

Active Spectroscopy

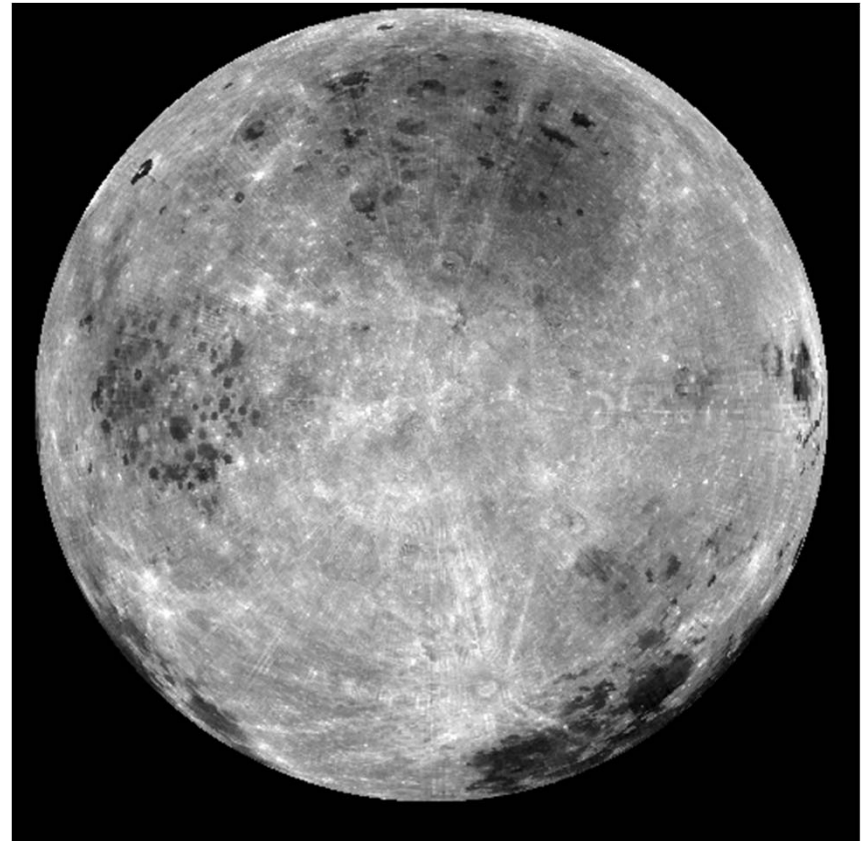
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Three Concepts Covered

- Laser Reflectometer
 - Tunable IR lasers coupled with proven LIDAR/LA technology
- IR Projector
 - High temperature blackbody IR projector provides active illumination of dark surfaces
- Solar Mirror
 - Satellite-based mirror directs sunlight onto polar surfaces

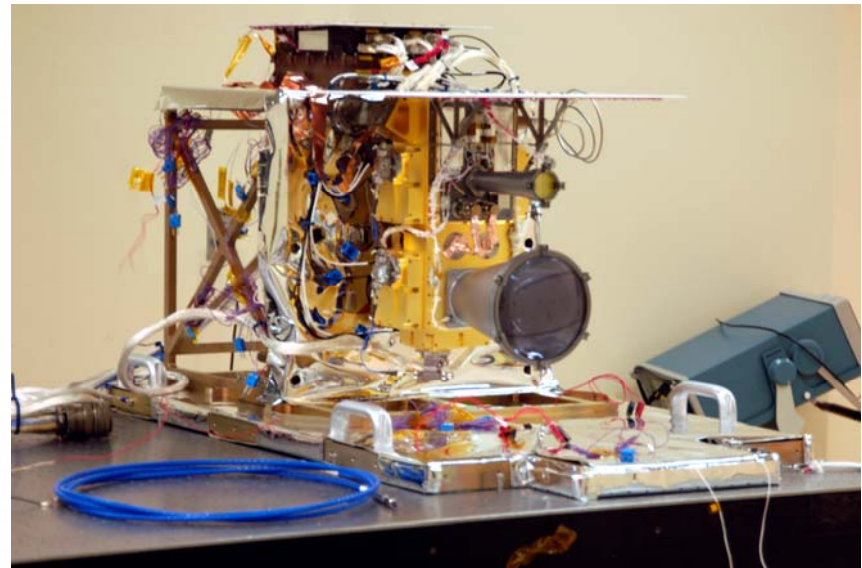
Laser Reflectometer

- Laser Reflectometer
 - Best general science
 - High SNR at km or smaller resolution
 - Global access
 - Conventional SMD-type instrument
 - 20kg, 25 W, \$30M (Class B)
 - Likely competitive in existing technology development and mission opportunities
 - Goddard supporting on IRAD
- UV possible, worth KISS study



Basic Parameters

Table 2. Signal and Noise Input parameters	
Transmit power	2.5 mJ/pulse
Receiver diameter; f-no.	160 mm; 4
SNR/Pulse	90 @ 100 m samples
System transmission	10%
Lunar background radiance	4 watts/m ² -um-sr; 10 watts/m ² -um-sr assumed
Detector size	64 microns
Bandpass filter	0.9 microns (2.65-3.55)
System read noise	560 e-
APD gain	1-900 (~5 will be used)
Net integration; sample time	45 ns; 2 ns
Dark current	3×10^{-10} A/cm ²
Nominal range	30 km
Quantum Efficiency	0.7



Polar Relevant Irradiance Sources

W/m²*

- Sunlight • $1.3 \cdot 10^3$
- Scattered moonlight, 70K** • $2.0 \cdot 10^0$
- **8 m reflector with imperfections** • $0.3 \cdot 10^0$
- **2.5 m reflector at 30 km** • $1.3 \cdot 10^{-1}$
- **1 m reflector at 30 km** • $1.9 \cdot 10^{-2}$
- Full Moon (from earth) • $2.1 \cdot 10^{-3}$
- **IR Projector** • $5.0 \cdot 10^{-5}$
- Integrated starlight • $3.0 \cdot 10^{-5}$
- Bright planets • $2.0 \cdot 10^{-6}$
- Zodiacal light • $1.2 \cdot 10^{-7}$
- Airglow • $5.1 \cdot 10^{-8}$
- Diffuse galactic light • $9.1 \cdot 10^{-9}$
- Cosmic light • $9.1 \cdot 10^{-10}$

*Integrated spectral radiance, different situation at 3um depending on source T

**Flux causes 70K equilibrium temperature

Polar Relevant Irradiance Sources: Issues

- Sunlight
 - Earthlight
 - Moonlight
 - Bright planets
 - Zodiacal light
 - Integrated starlight
 - Airglow
 - Diffuse galactic light
 - Cosmic light
- | |
|---|
| <ul style="list-style-type: none">• Water colored!• 3 um band! |
|---|

IR Projector, The Good News

Projector

- 10 cm diameter unobscured optic
- f/2, 0.2 sr
- 5mm² 3000K blackbody
 - @ 3um, 1.5e5 watts/m²-um-sr
 - Area: 2.5e-5 m²
- At 3um, # watts/um exit projector,
 - radiance*solid angle*area
 - 0.6 watts/um (no losses)
- Focal length projector 200mm
- Spot at 25 km is 280 m
 - 62,000 m²
- Irradiance on spot per unit bandpass:
 - 9.6 microwatts/um-m²
- Radiance of spot
 - Irradiance times albedo/pi
 - 1.5 x 10⁻⁶/um-m²-sr
- Thermal emission
 - negligible in the cold places
 - Equivalent to the emission by a 180 K bb at 3 um)

Receiver

- 10 cm diameter unobscured
- f/2, 200 mm focal length
- Detector 5mm²
- A-omega: 2.5e-5 m²*.2sr
 - .5 e -5 m²-sr
- Power/micron collected:
 - Radiance times A-omega
 - 1.5x10⁻⁶ watts/m²-um-sr *.5 e -5 m²-sr
 - 7.5 x 10⁻¹² watts/um
 - 1.1 x 10⁸ ph/s-um
- Assume 0.1 total transmission
- Assume 300 nm bandpass
- 3e6 ph/second
- 1 km ground sample, 400 ms integration
- 0.95x10⁶ electrons produced
- Neglecting other noise sources
 - SNR=970

IR Projector, Potential Problems

- Irradiance on detector is very small
- **.007** nA/cm² photocurrent
- 3 μ m cutoff HgCdTe has dark current **0.02** nA/cm² at 77K
- 2.5 micron cutoff very low dark current HgCdTe has 0.1nA/cm² at 120K, .01nA/cm² at 90K
- **DARK CURRENT IS A MAJOR POTENTIAL PROBLEM at 3 μ m**
- **2 micron region is a bit brighter and detectors have lower dark current**
 - SNR requirement stiffer
- **At 1.4 micron ice band detectors have negligible dark current**
- **Need to study thermal detectors**
- **Need to study all kinds of detectors**
- Scattered moonlight is common (unavoidable?) and v large

Solar Mirror

1-m mirror

- At 3 μm , solar radiance is 0.4×10^6 watts/ $\text{m}^2\text{-}\mu\text{m}\text{-sr}$
- Radiance of surface: reflectance radiance * solid angle of sun/ π
 - 3 watts/ $\text{m}^2\text{-}\mu\text{m}\text{-sr}$
- 1-m mirror at 30 km gives 4×10^{-5} watts/ $\text{m}^2\text{-}\mu\text{m}\text{-sr}$, @3 μm
 - Projector is 1.5×10^{-6}
- 8-m mirror with $\frac{1}{2}$ degree imperfections
 - 16x the flux of perfect 1-m
 - Allows increase of resolution to $\sim 20\text{nm}$

Receiver

- 10 cm diameter unobscured
- f/2, 200 mm focal length
- Detector 5mm^2
- A- ω : $2.5 \times 10^{-5} \text{m}^2 \cdot .2\text{sr}$
 - $.5 \times 10^{-5} \text{m}^2\text{-sr}$
- Assume 0.1 total transmission
- Assume 300 nm bandpass
- Reasonable losses, 90%
- 7.5×10^7 ph/second
- .1 km ground sample, 40 ms integration
- 2.5×10^6 electrons produced
- Neglecting other noise sources
 - SNR=1580
- Photocurrent is 10x dark current
- Tempting to go to higher spectral resolution but dark current becomes important again
- There appears to be ample margin however

Mirror issues

- Can get higher radiances with actual focus
 - Relay Mirror Experiment (RME) demonstrated extremely long focal length focus
 - 1 meter mirror, 400km focal length!
- Pointing is a bigger challenge
- Meter class high quality mirror is some kind of a challenge
- Inflatable?
- Are solar sail technologies flat enough?
- Scattered moonlight is common (unavoidable?) and large

Starlight!

- Aggressive passive experiment
- Turner, R.E., Night Sky Spectral Radiance Models, Contract DAABO7-98-D-H752, CECOM Night Vision and Electronic Sensors Directorate, Ft. Belvoir, VA, April 2001
- Distribution may be variable owing to distribution of stars
 - See LAMP publications
- Roughly solar color, 3 μ m calculations based on solar scaled to two microns
- Irradiance calculated assuming a hemisphere (pi steradians projected area)
- 6×10^{-7} watts/m²- μ m-sr
 - Projector: 1.5×10^{-6} / μ m-m²-sr
 - Mirror: 4×10^{-5} watts/m²- μ m-sr

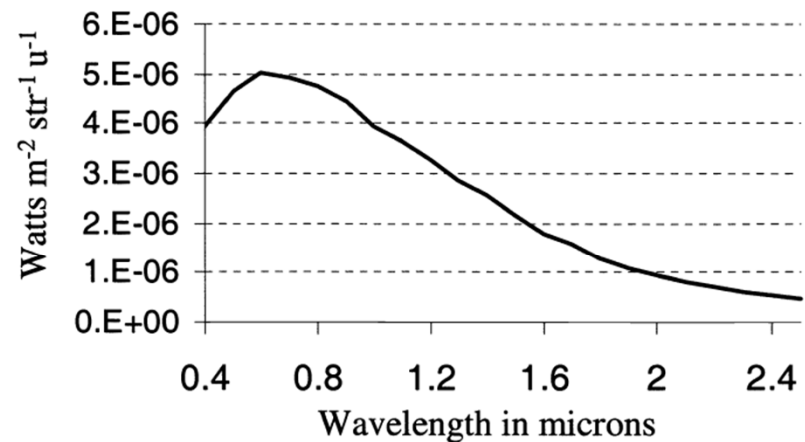


Figure 4. The total stellar spectral radiance including interstellar absorption.

Starlight

Receiver

- 10 cm diameter unobscured
 - f/2, 200 mm focal length
 - Detector 5mm^2
 - $A\text{-}\omega: 2.5 \times 10^{-5} \text{ m}^2 \cdot .2\text{sr}$
 - $.5 \times 10^{-5} \text{ m}^2\text{-sr}$
 - Power/micron collected:
 - Radiance times $A\text{-}\omega$
 - $6 \times 10^{-7} \text{ watts/m}^2\text{-}\mu\text{m}\text{-sr} \cdot .5 \text{ e}^{-5} \text{ m}^2\text{-sr}$
 - $3 \times 10^{-12} \text{ watts}/\mu\text{m}$
 - $.4 \times 10^8 \text{ ph/s-}\mu\text{m}$
- Assume 0.1 total transmission
 - Assume 300 nm bandpass
 - 1.2×10^6 ph/second
 - 2.5 km ground sample, 1000 ms integration
 - 1.2×10^6 electrons produced
 - Neglecting other noise sources
 - SNR=1100

Starlight, The Potential Problem

- .003 nA/cm²—this is very small
- 3 μ m cutoff HgCdTe has 0.02nA/cm²
- DARK CURRENT IS A POTENTIAL PROBLEM at 3 μ m
- Less of a problem at shorter wavelengths
- Scattered moonlight is common (unavoidable?) and large

Conclusions I

- Laser spectrometer high science, relatively mature, Goddard IRAD in progress, proposals submitted
- UV measurements feasible, possible study topic, instrument would be similar to LOLA

Conclusions II

- IR Searchlight
 - Flexible concept of operation
 - Power consumptive
 - Very challenging detector engineering problem
 - Low intensity on focal plane
 - Dark current competes
 - Potential game changer
 - Worth detailed engineering study
 - Better at 3 μ m
 - Global science possible
 - Stray moonlight!
- Starlight
 - Similar to projector in detection challenge but a bit worse
 - Least complex
 - Data available globally
 - Better at shorter wavelengths
 - Global science possible
 - Stray moonlight!
- Solar reflector
 - Better detection problem than projector
 - Probably within uncertainty due to engineering cleverness
 - Signal depends critically upon mirror quality
 - Large errors permissible with a typical optical system, but requires a much flatter surface than a solar sail
 - Stretched membrane, inflatable may be options
 - Pointing requirements small fraction of a degree
 - Better at shorter wavelengths
 - Science at poles, possible non-polar science near terminator
 - Stray moonlight

Conclusions III

- Receiver problem roughly similar for all three approaches
- Signal levels may diverge on further study but at the outset all approaches can leverage the same receiver
- Stray moonlight needs to be mapped and modeled to see where small enough to ignore
- Mirror may provide most signal, largest spacecraft operations challenge
 - Mirror quality v size
 - Pointing
- Projector may be competitive, especially at 3um
 - Study needs to optimize performance/power with receiver design
- Starlight intensity needs validation
 - Perhaps LAMP team

Summary

	Pros	Cons	Recommendation
Laser Reflectometer	Diverse high value measurements	Expensive, heavy, evolutionary not revolutionary, reasonable prospect of NASA conventional funding; Goddard IRAD underway	Investigate UV bands
IR Projector	Flexible concept of operation, potentially revolutionary, global science	Engineering challenge for detection, competes with starlight complicating data analysis; power consumptive	Study detection problem; projector design including power
Solar mirror	Simple detection problem, leverages solar sail technology, potentially revolutionary	Tricky spacecraft control requirements; mirror quality; , local science	Study control, mirror technology problem, detector design
Starlight	Least complex, global science	Very challenging receiver	Study receiver, especially dark current