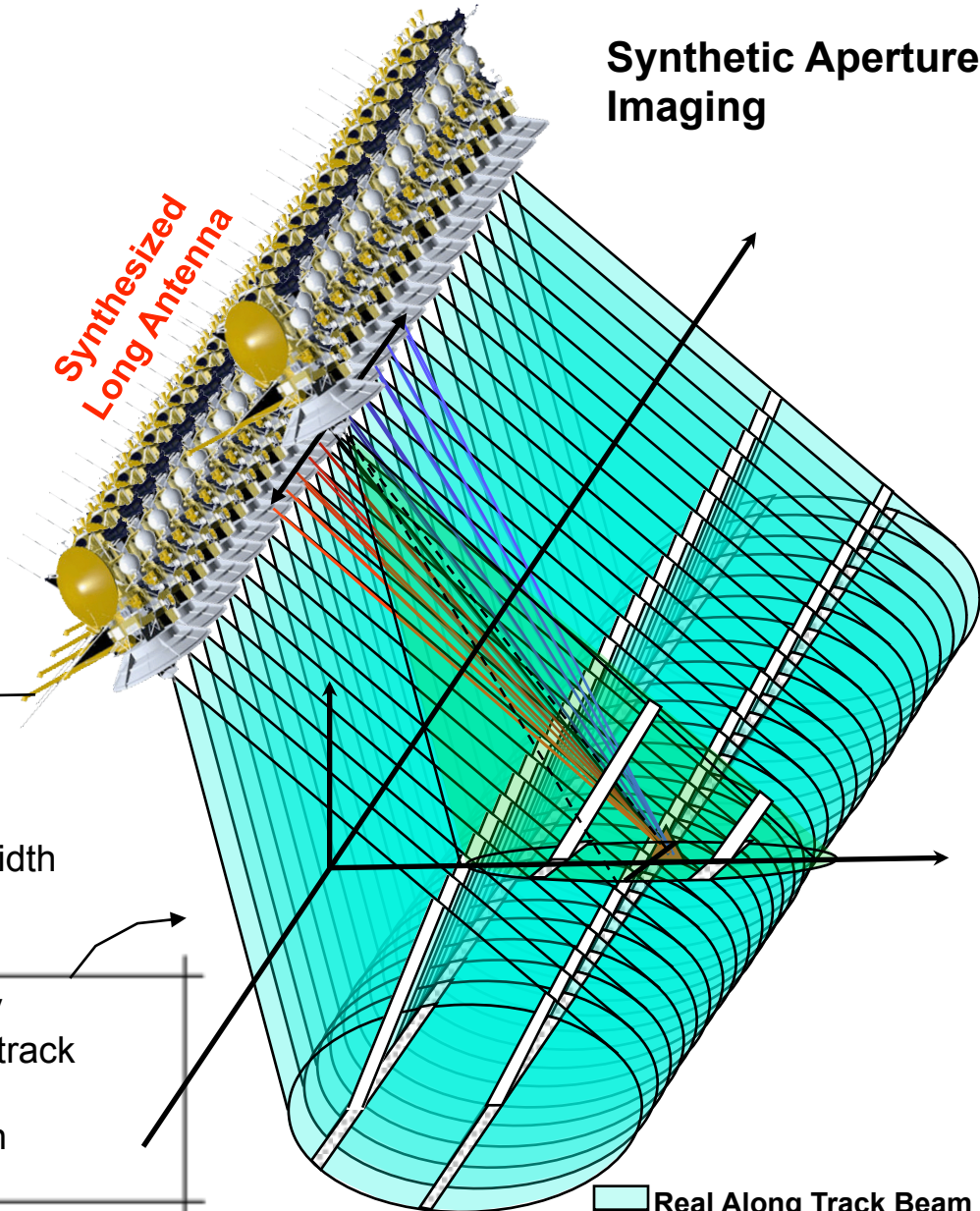
SAR Imaging Concept



Real Aperture Imaging



Synthetic Aperture Imaging



Along-Track resolution achieved by coherently combining echoes from multiple pulses along-track

- Resolution proportional to antenna length
- Independent of Range/Frequency

Real Along Track Beam
Synthesized Along Track Beam



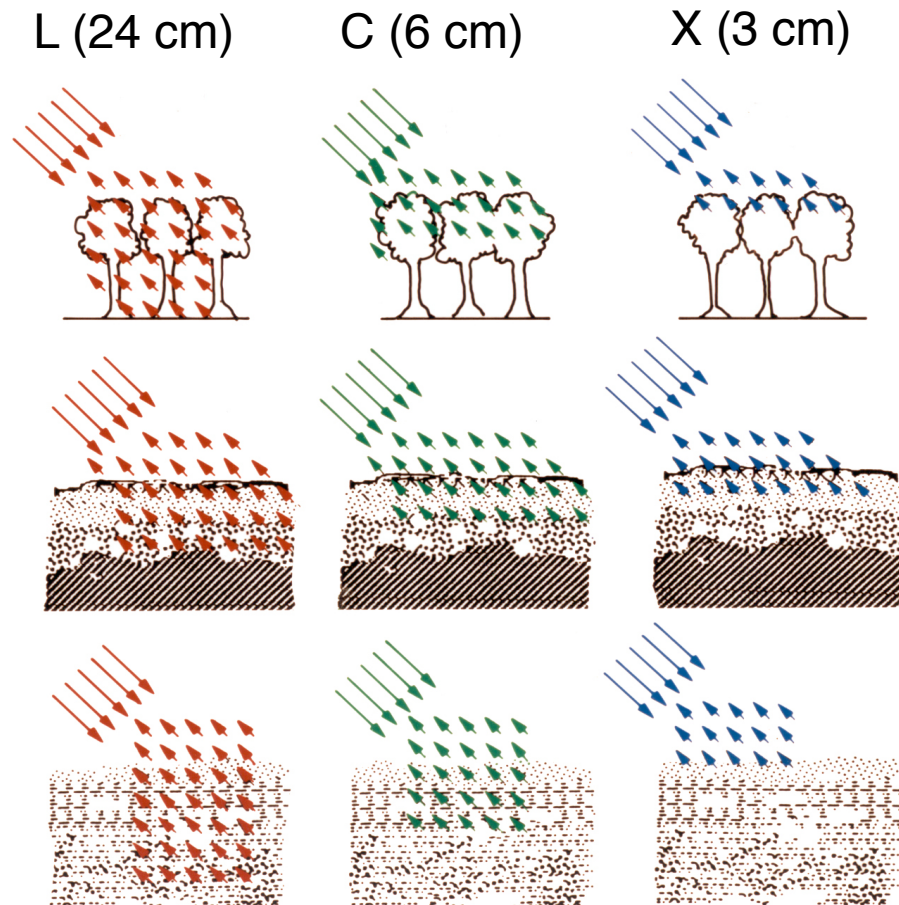
Wavelengths — a Measure of Surface and Subsurface Scale Sizes

Light interacts most strongly with objects on the size of the wavelength.

Forest: Leaves reflect X-band wavelengths but not L-band.

Dry soils: The surface looks rough to X-band but not L-band.

Ice: The surface and layering look rough to X-band but not L-band.





National Aeronautics and
Space Administration
Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

Modeling of Surface and Subsurface Scattering at Long Wavelength



- Radar signal penetrates dry sand, revealing bedrock structure below.
- Correlation of field measurements, data, and electromagnetic simulations yield greatest insight into applicability of techniques in other places
 - Other desert environments
 - Mars and other planetary bodies

Optical Imagery

L-band (24 cm)
SIR-A Observation



Northwest Sudan: Selma Sand Sheet. 8



National Aeronautics and
Space Administration

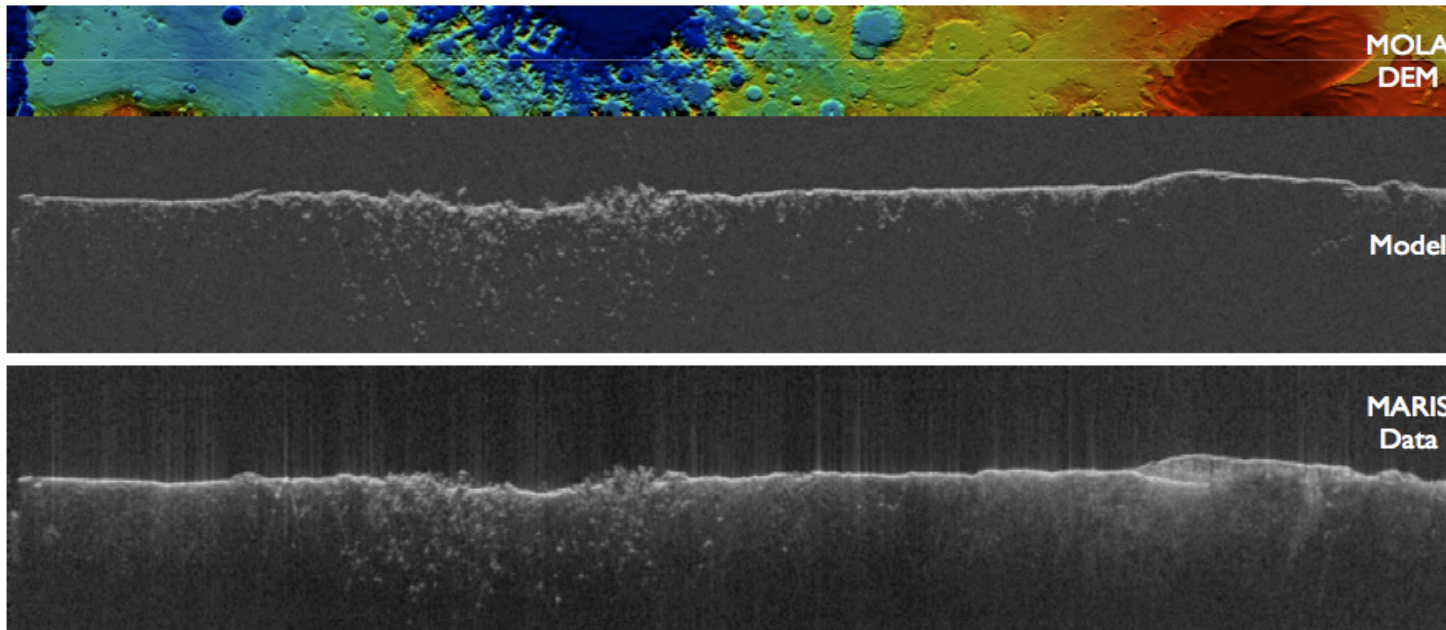
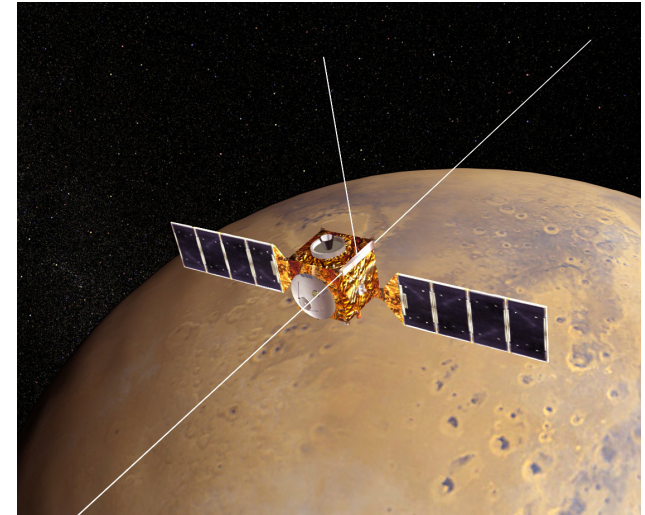
Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

Mars Advanced Radar for Subsurface and Ionospheric Sounding (MARSIS) on ESA Mars Express



Mission/Goals

- **Primary Goal:** To characterize the surface and subsurface electromagnetic behavior/variation in order to elucidate the geology (Search for water, material property, stratigraphy, structure, etc) at global scales with penetration depth of up to 5 km.
- **Secondary Goal:** To characterize the ionosphere of Mars



Modeling of
surface scattering
at very low
frequency reveals
subsurface
features

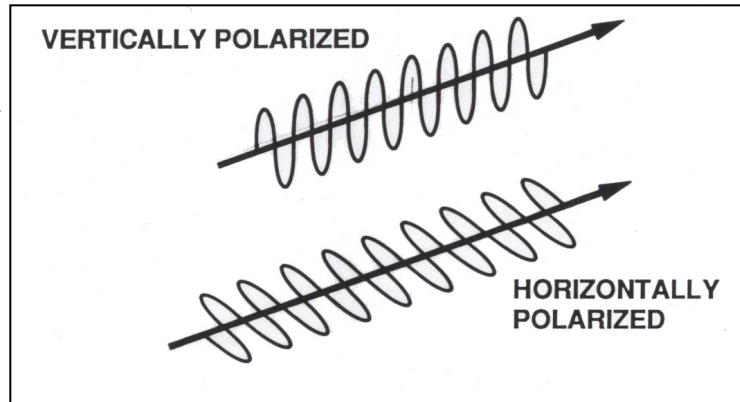


National Aeronautics and Space Administration
Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California



Polarization — a Measure of Surface Orientations and Properties

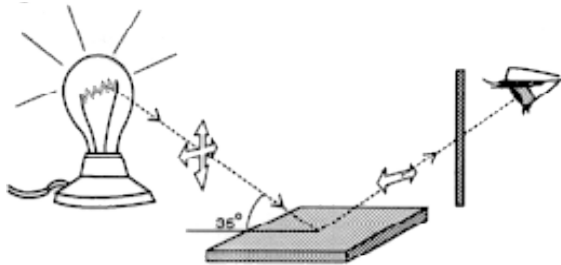
Wave Polarization



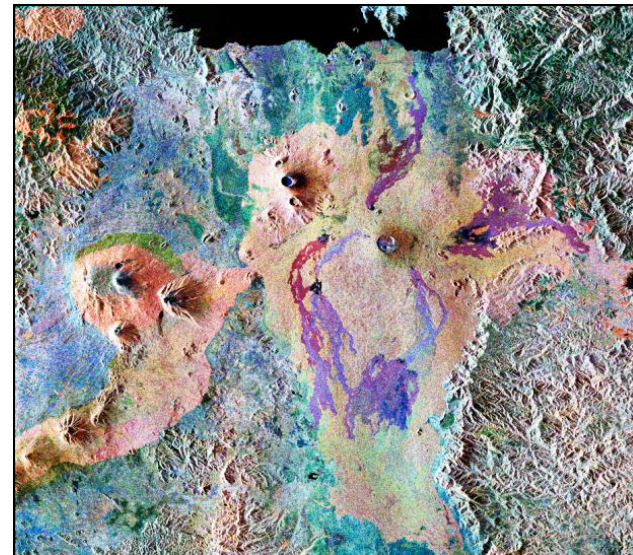
www.colorado.edu/physics/2000

Vertical polarization passes through horizontally arranged absorbers.

Horizontal polarization does not pass through



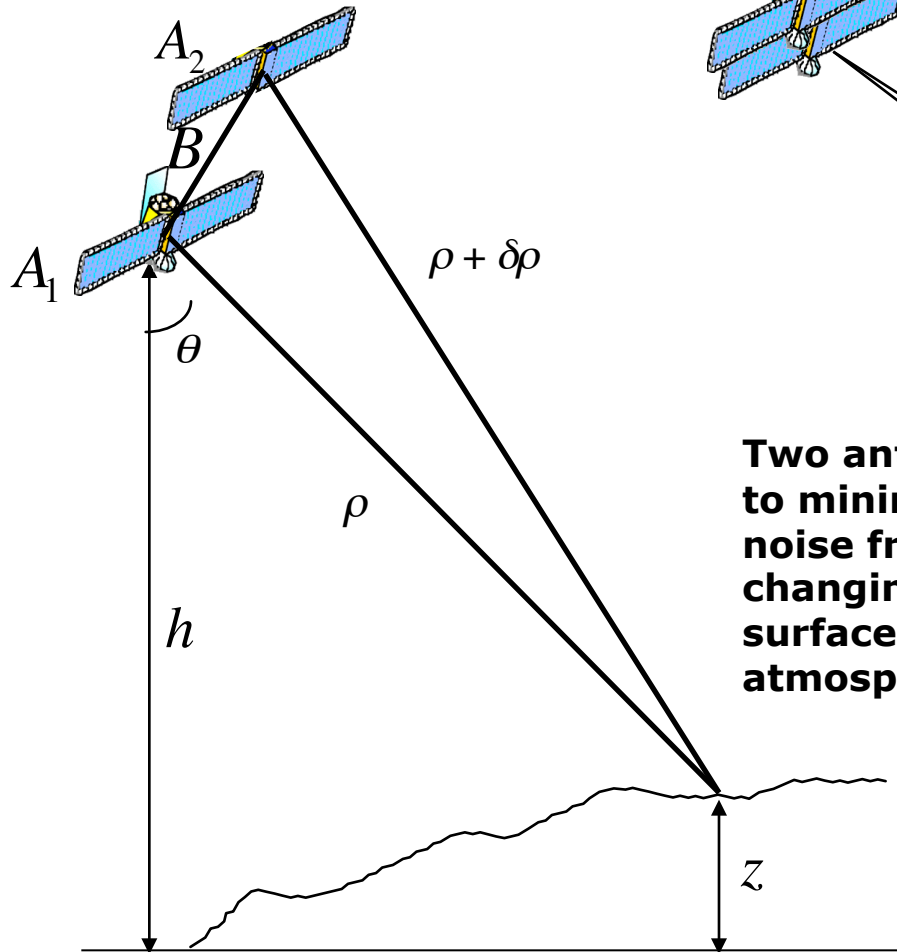
Mostly horizontal polarization is reflected from a flat surface.



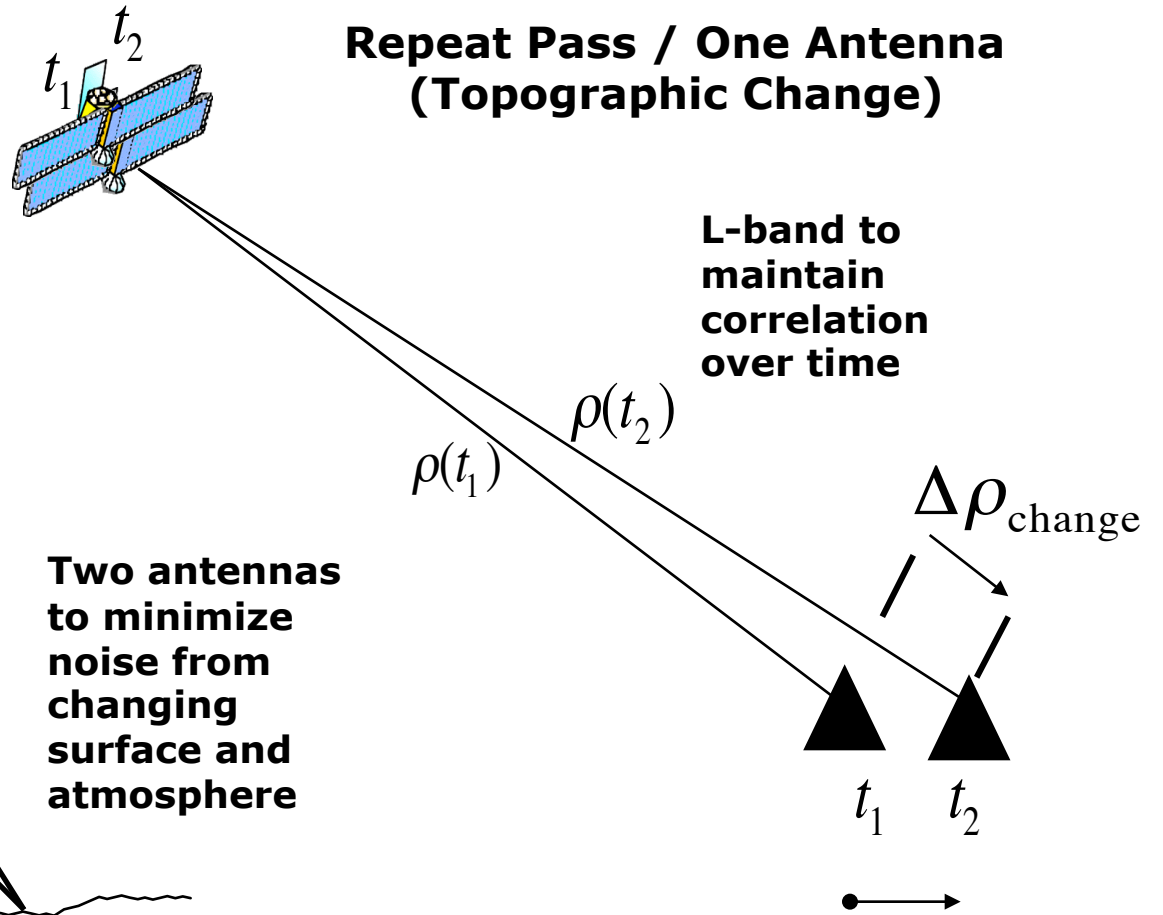
Polarimetry from SIR-C/X-SAR

Interferometry Basics

**Single Pass / Two Antennas
(Topography)**



**Repeat Pass / One Antenna
(Topographic Change)**



**L-band to
maintain
correlation
over time**

**Two antennas
to minimize
noise from
changing
surface and
atmosphere**



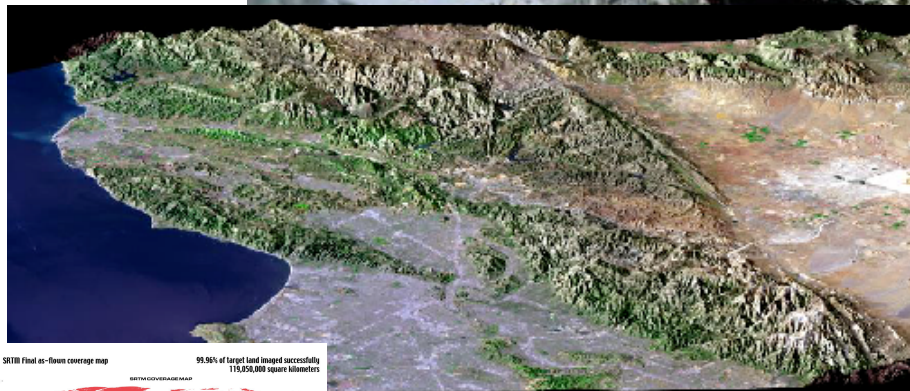
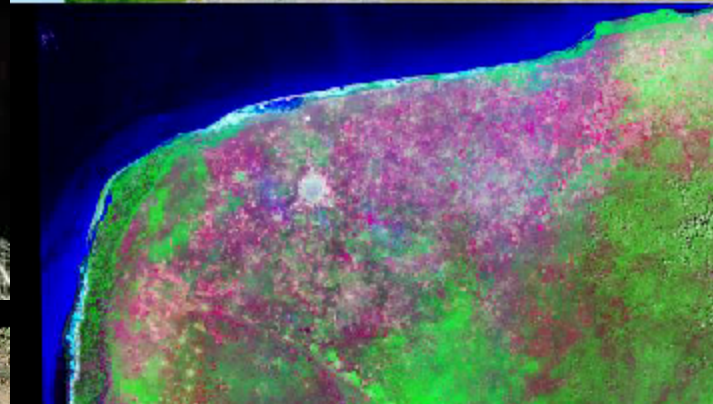
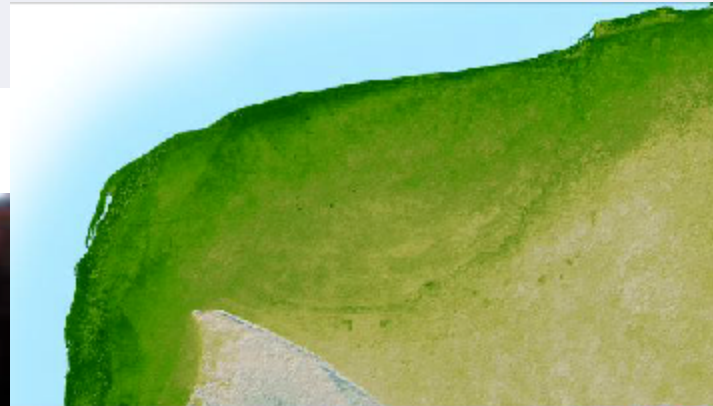
National Aeronautics and Space Administration

Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

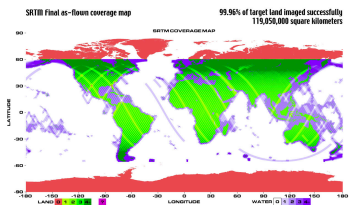
Shuttle Radar Topography Mission (SRTM)



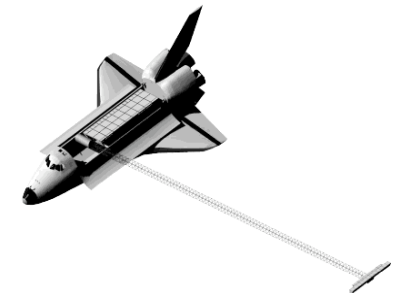
- Mapped 80% of Earth
- 30 m horizontal data points
- 10 m vertical accuracy



SRTM image of Yucatan showing Chicxulub Crater, site of K-T extinction impact. Bottom image is from Landsat showing Merida



3-dimensional SRTM view of Los Angeles (with Landsat data) showing San Andreas fault

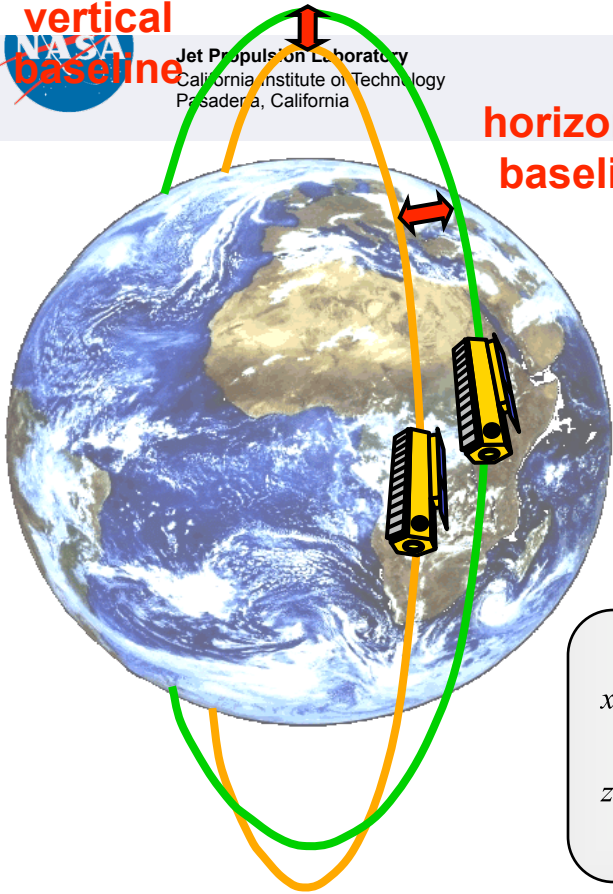


Tandem-X/L HELIX Formation

vertical
baseline

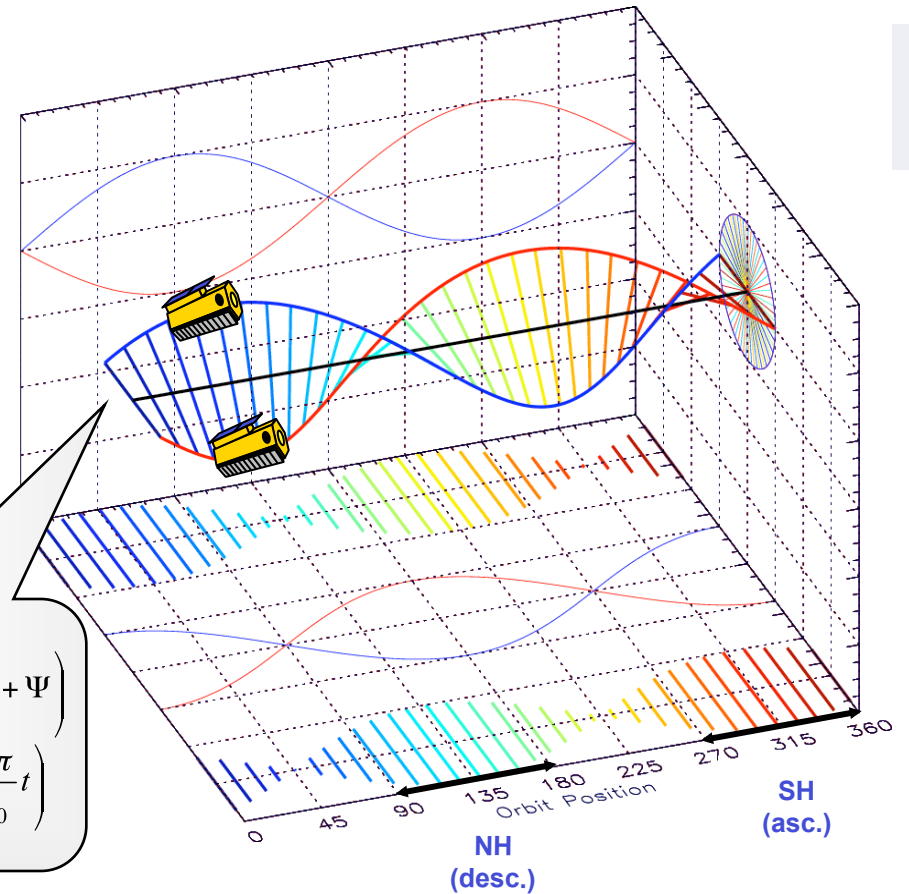
Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

horizontal
baseline



$$x(t) \approx a \cdot \Delta e \cdot \sin\left(\frac{2\pi}{T_0} t + \Psi\right)$$

$$z(t) \approx a \cdot \Delta \Omega \cdot \cos\left(\frac{2\pi}{T_0} t\right)$$



HELIX satellite formation enables safe operation

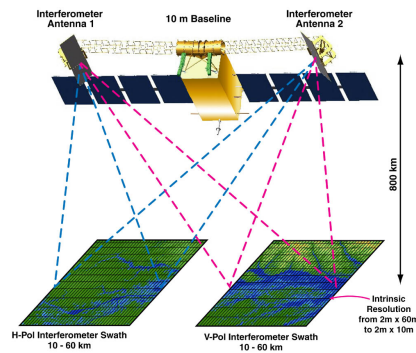
- *horizontal cross-track separation at equator by different ascending nodes*
- *vertical (radial) separation at poles by orbits with different eccentricity vectors (periodic motion of libration has to be compensated by regular manoeuvres)*



National Aeronautics and
Space Administration

Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

JPL Ka-band Radar Interferometry



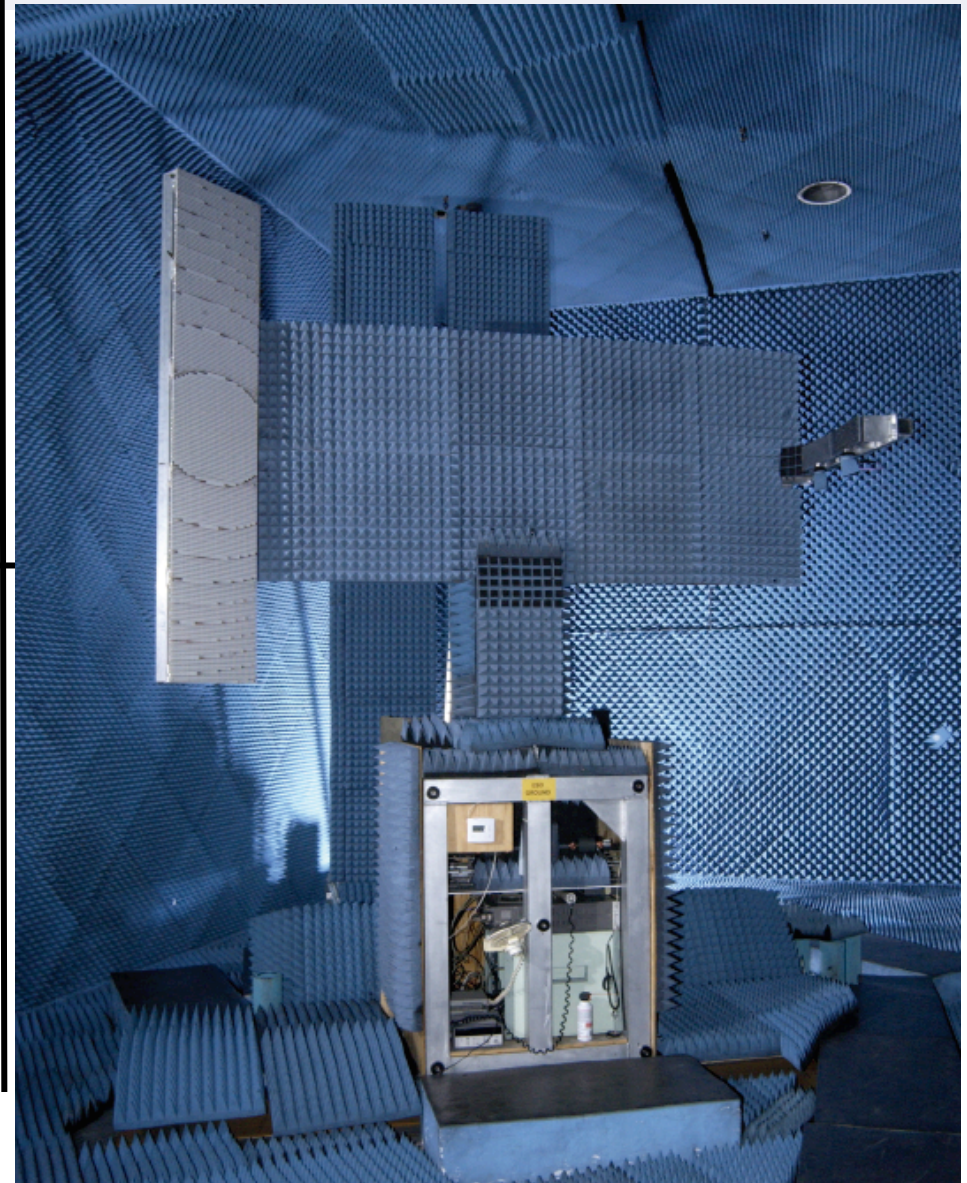
*Ka-band InSAR for glacier and
land ice surface topography
(GLISTIN)*

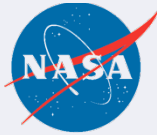


*KaRIN concept for mapping
surface water & ocean topography
(SWOT)*

Technologies:

- Ka-band center frequency to minimize electronics mass & volume
 - Mass varies by system design, but as low as 25 kg
 - Power usage varies by system design, but as low as 150 W.
 - Onboard processing can reduce data rate to as low as 2 Mbps.
- Significant re-use of technologies developed for JPL Ka-band altimeter / velocimeter, including:
 - Low-power, millimeter-wave hybrid front-end circuitry
 - Wide bandwidth analog design, with digital filtering & downconversion
- Additional heritage from JPL Wide Swath Ocean Altimeter development effort
 - Antenna / feed system
 - Algorithms for achieving centimetric height accuracy





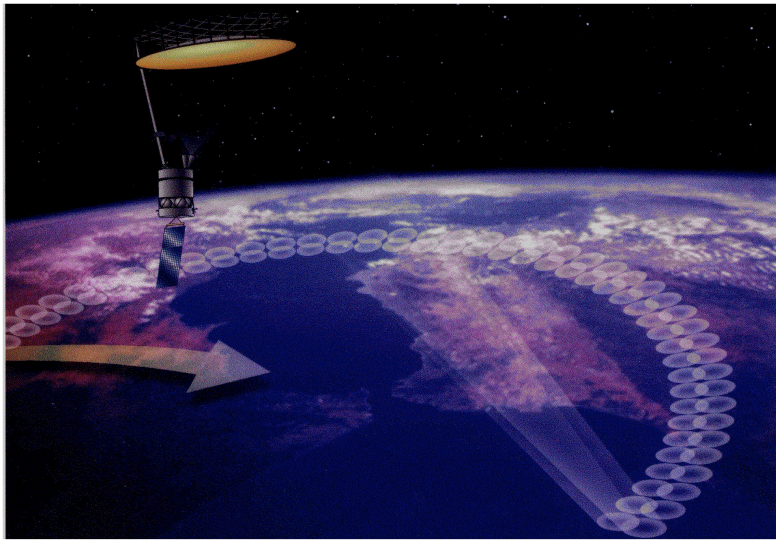
National Aeronautics and
Space Administration

Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California

Soil Moisture Mission Concept



Recommended by NRC decadal survey for early launch



Science Questions

Soil Moisture

- How are global precipitation, evaporation, and water cycling changing?
- How are global ecosystems changing?
- What are the effects of clouds and surface hydrological processes on Earth's climate?

Ocean Vector Winds

- How are global precipitation, evaporation, and the cycling of water changing?
- How is the global ocean circulation varying on interannual, decadal, and longer time scales?
- How can weather forecast duration and reliability be improved by new space-based observations, data assimilation, and modeling?

Precipitation

- How can weather forecast duration and reliability be improved by new space-based observations, data assimilation, and modeling?

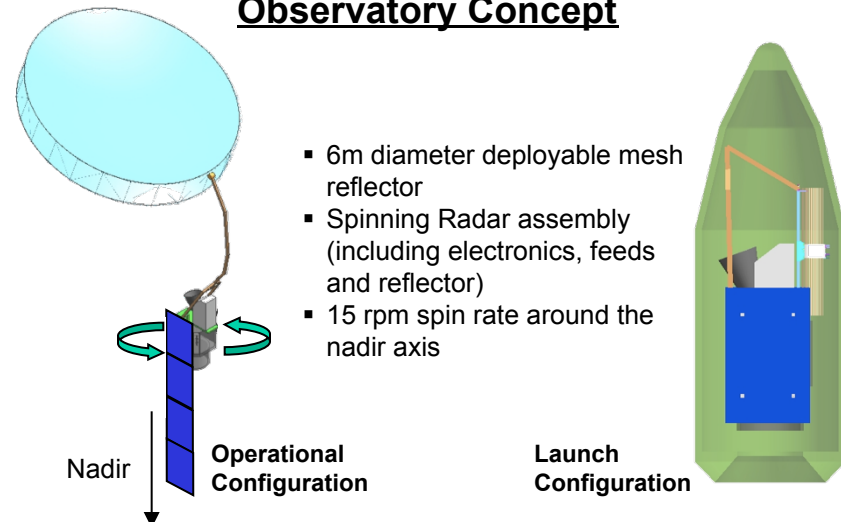
Features of the Mission Concept

- Launch Mid-2014
- Launch Vehicle:
 - Baseline: Taurus 3210 (small class)
 - Option 1&2: Delta 2320-10 (medium class)
- Orbit: ~98° Inc., 670 km Alt., 3 day quasi repeating ground track, 6 am ascending
- Mission Duration: 2 years (All Options)
- Payload: L-band radar and radiometer (HYDROS) (Baseline); + Ku-band radar (Option 1); +C-band radar and C,X,K,Ka-band radiometers (Option 2)
- Payload Avg. Power: 400W (Baseline) – 1332W (Option 2)
- Payload Mass: 233 kg (Baseline) – 445 kg (Option 2)
- Pointing: Within 0.1 deg to Nadir (Control); 0.05 deg (knowledge)
- Downlink Rate: 80 Mbps (Baseline) – 320 Mbps (Option 2)

Technology Development Needs

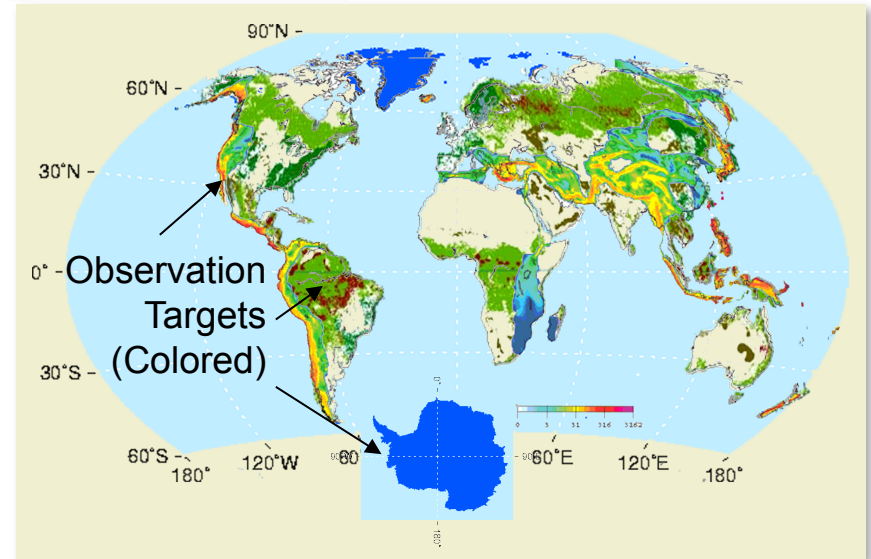
- Radar Electronics/Feed/Reflector spinning assembly

Observatory Concept

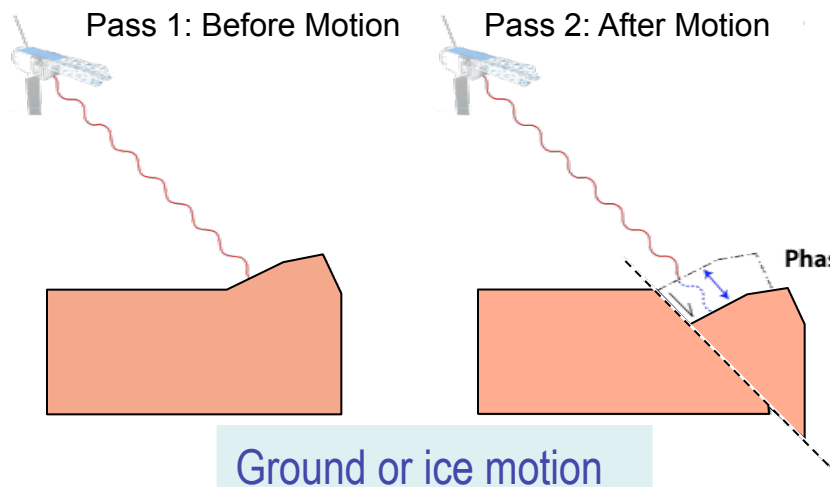


DESDynI: Deformation, Ecosystem Science, and Dynamics of Ice

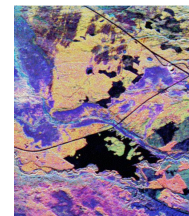
- Recommended by the NRC Decadal Survey for near-term launch to address important scientific questions of high societal impact:
- How do we manage the changing landscape caused by the massive release of energy of earthquakes and volcanoes?
- How are Earth's carbon cycle and ecosystems changing, and what are the consequences?
- What drives the changes in ice masses and how does it relate to the climate?



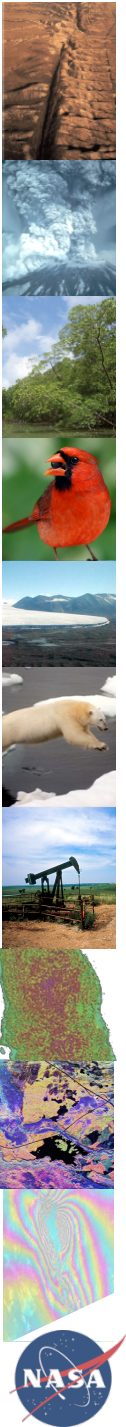
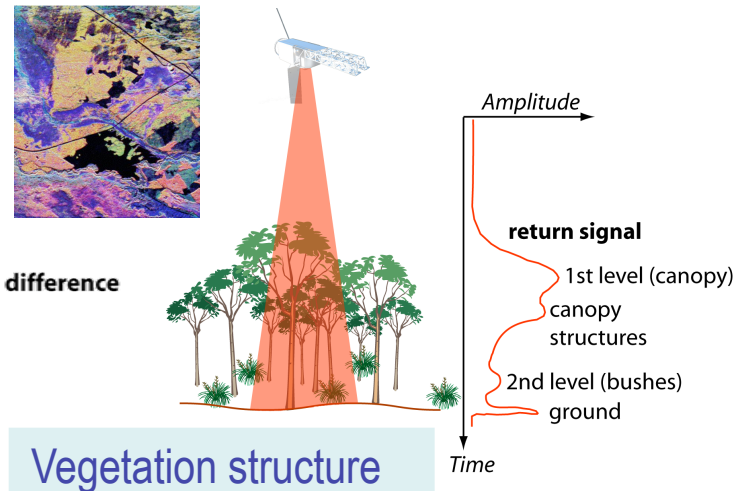
Repeat Pass InSAR



Polarimetric SAR



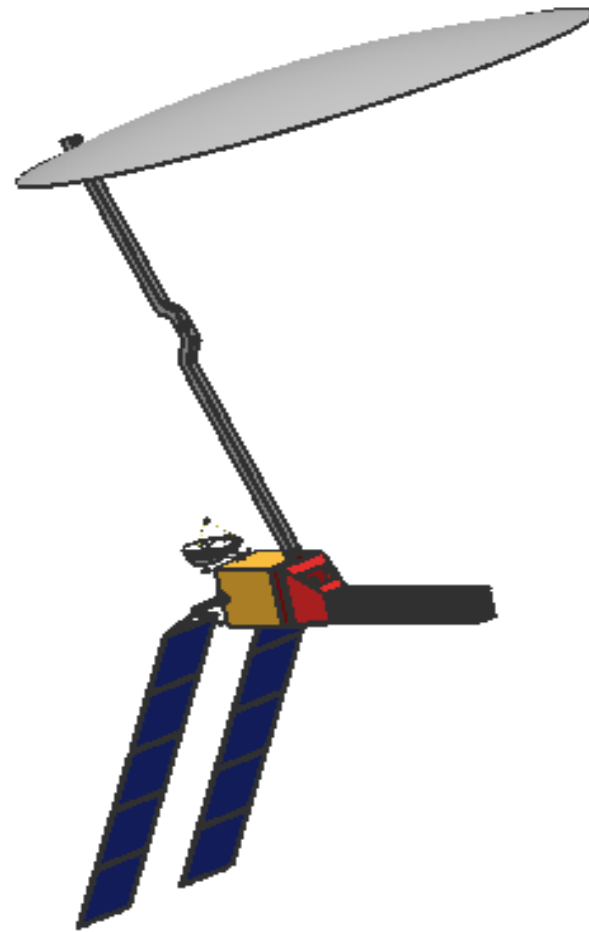
Multibeam LIDAR



DESDynI Two-Platform Configuration



Lidar-craft

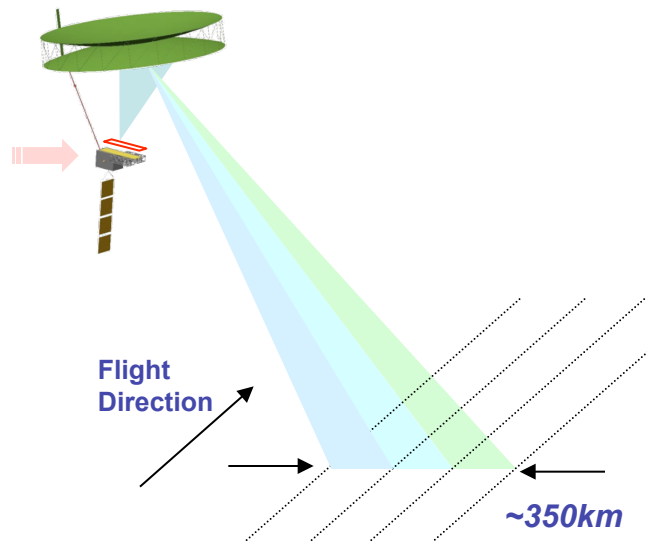


Radar-craft



DESDynI Instruments

L-Band Synthetic Aperture Radar

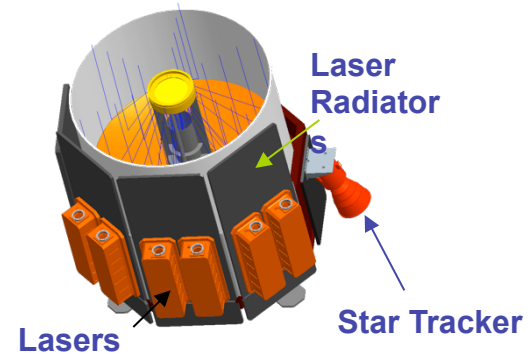


Technology

Key advanced technology investments have already been made

- L-band TR modules, antenna designs, trade studies, and modeling and simulation
- Under ESTO, UAVSAR system for quad-pol InSAR from aircraft
- Detailed engineering and packaging tasks remain

Multi-beam Lidar



- Lasers, Telescope, Gyro, and Star Tracker all tightly-coupled on composite optical bench
- Primary mirror diameter: 1.2m

Technology

Laser transmitter is currently at TRL 6*:

- GSFC-designed full waveform HOMER laser tested to full flight performance requirements (output power, rep rate, beam quality, efficiency, and lifetime)
- All components space qualified (TRL 6 or higher)
- Testing of laser ETU verifying the Multi-Beam Lidar performance in a relevant environment (vibration, thermal vacuum, etc.) to TRL 6.
- Detailed engineering and packaging tasks remain



GESS Roadmap

