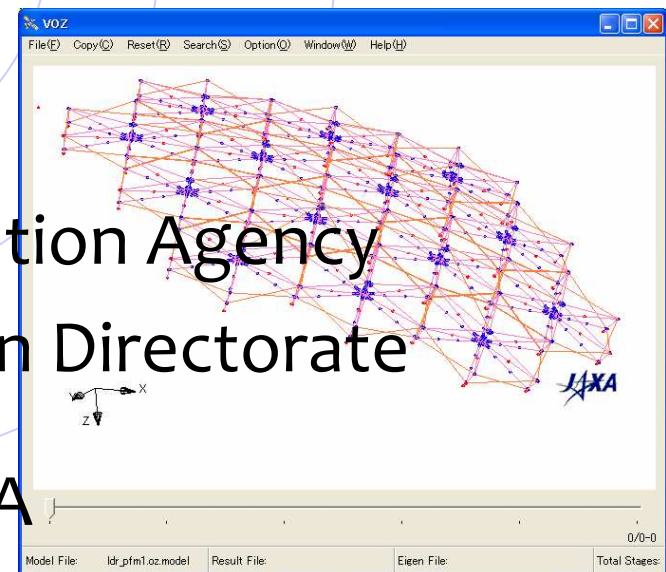


Large Deployable Reflector on Engineering Test Satellite VIII

Japan Aerospace Exploration Agency
Space Applications Mission Directorate

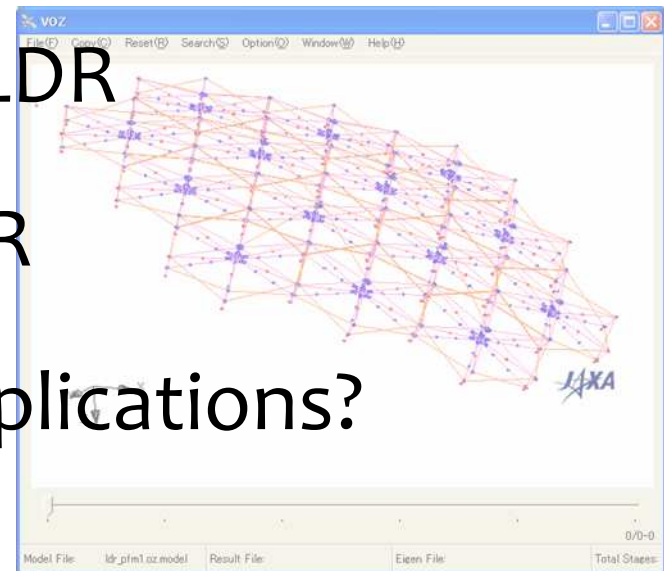
Satoru OZAWA



Today's Talk

Large Deployable Reflector (LDR)

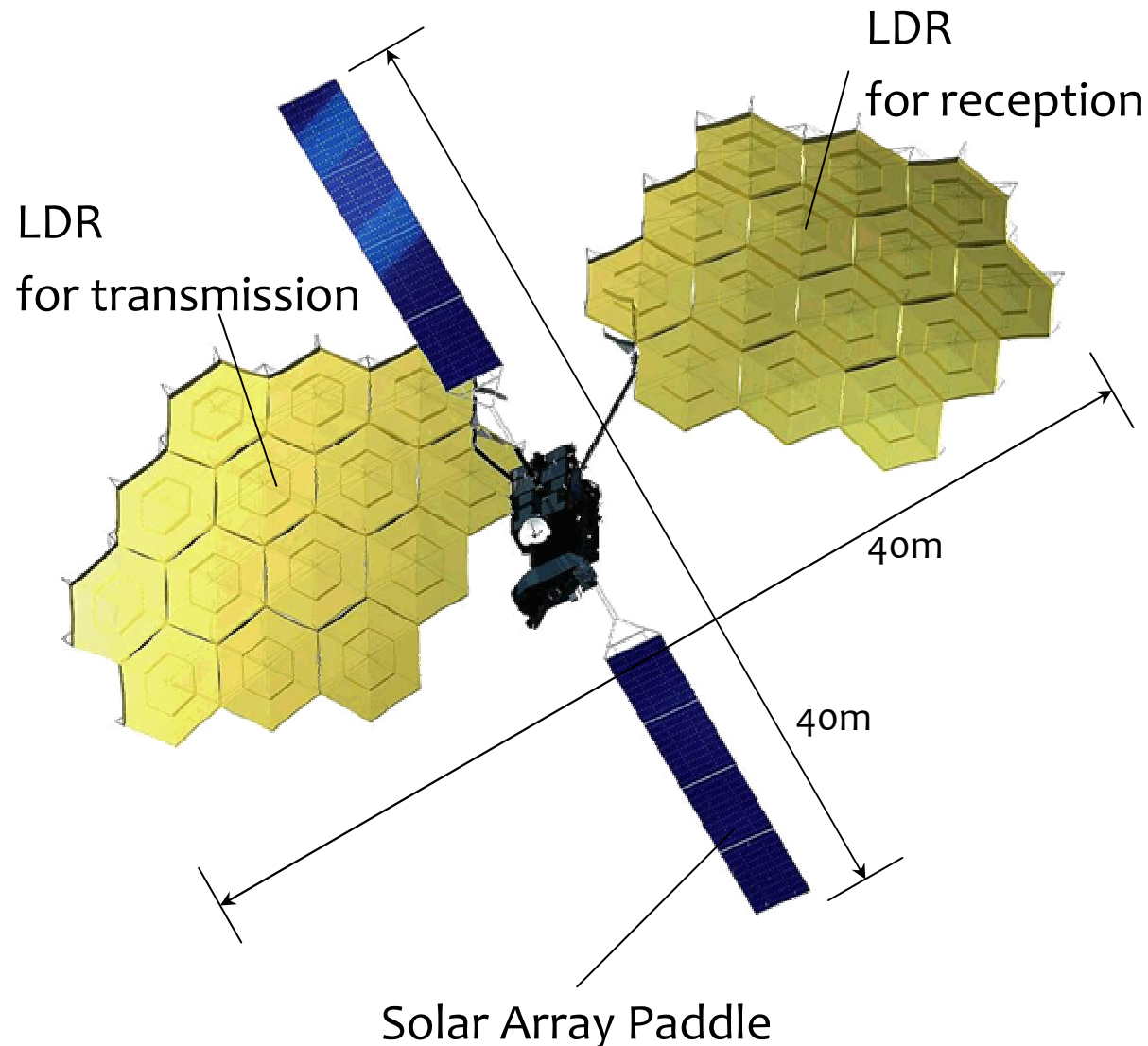
- What is Engineering Test Satellite VIII?
- What is Large Deployable Reflector?
- Development strategy of LDR
- In-orbit deployment of LDR
- What are Future Space Applications?





What is the Engineering Test Satellite VIII?

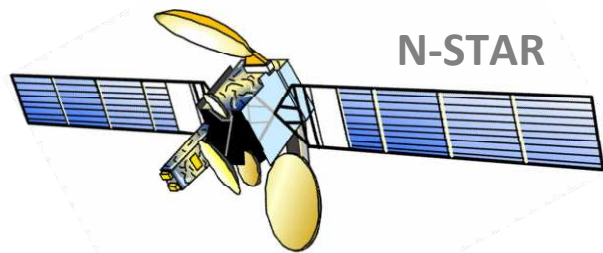
Overview of the Engineering Test Satellite VIII



Type	Communication satellite
Frequency band	S band
Mass	3000 kg on-orbit at start of mission
Electric power	7.5 kW
Orbit	146° E longitude
Design life	10 years (s/c) 3 years (mission)

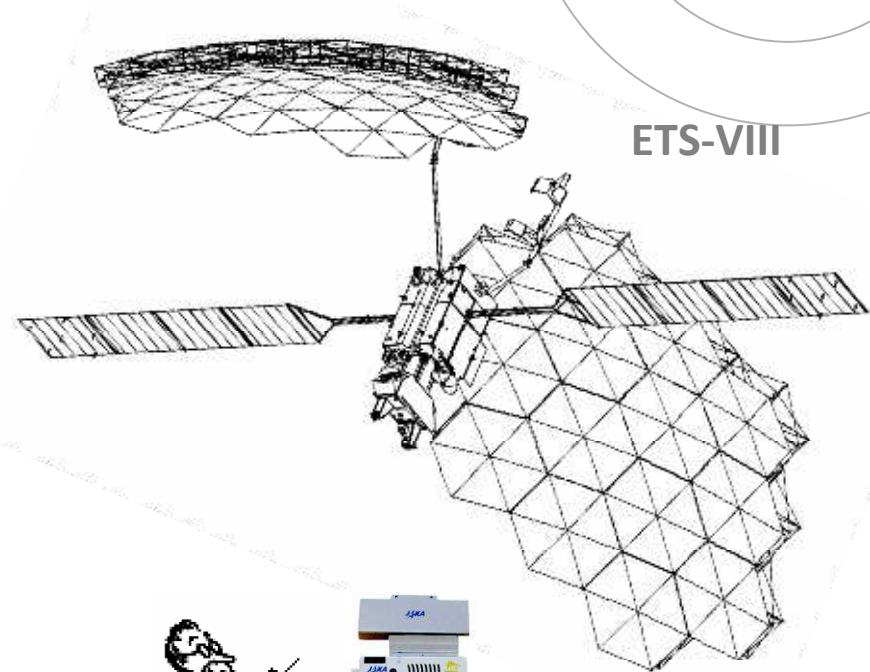
Objective of Engineering Test Satellite VIII

Mobile Communication Satellite

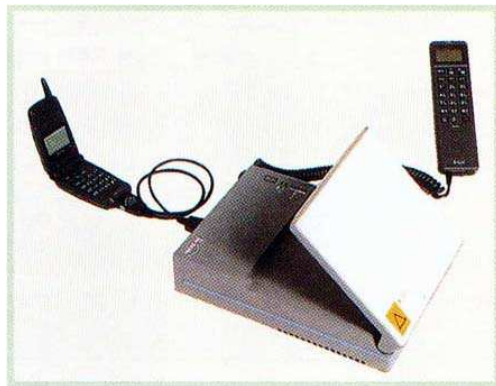


N-STAR

LARGE



ETS-VIII



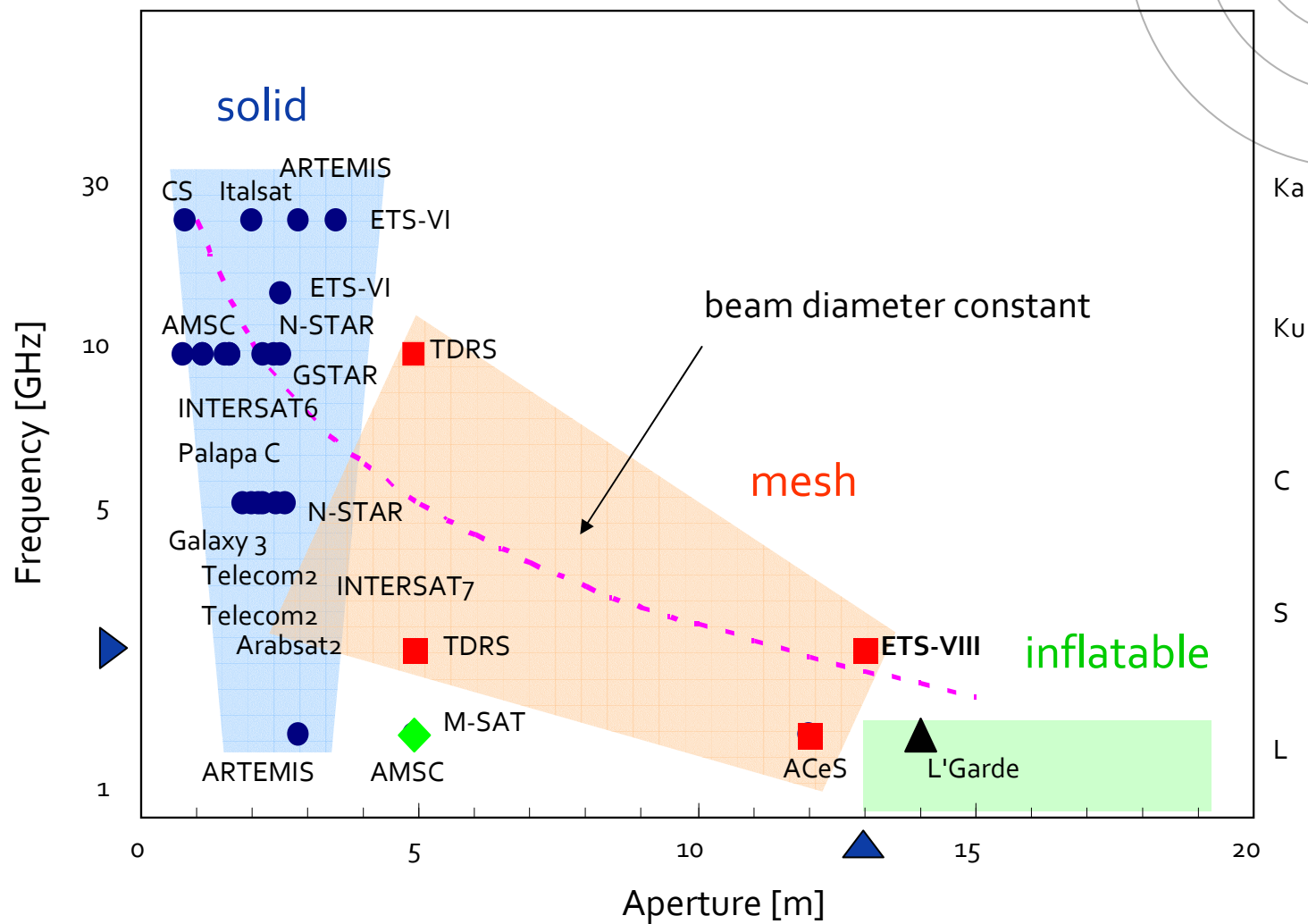
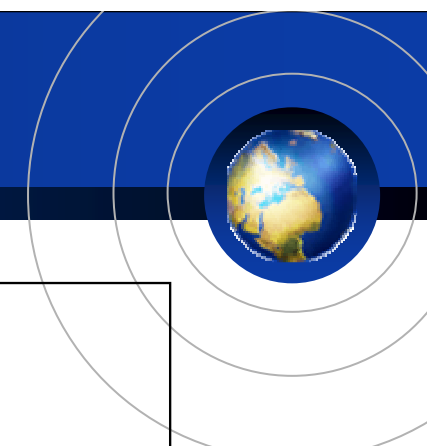
Portable size terminal

SMALL



Handheld size terminal

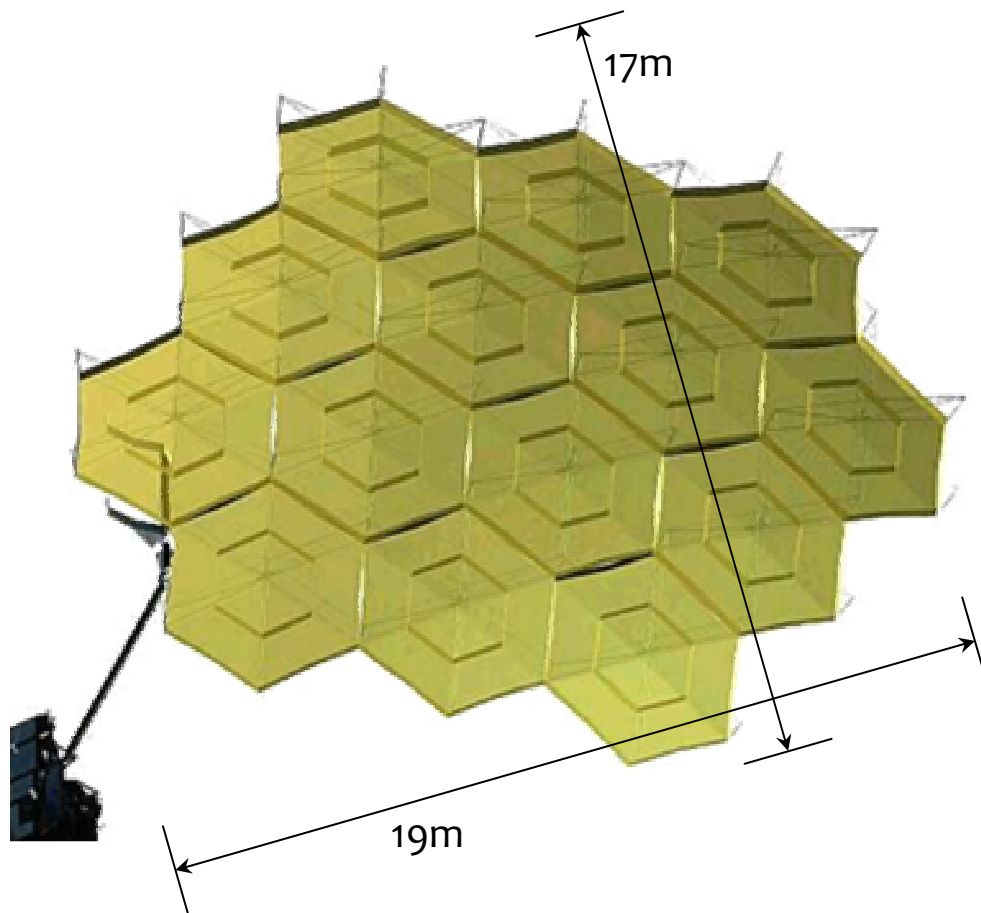
Target of Engineering Test Satellite VIII





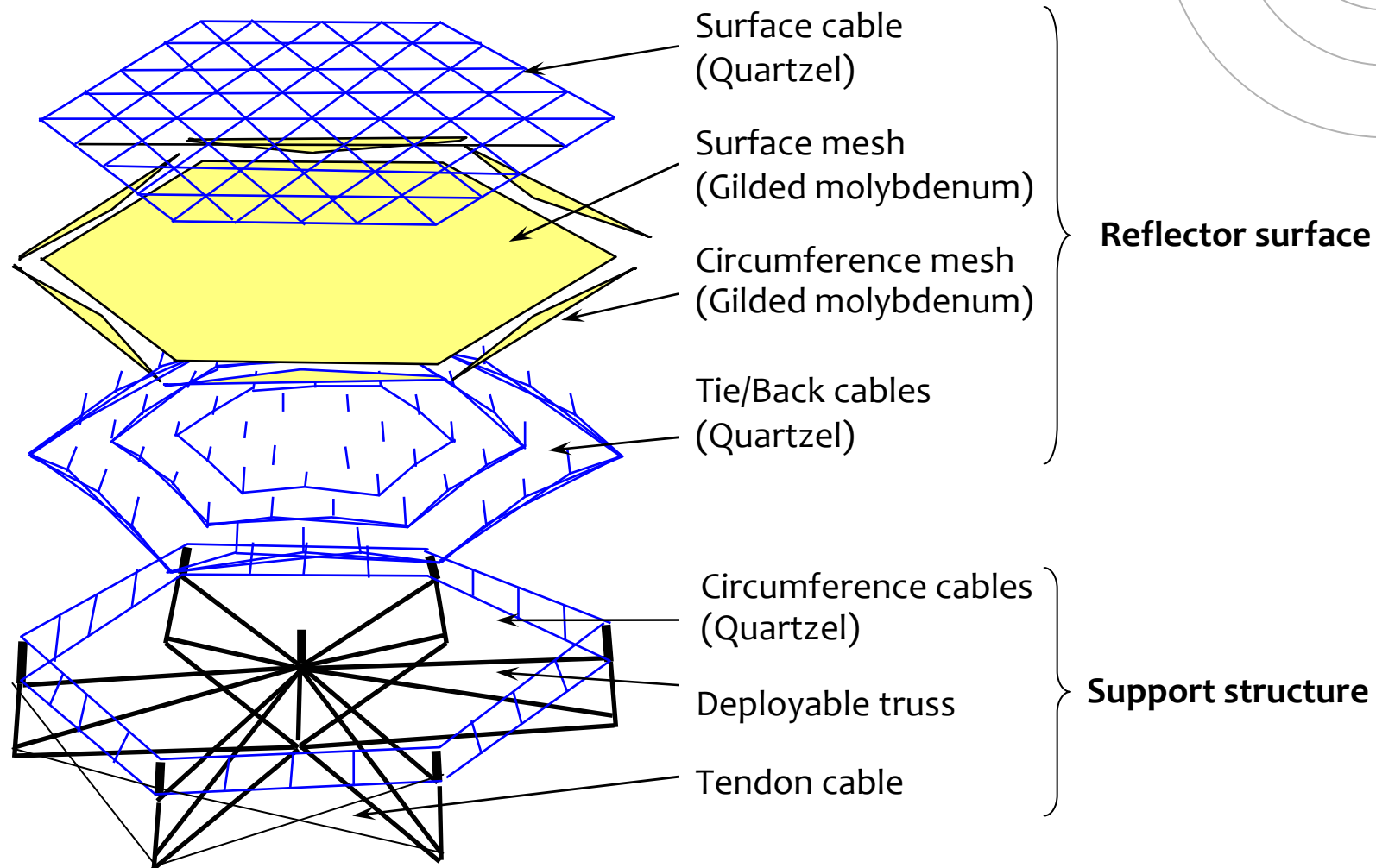
What is Large Deployable Reflector?

Overview

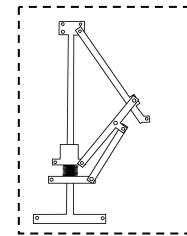
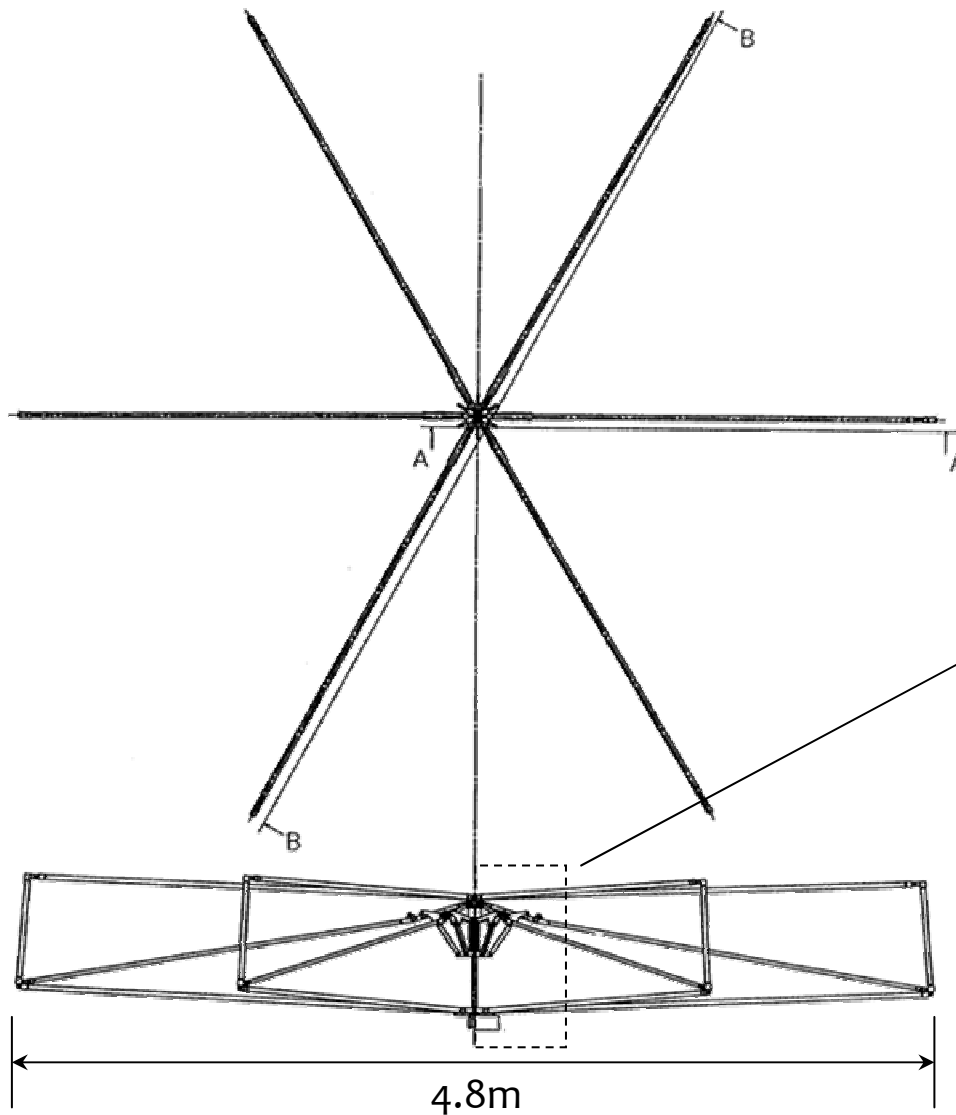
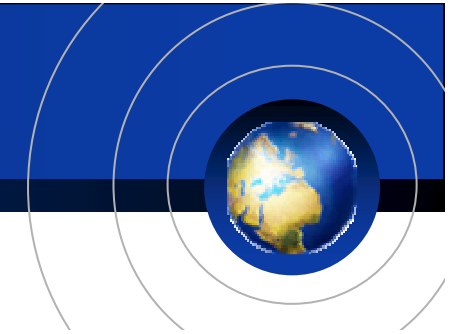


Type	Offset Parabola
Size	About 19m x 17m
Nominal aperture	13m (inscribed circle)
Number of modules	14
Offset angle	51.2°
Stiffness	0.09Hz or more
Accuracy	2.4mmrms or less
Weight Density	105kg (reflector) 650g/m ²


Composition of each module



Support structure

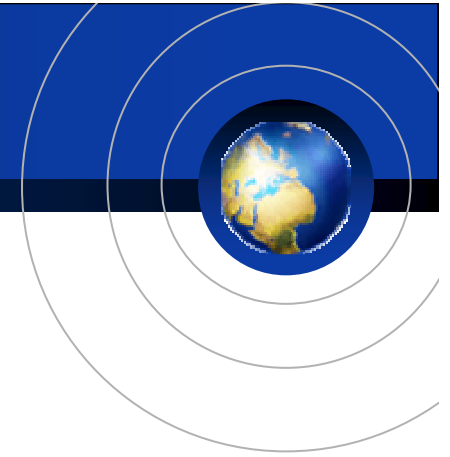


- Radial truss structure.
- Spring deploys structure.
- Motor controls deployment speed and enables rewind.



Development strategy of Large Deployable Reflector

Development strategy of the Large Deployable Reflector



Strategy

- Two main pillars:
Module and Analysis
- Ground tests and Updating analysis model
- High accurate surface forming
- Analysis precision evaluation by LDREX-2
- Evaluation of Large Deployable Reflector

Two main pillars

Module and Analysis

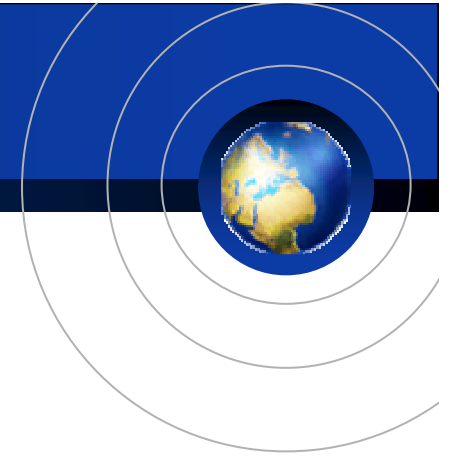


Strategy

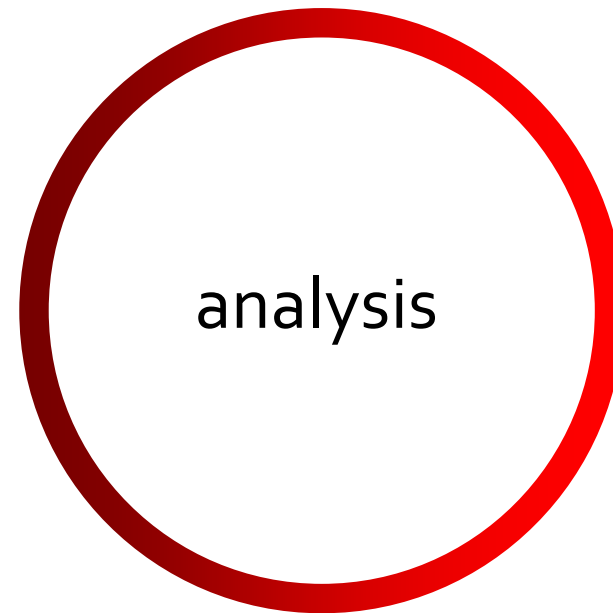
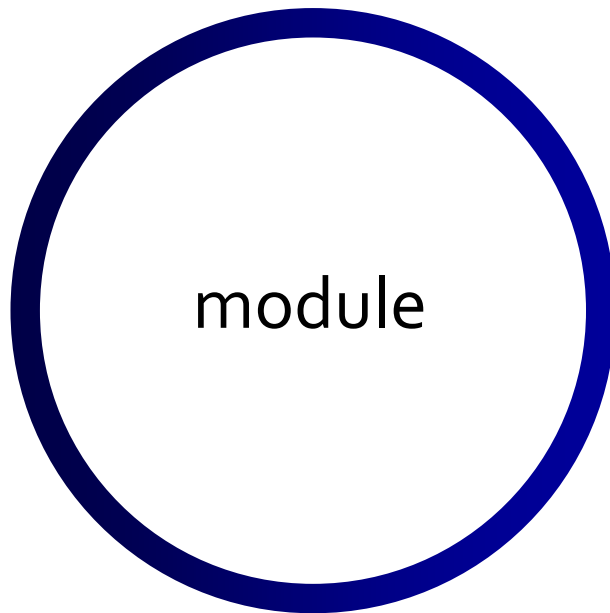
- **Two main pillars:**
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- Evaluation of Large Deployable Reflector

Two main pillars

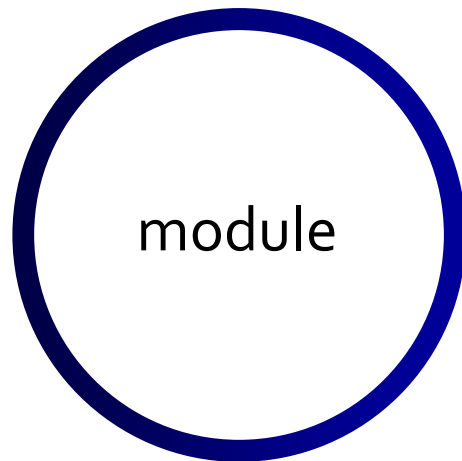
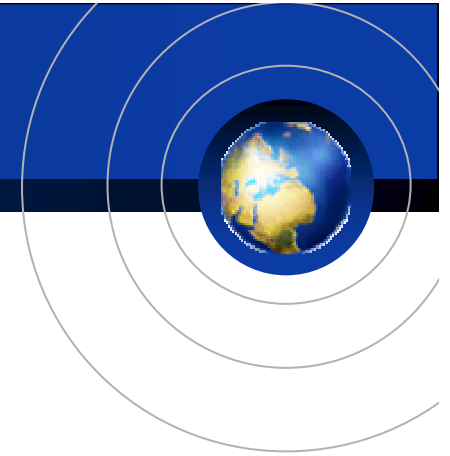
Module and Analysis



Two main pillars

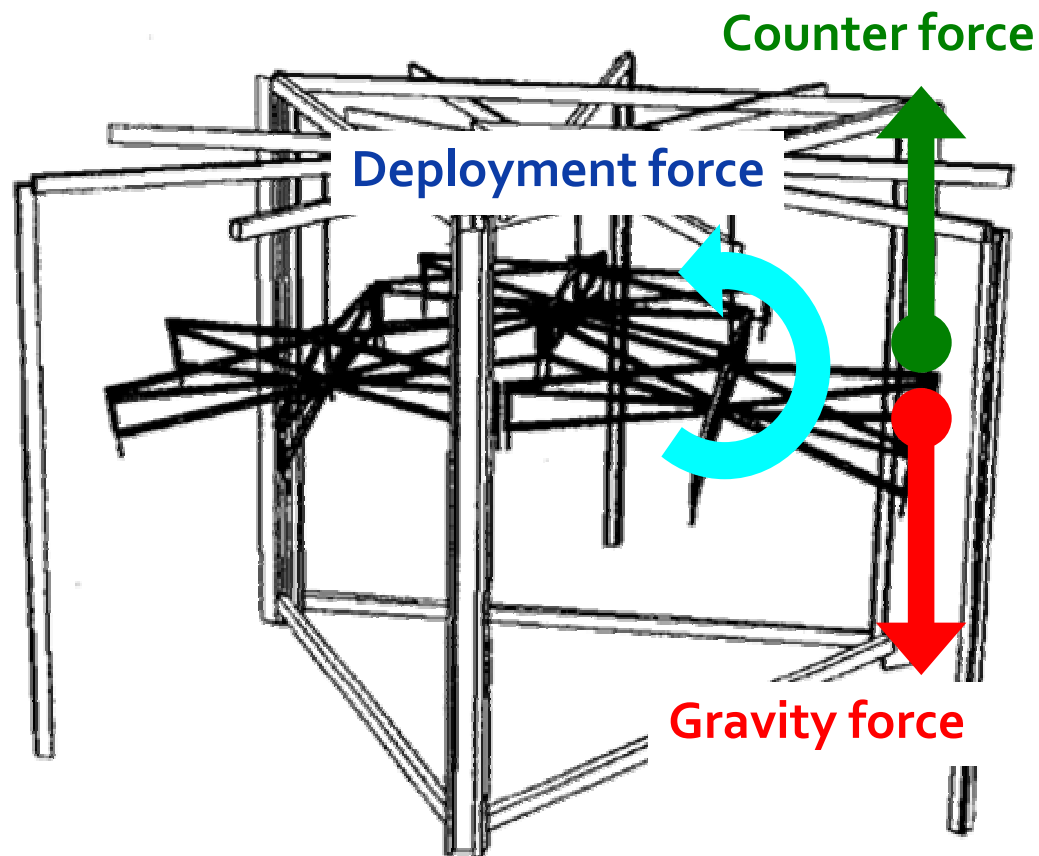


Modular mesh antenna concept



Modular mesh antenna concept

Deployment force vs. Gravity



Large deployable structure

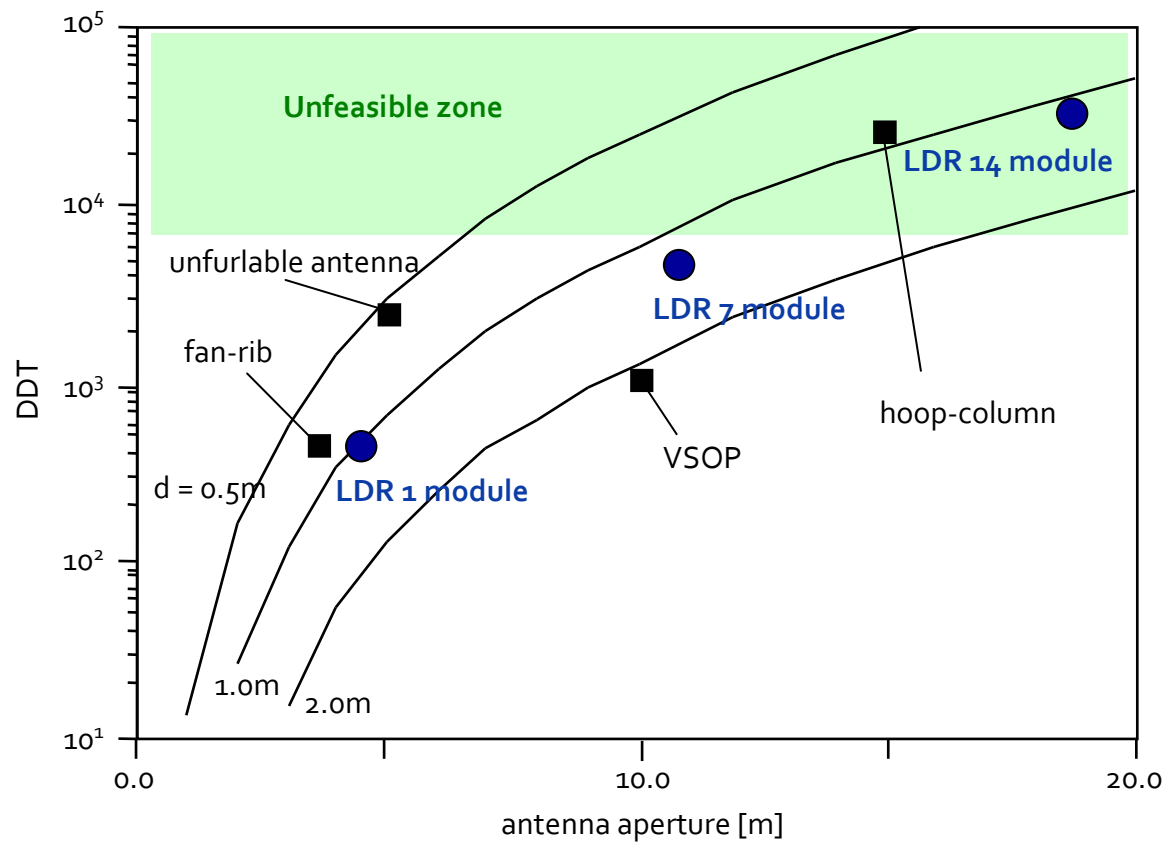


Gravity \gg Deployment force
Errors \gg Deployment Force



Unable to evaluate deployment force

Ground test difficulty evaluation



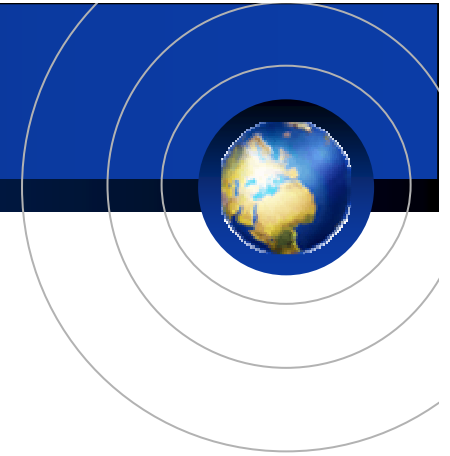
Difficulty of deployment test

$$DDT = \frac{\text{gravity force}}{\text{deployment force}}$$

- Small aperture for test/manufacturing.
- Large aperture for use.

We apply the modular mesh antenna concept.

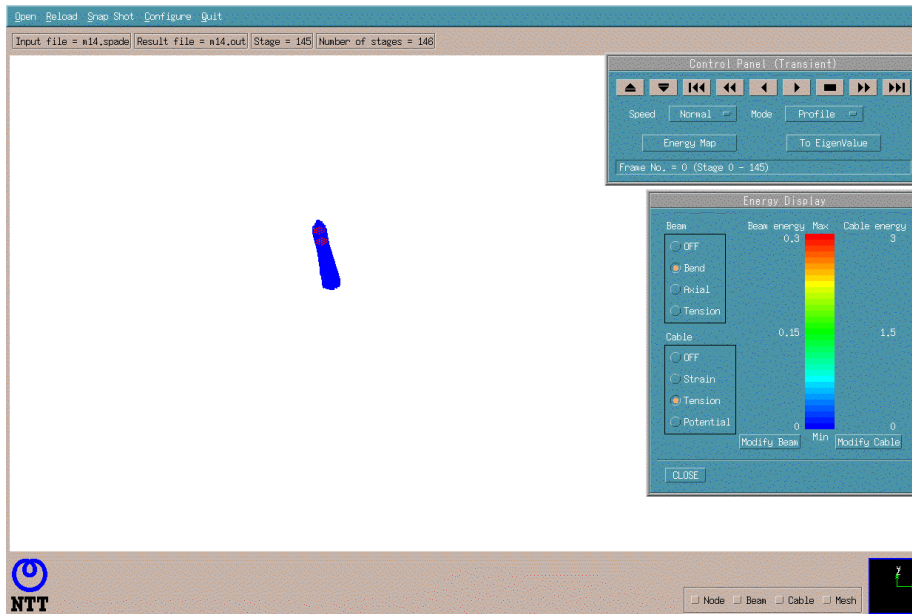
Development strategy of Large Deployable Reflector



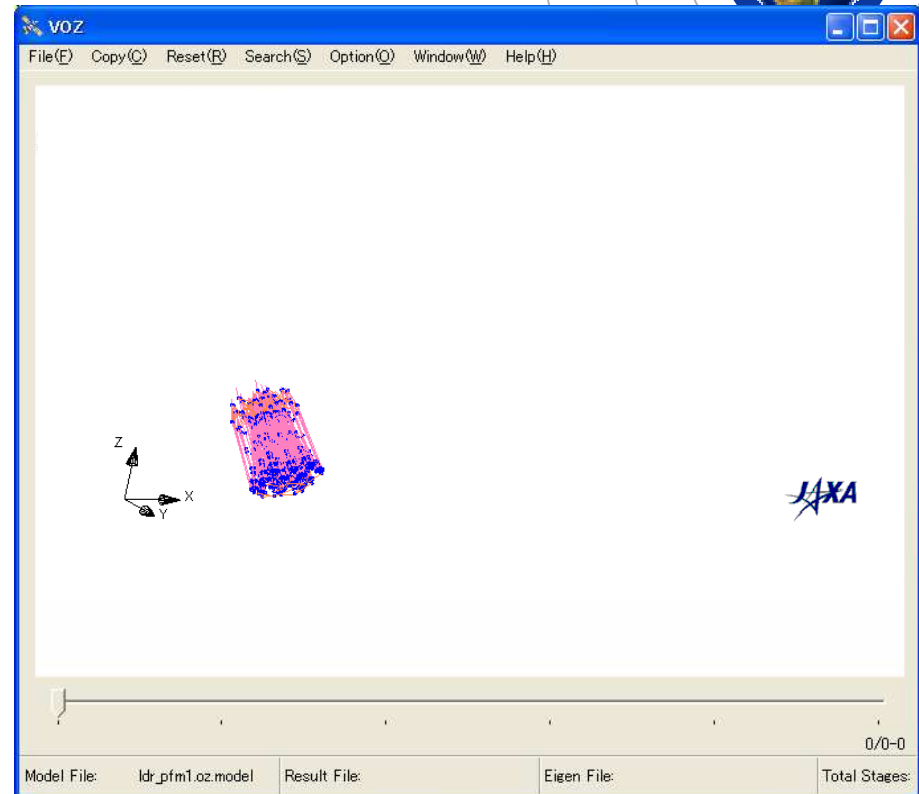
analysis

High precision deployment analysis

Deployment analysis technology for high precision prediction



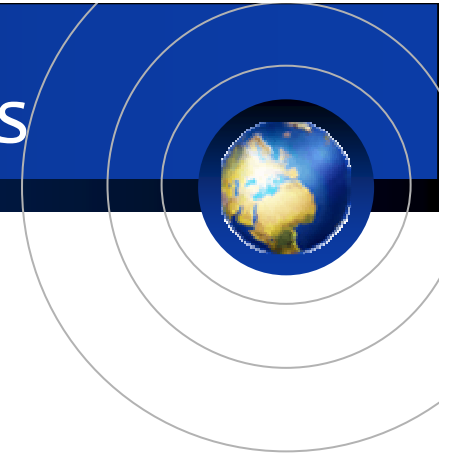
SPADE



Origami/ETS

SPADE was made for **high precision deployment analyses** by NTT (Nippon Telegraph and Telephone Corporation). Currently Origami/ETS inherits its technology and is still being expanded by JAXA. These software contributed to LDR.

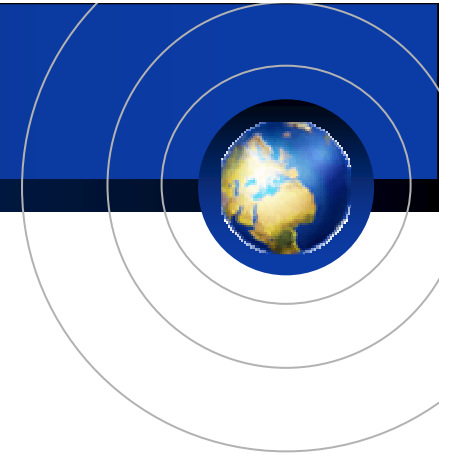
Update analysis model by ground tests



Strategy

- Two main pillars
Module and Analysis
- **Ground tests and Updating analysis model**
- High accurate surface forming
- Analysis precision evaluation by LDREX-2
- Evaluation of Large Deployable Reflector

High accurate surface forming



Strategy

- Two important pillars
Module and Analysis
- Ground tests and Updating analysis model
- **High accurate surface forming**
- Analysis precision evaluation by LDREX-2
- Evaluation of Large Deployable Reflector

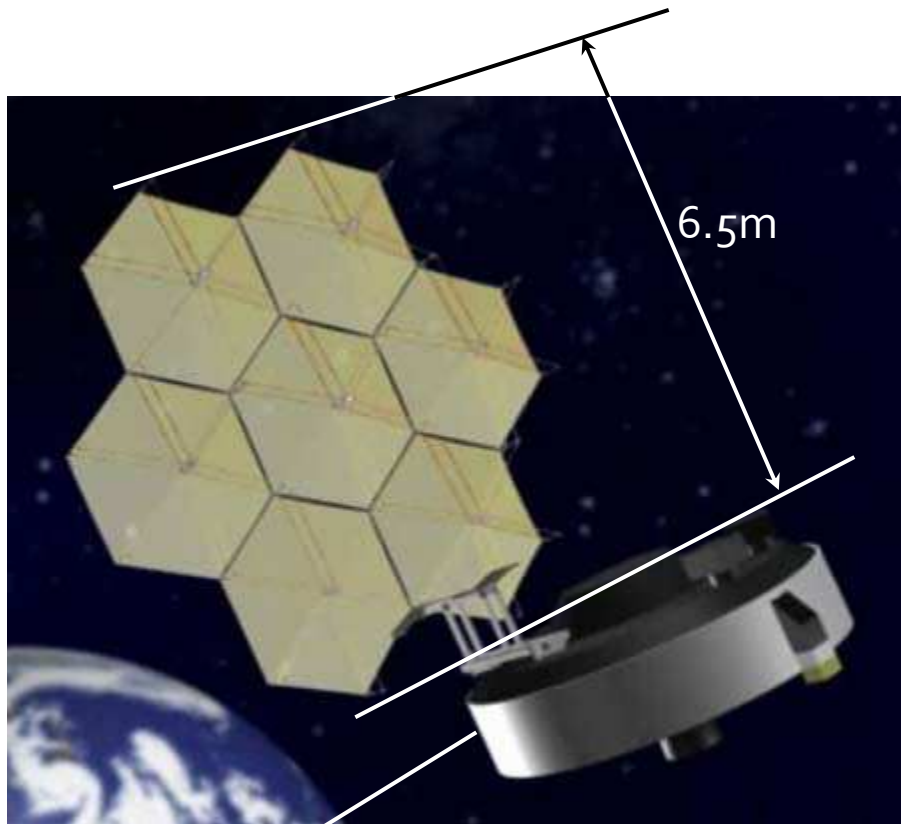
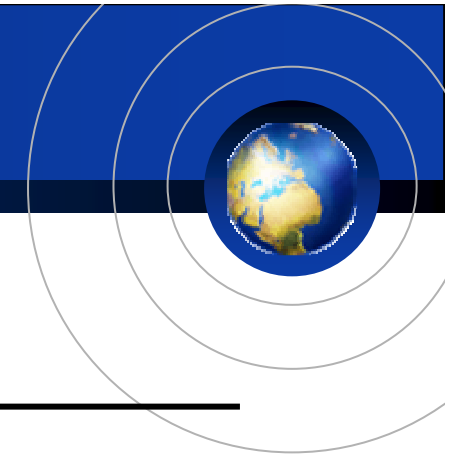
Analysis precision evaluation by LDREX-2



Strategy

- Two main pillars
Module and Analysis
- Ground tests and Updating analysis model
- High accurate surface forming
- **Analysis precision evaluation by LDREX-2**
- Evaluation of Large Deployable Reflector

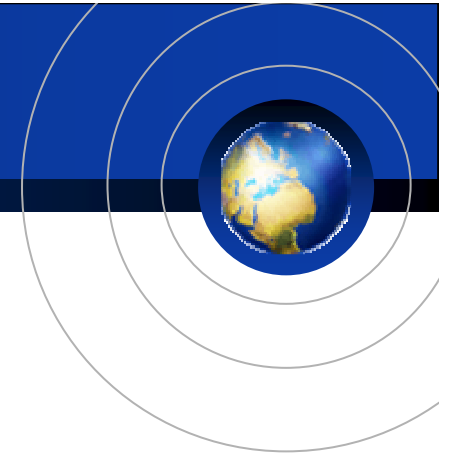
Analysis precision evaluation by the LDREX-2 in-orbit experiment



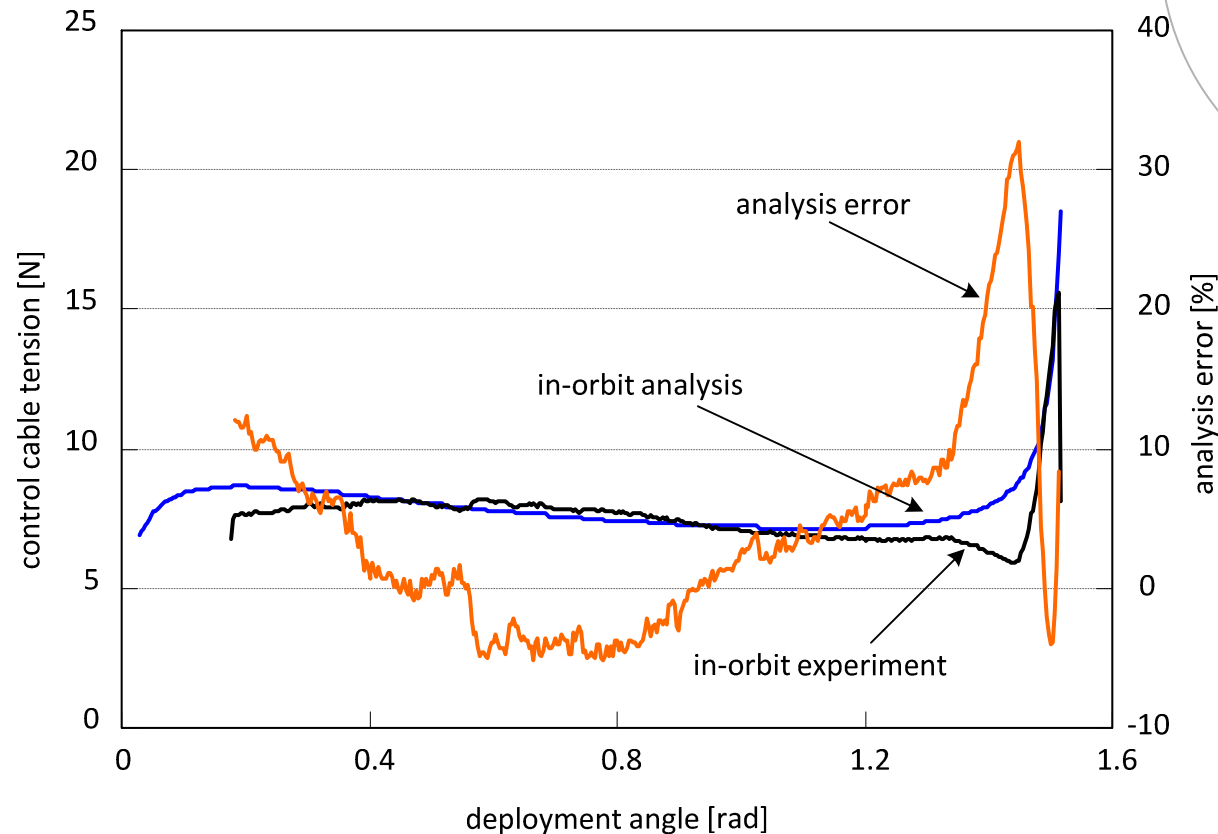
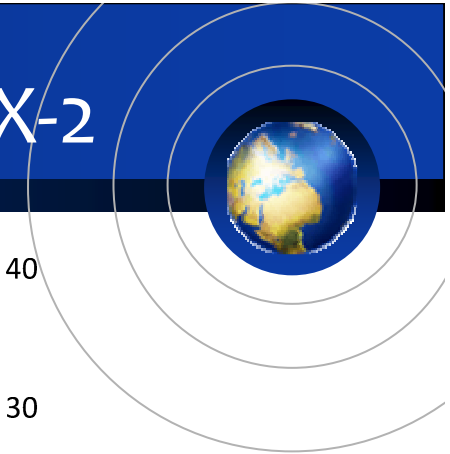
ASAP5
(ARINE5 Structure for Auxiliary Payload)

Mass	211 kg
Size	6.5m x 6.5m
Number of modules	7
Launcher	Ariane 5ECA
Orbit	GTO

LDREX-2 deployment procedure and result



Analysis precision evaluation by LDREX-2



Analysis result agreed with experiment within 10% at almost every angle. The maximum error is 32% right before full deployment. Analysis simulates well and can be used for in-orbit prediction of LDR.

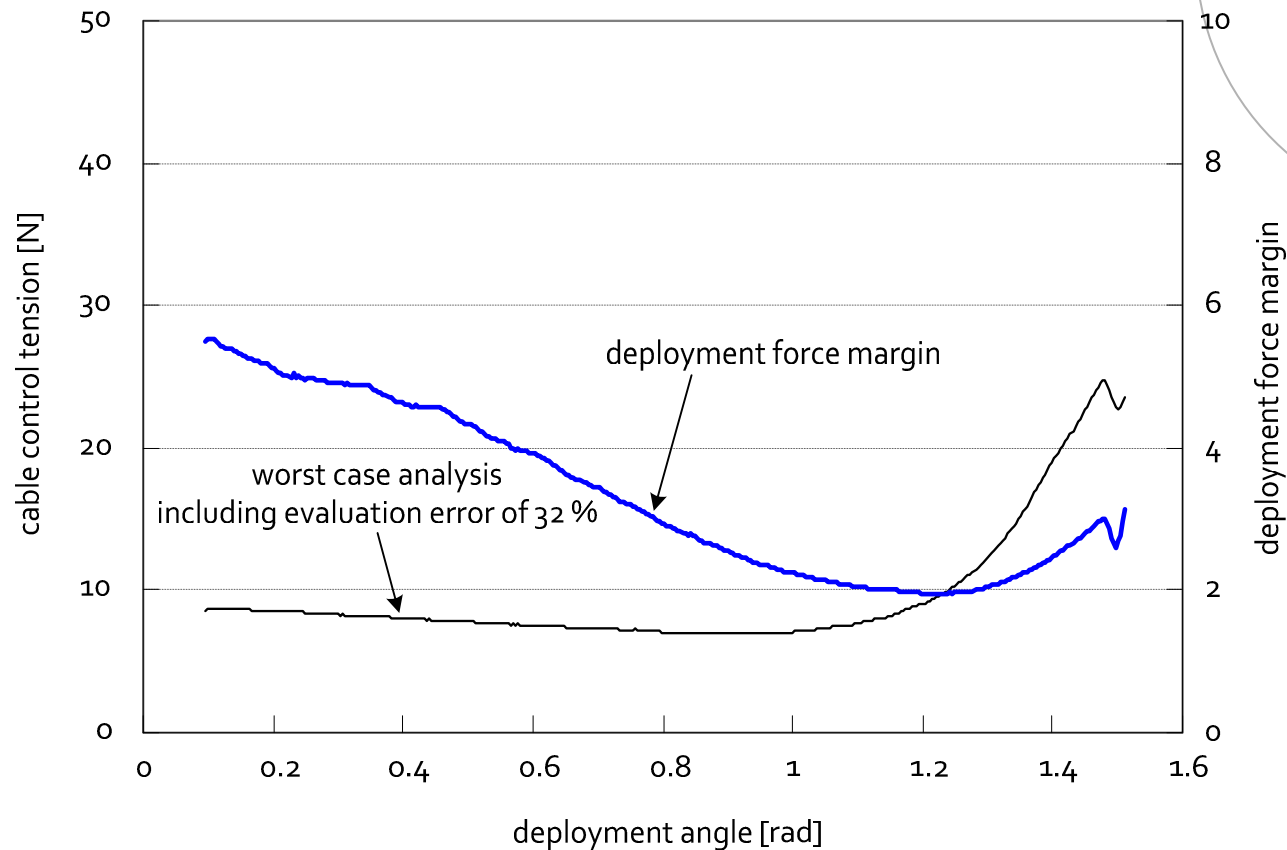
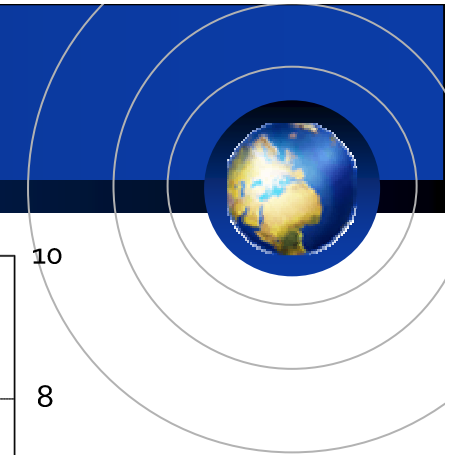
Evaluation of Large Deployable Reflector



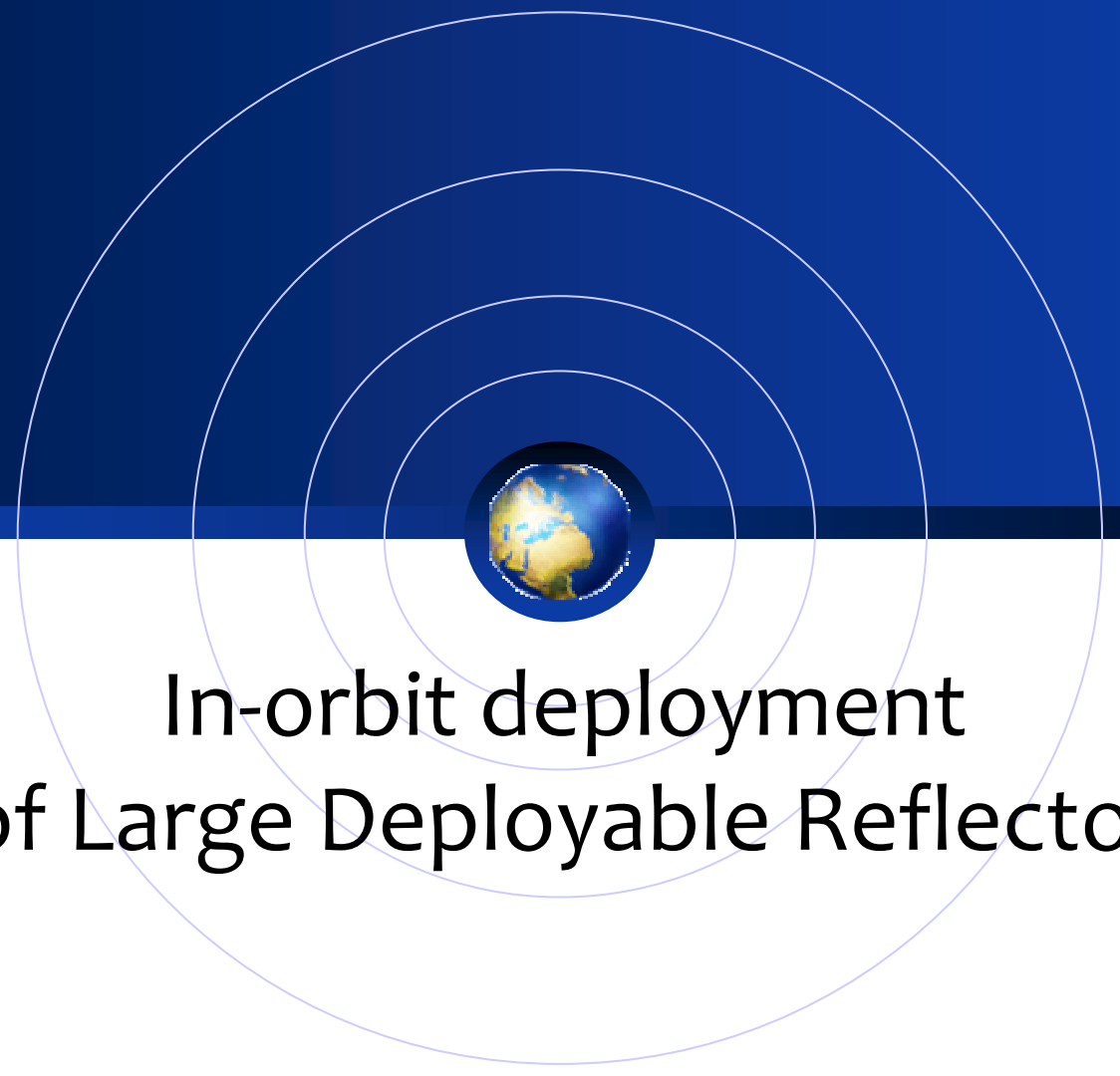
Strategy

- Two main pillars
Module and Analysis
- Ground tests and Updating analysis model
- High accurate surface forming
- Analysis precision evaluation by LDREX-2
- **Evaluation of Large Deployable Reflector**

Worst case analysis results of the Large Deployable Reflector



The evaluation of in-orbit deployment of LDR with maximum 32% error considered shows that the deployment force margin is greater than 2. The **reflector is guaranteed to deploy** through the analysis and LDREX-2.

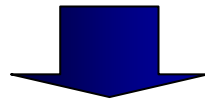
A diagram showing a small globe of Earth at the center, with four concentric white circles representing orbital paths. The background is a solid dark blue, with a horizontal band of lighter blue and black at the bottom, suggesting a horizon or ground level.

In-orbit deployment of Large Deployable Reflector

Launch of Engineering Test Satellite VIII



Final tests

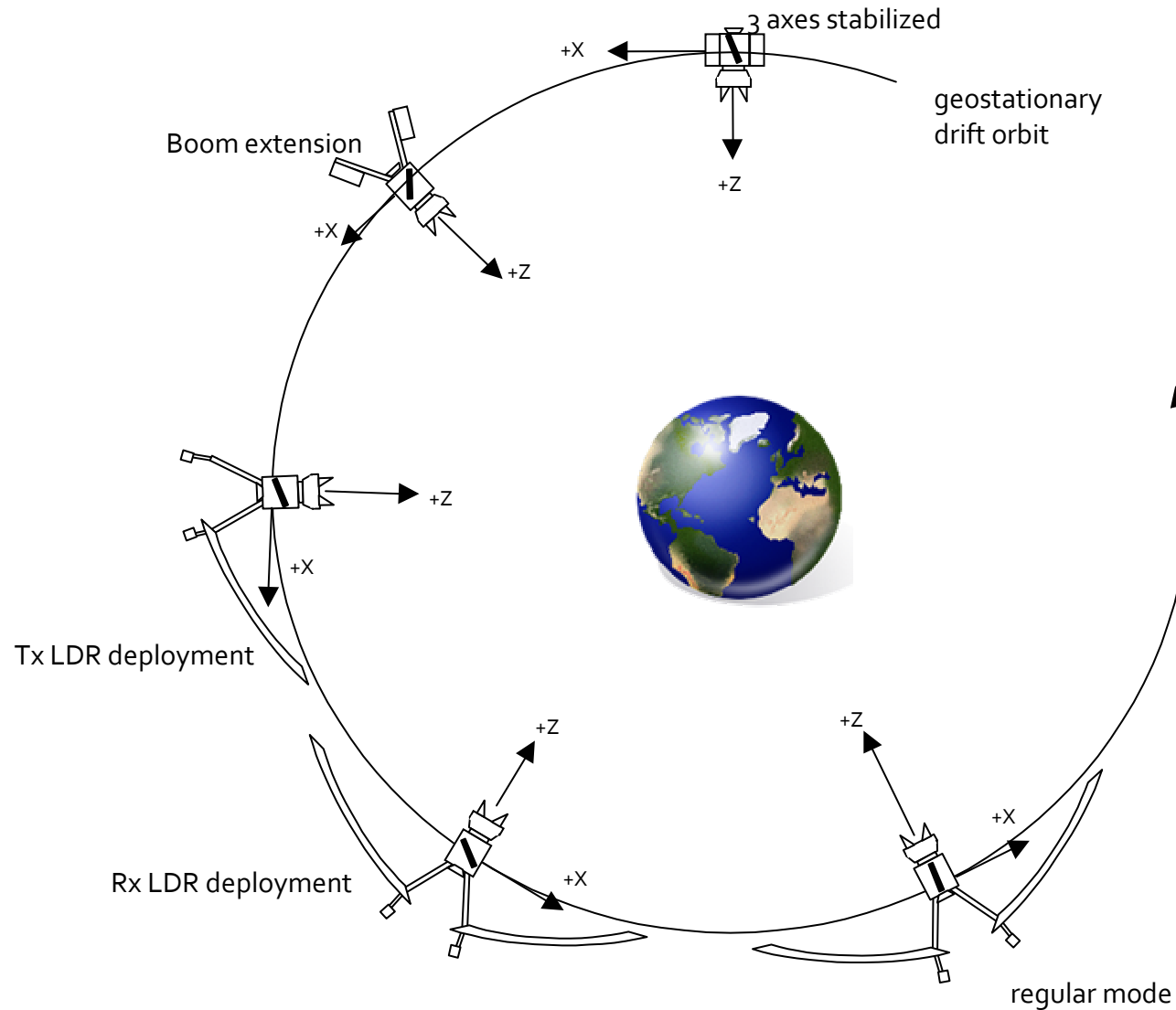


Stacked on fairing

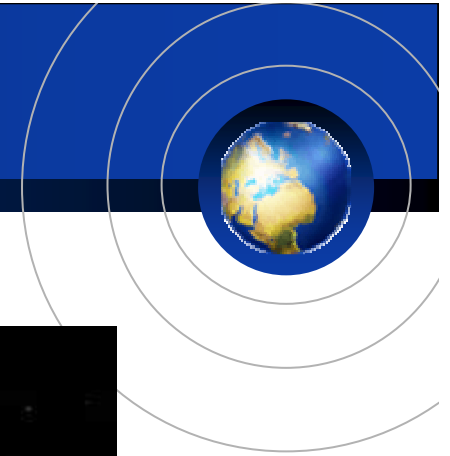


Launched on 18 Dec 2006

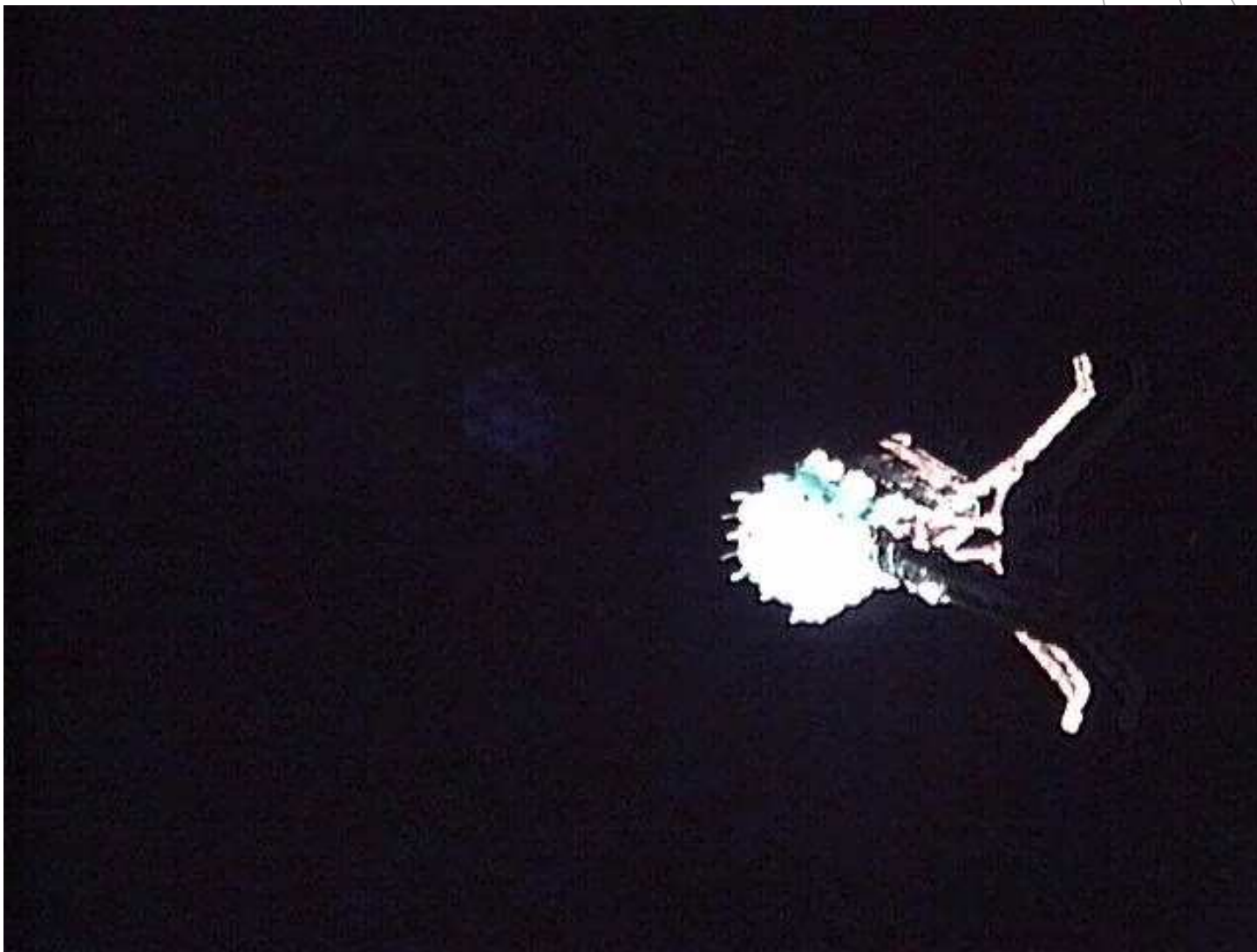
Events of Engineering Test Satellite VIII



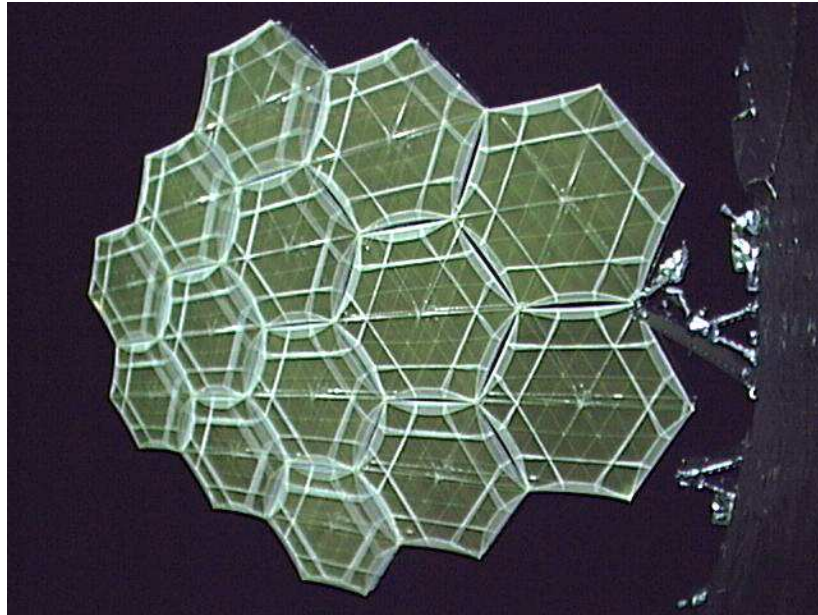
Deployment sequence



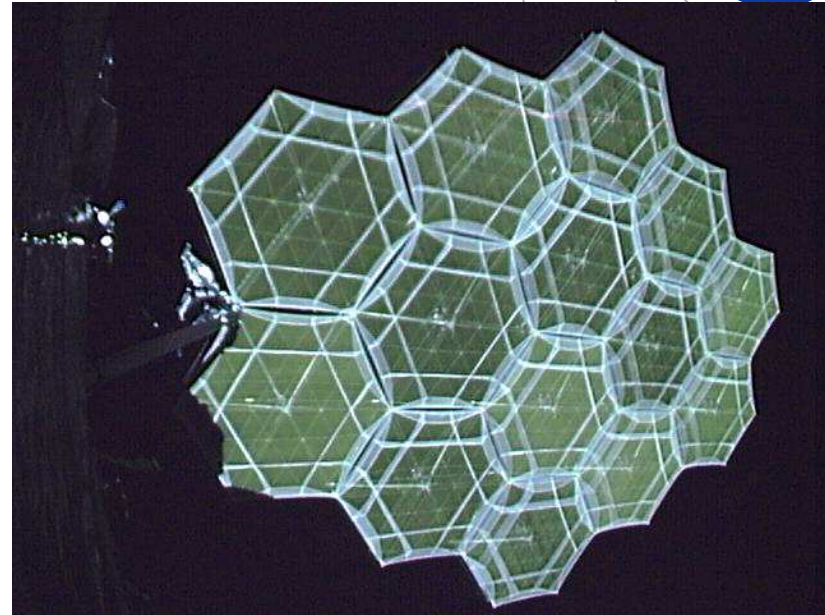
Deployment (1/2)



Deployment (2/2)



Deployed Rx LDR



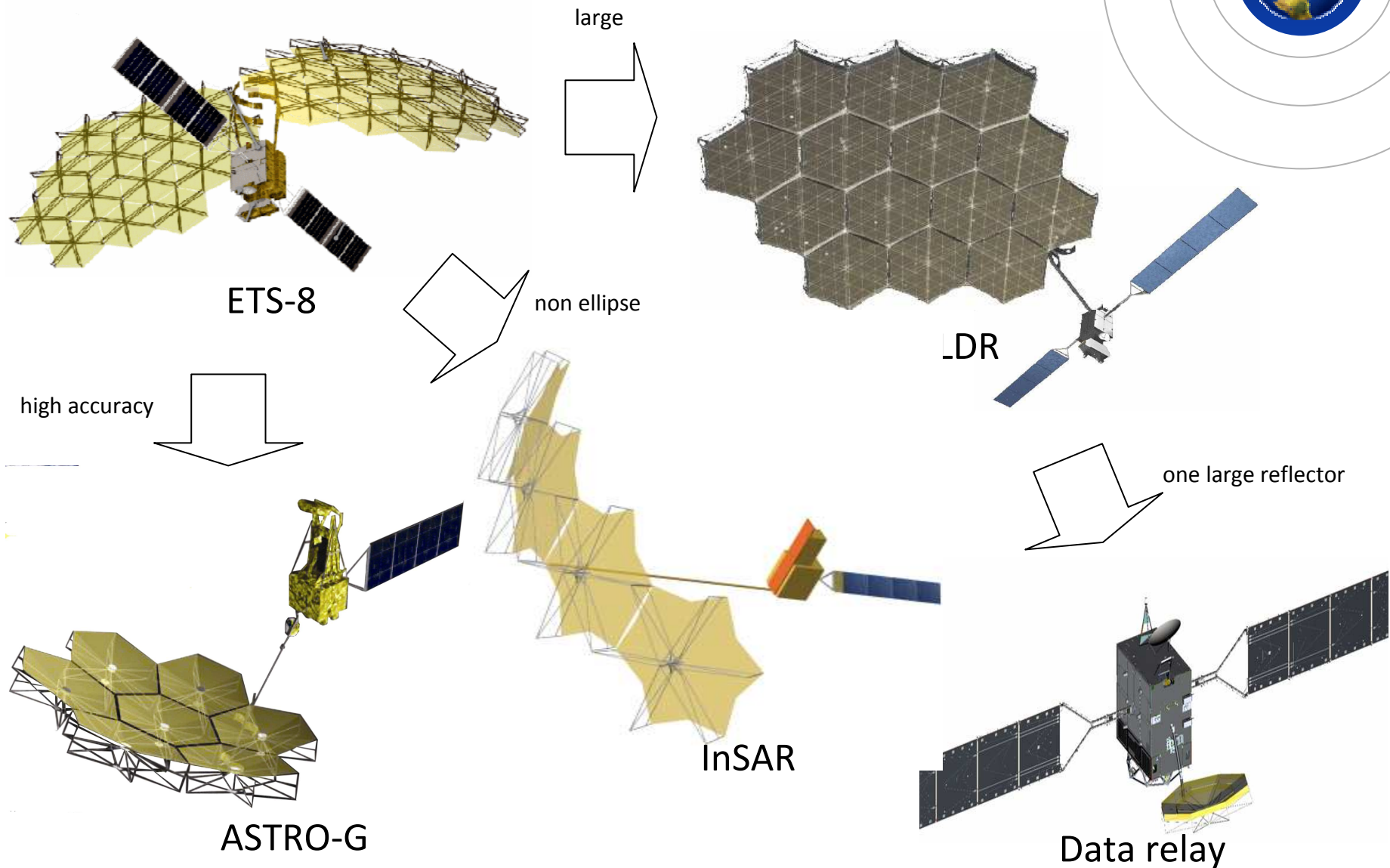
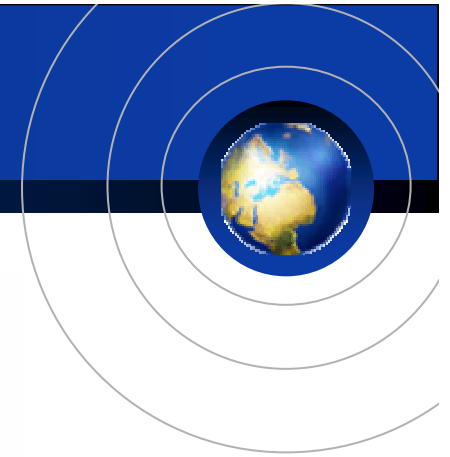
Deployed Tx LDR

The in-orbit deployment of Large Deployable Reflectors demonstrated that the modular mesh antenna concept and the high precision analysis technology are very effective. Furthermore, it is also confirmed that the surface accuracy satisfies the specification against thermal environments throughout the year. These methods can be used for various applications.

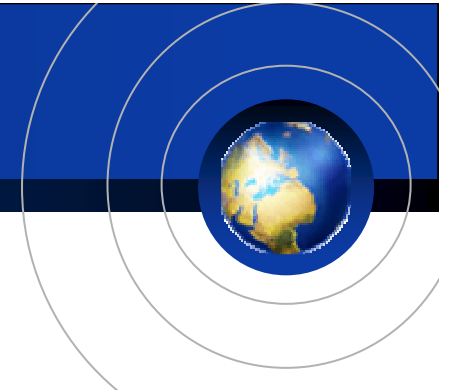


Applications of Large Deployable Reflector

Applications of the Large Deployable Reflector

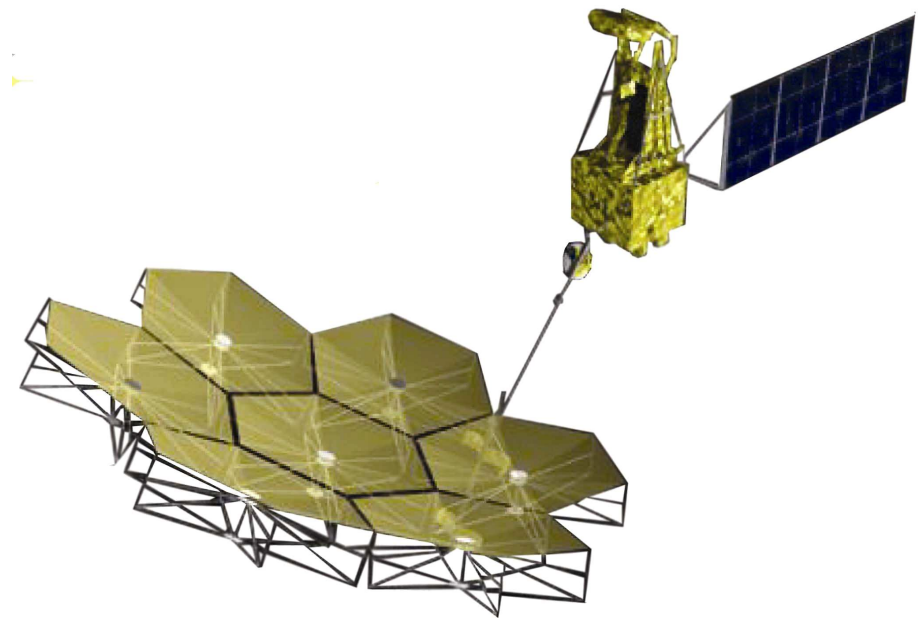


Application to High accuracy reflector



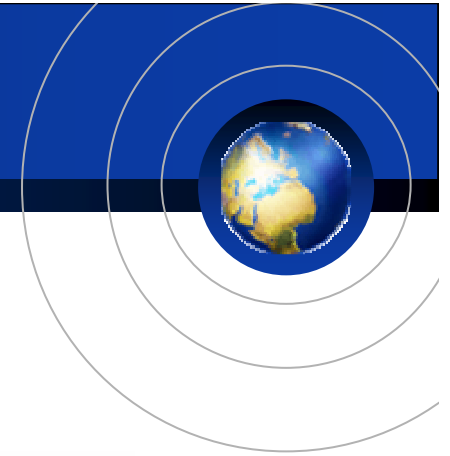
ASTRO-G

- Radio astronomy
- 9.26m aperture
- 0.4mmrms surface accuracy for 43GHz observation



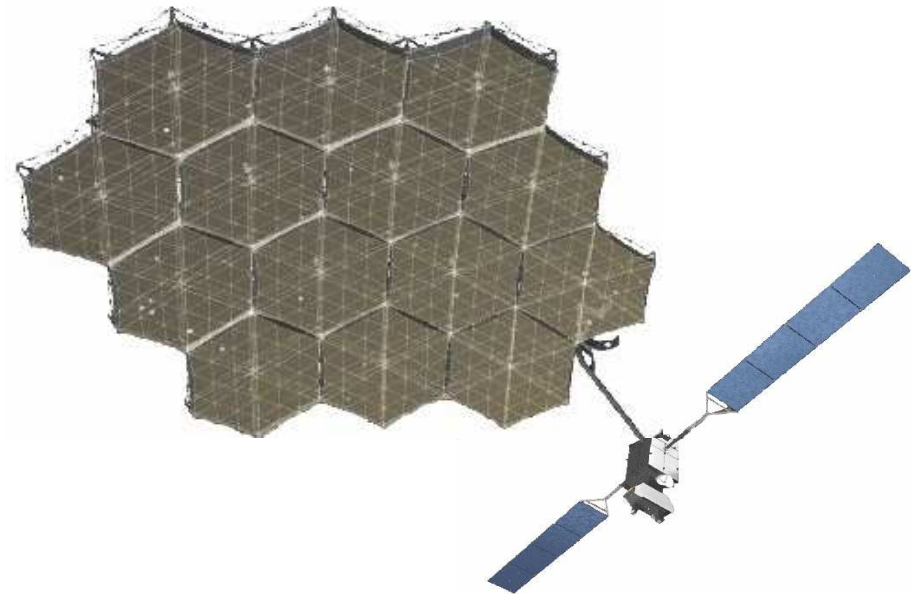
The modular mesh antenna concept can achieve the desired accuracy by the adjustment and measurement of its surface during the manufacture of each module.

Application to Very large reflector



30m class reflector

- Mobile communication
- 30m aperture
- S band

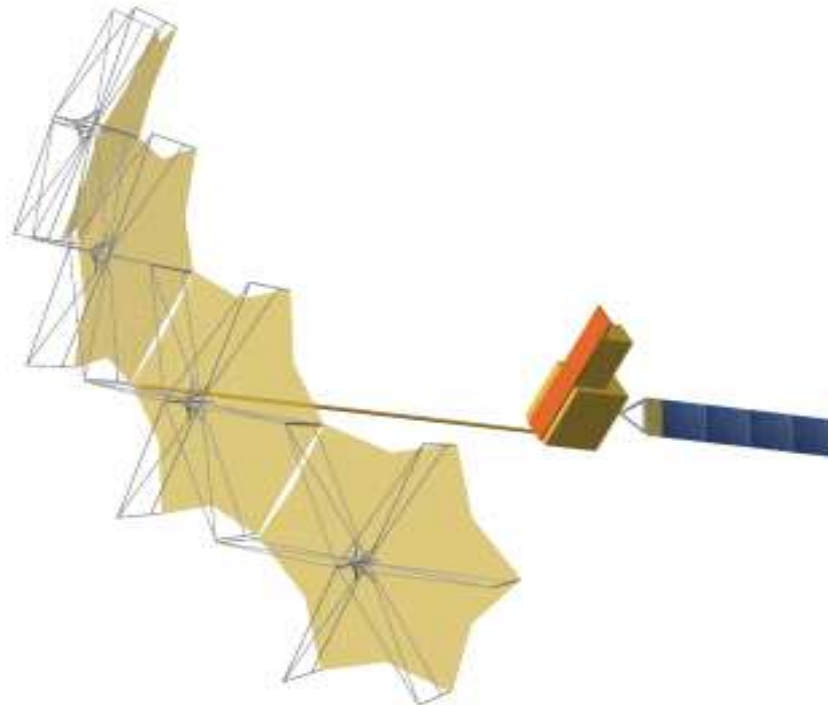


The modular mesh antenna concept can enlarge deployable reflectors by extending each module's aperture or increasing the number of modules.

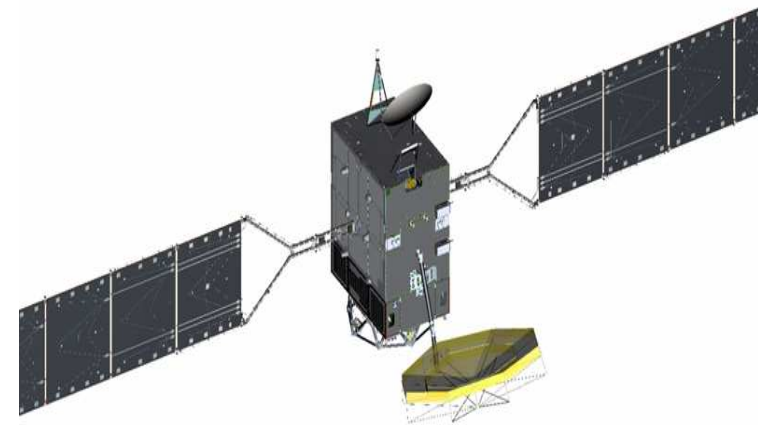
Application to a variety of missions



InSAR



Data relay

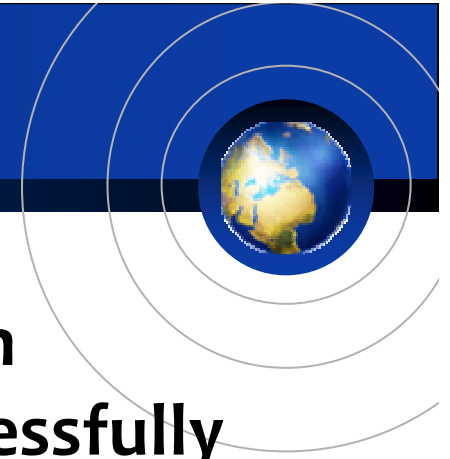


The modular mesh antenna concept can be applied to a huge variety of missions by changing the combination of modules, incorporating high precision modules, increasing the number of modules, and enlarging modules.



Concluding remarks

Concluding remarks



- **Large Deployable Reflectors mounted on Engineering Test Satellite VIII were successfully deployed.**
- **The two important pillars of Large Deployable Reflector are the modular mesh antenna concept and the high precision analysis technology.**
- **The modular mesh antenna concept can be applied to a huge variety of missions, such as radio astronomy missions, interferometric synthetic aperture radars, communications and broadcastings.**

