



Solar Sails for CubeSats

Keck Institute for Space Studies (KISS) Workshop 2012 July 11 California Institute of Technology

Extracted from material by: Robert L. Staehle, Diana Blaney, Hamid Hemmati, Dayton Jones, Andrew Klesh, Joseph Lazio, Paulett Liewer, Martin Wen-Yu Lo, Pantazis Mouroulis, Neil Murphy, Paula J. Pingree, Thor Wilson, Chen-Wan Yen Jet Propulsion Laboratory, California Institute of Technology

Jordi Puig-Suari, Austin Williams
California Polytechnic University, San Luis Obispo

Bruce Betts, Louis Friedman The Planetary Society

Tomas Svitek
Stellar Exploration

Brian Anderson, Channing Chow University of Southern California

Preliminary progress report:

The NASA Innovative Advanced Concepts (NIAC) task on which this reports is still in progress. No mission described herein has been approved or funded.

Six Technology Challenges





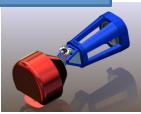
1. Interplanetary environment

Getting to cubesats Interplanetary



6. Maximizing downlink info content

5.Instruments



<u>Taxonomy</u>

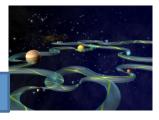
- Launch off C₃>0 ~ballistic traj
- Depart from "Mothership", 10s to 100s m/sec
 - Companion
 - Orbiter
 - Lander
 - Impactor
- Self-propelled
 - $1 10 \, km/sec/yr$
 - Electric
 - Solar Sail



2. Telecommunications



3. Propulsion (where needed)



4. Navigation

A Workable Interplanetary CubeSat System Architecture emerges from the maturation of

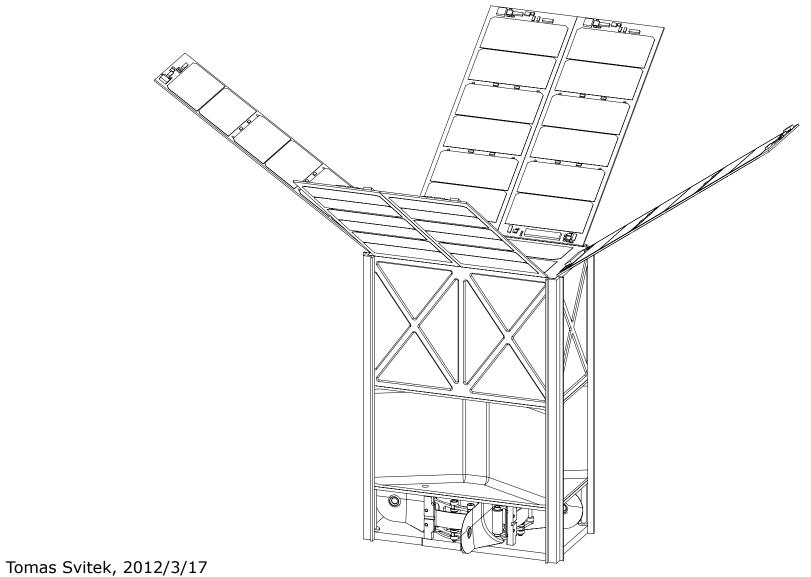
six key technologies

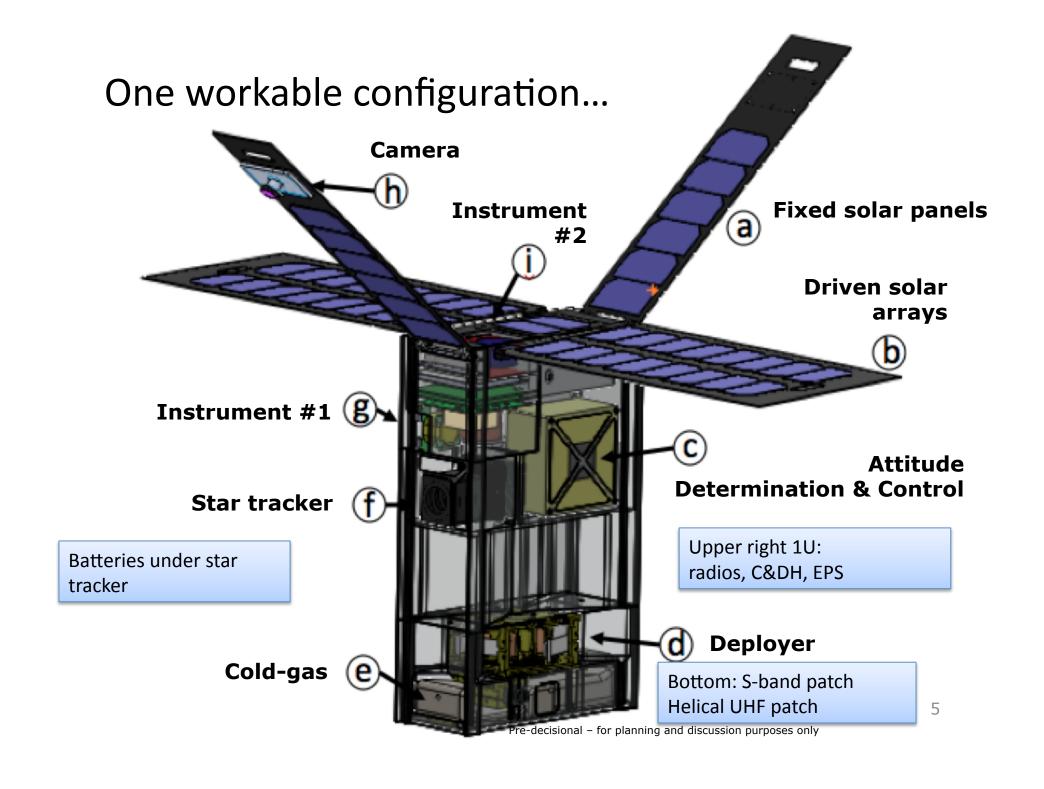


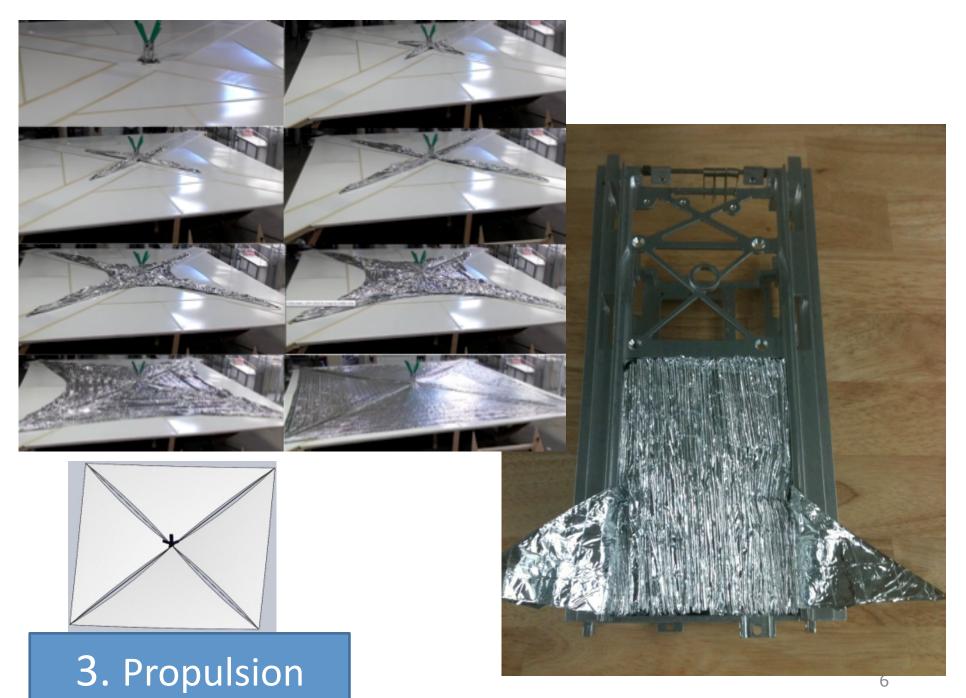
LightSail 1tm: Planetary Society, Stellar Exploration, CalPoly-SLO

RAX-2: University of Michigan

One Preliminary Configuration







Solar Sail Possibilities

Current technology

1 μg @1 AU → theoretical ~300 m/sec/yr

- Ikaros (2010: 1 μg), LightSail^[tm] 1 (2013?: 6 μg),
- Electrochromic surfaces for 2-axis control
- Switch to Kapton^[tm] from Mylar^[tm] would yield multi-year life
- Next 5-10 year projection (2021: 20 μg)
 - Tip vanes configured to provide 3-axis electrochromic control without moving parts.
 - Material thickness decrease 2-3X to enable larger sail packed into limited CubeSat volume.
 - Advanced (more expensive) material booms to enable longer boom to handle larger sail for same boom mass & volume.
- Next 10-20 years (2026: <100 μg?)
 - Even thinner materials, sublimating substrate, more advanced booms.
 - High temp materials to allow close solar approach, high ΔV in short time.
 - (a 91 µg (at 1 AU) sail starting from 0.3 AU reaches 100 AU in 17 yrs; 0.2 AU → 13 yrs)
 - Most spacecraft functions printed on inner part of sail.*
- * As discussed at Kendra Short/JPL 2012/3/19 NIAC Printable Spacecraft Workshop

5/31/20

In principle, you can build a sail...

- For propulsion, or as an antenna, or both.
- That is maneuverable in position and attitude.
- Up to ~10 m on a side square, or most any axisymmetric shape, to fit into ~2 U of a CubeSat, or larger on a bigger satellite.
- With low to moderate geometric accuracy (centimeters to millimeters)
- Having conducting and non-conducting portions at scales from <1 mm to meters, with whatever connectivity you can draw on two sides of an insulating sheet.
- With conducting (easy) or non-conducting (harder) structural spars.
- 3-axis stabilized or spinning.

If you don't want a sail, you can...

- Build a tether/antenna to kilometers in length.
- Use crossed tethers and spin.
- Add structure for 3rd dimension (e.g., inflatables, deployable booms)
- Use more of you imagination...