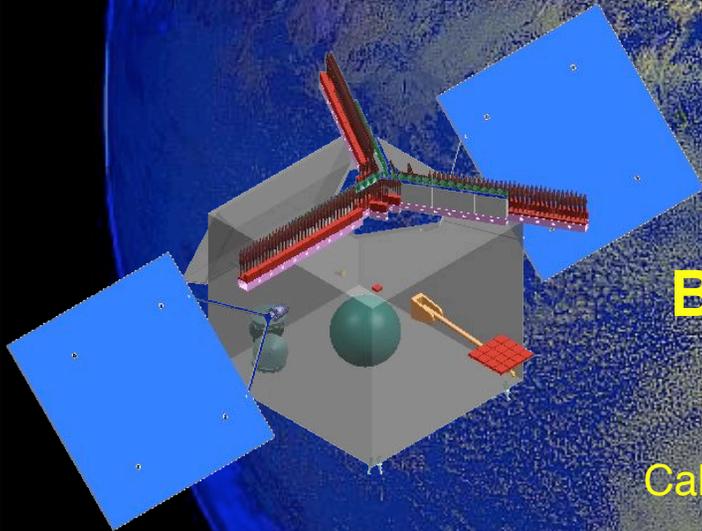




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# Earth Remote Sensing Applications with Microwave Array Spectrometers



**Bjorn Lambrigtsen**

Jet Propulsion Laboratory  
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KISS MMIC Workshop  
Caltech, Pasadena, July 21-25 2008



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# Earth Science Applications

## Soil moisture; ocean salinity $\Rightarrow$ Climate

- Aircraft (5-10 km):  $\sim 180^\circ$
- LEO satellites (700-1000 km):  $\sim 130^\circ$
- L-band (1.4 GHz, 21 cm)
- Surface emissivity

*ESTAR (1990), 2DSTAR (2004)*  
*SMOS (2009)*

## Hurricane wind speed $\Rightarrow$ Weather

- Aircraft (5-10 km):  $\sim 180^\circ$
- C-band (4-7 GHz, 4-7 cm)
- Surface emissivity (wind-induced foam)

*HIRAD (2010)*

## Rain mapping $\Rightarrow$ Weather

- LEO satellites (400-800 km):  $140^\circ$
- X-band (10.7 GHz, 3 cm)
- Thermal absorption/emission of liquid raindrops

*LRR (2006, aircraft)*

## Atmospheric sounding $\Rightarrow$ Weather

- MEO/GEO satellites (10000-36000 km):  $45^\circ/17^\circ$
- V-band & up (50-200 GHz, 1.5-6 mm)
- Thermal absorption/emission of oxygen (temperature), water vapor & liquid

*GeoSTAR (2016)*

*Generally very large sources,  $T_b \sim 100-300$  K; require NEDT  $\sim 0.1-1$  K*



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# Why Array Radiometers?

- **Spatial resolution & coverage!**
  - For applications that require relatively very large apertures
  - Difficult to build, deploy, scan large parabolic reflectors
  - In some cases better NEDT can be attained with array radiometers
- **Array types**
  - Focal-plane arrays: not much used in Earth applications so far
    - Can be used to improve spatial-coverage/integration-time in scanned systems
  - Phased arrays (beam synthesis): not used in Earth applications
    - Earth applications require very high “beam” quality  $\Rightarrow$  Filled array  $\Rightarrow$  Costly
  - STAR arrays (aperture synthesis): dominant in Earth applications
    - Enables large effective aperture using few elements
- **1-D STAR array**
  - Simple; requires few elements & small correlator
  - Can be used if the scene is moving
- **2-D STAR array**
  - Complex; requires large number of elements & very large correlator
  - Must be used if the scene is stationary
  - May be used with moving scene to improve effective NEDT (multiple looks)

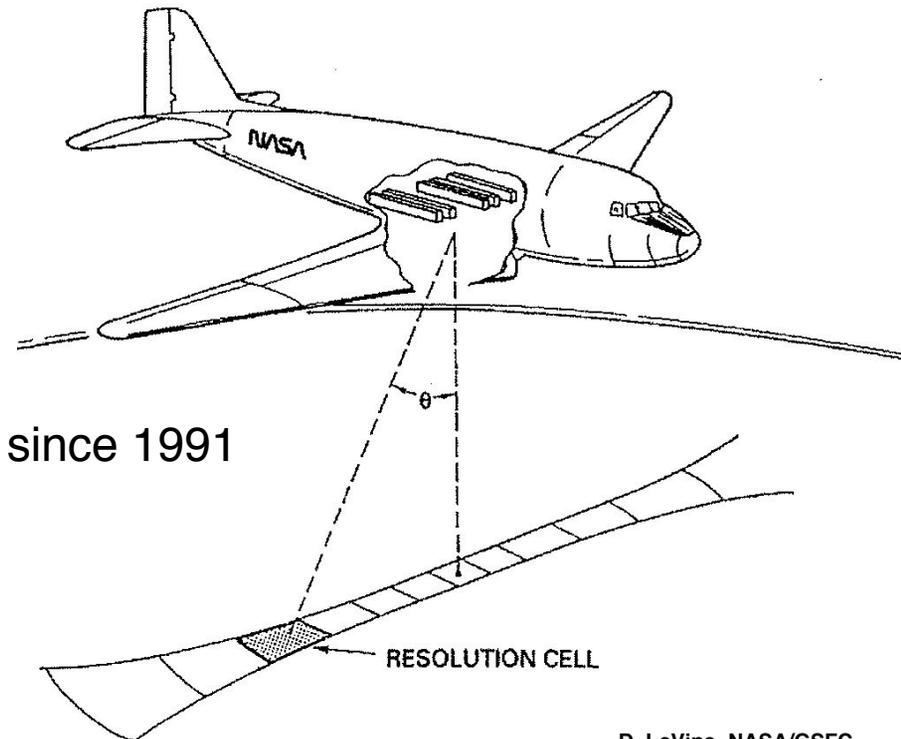


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# First: ESTAR

- 1-D sparse array @ 1.4 GHz
  - 5 slotted-waveguide antennas  
⇒ 5 fan beams
  - Dual polarization
  - Min. baseline =  $0.5 \lambda$   
⇒ no aliasing in  $180^\circ$  FOV
- Flown in numerous field campaigns since 1991
  - Soil moisture



D. LeVine, NASA/GSFC



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# Recent 1-D Array: LRR



10.6 GHz, 14 elements  
Precipitation (prototype)

C. Ruf, U. Michigan



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# VLA: Model for 2-D STARs



First Y-configuration

<http://www.vla.nrao.edu/>

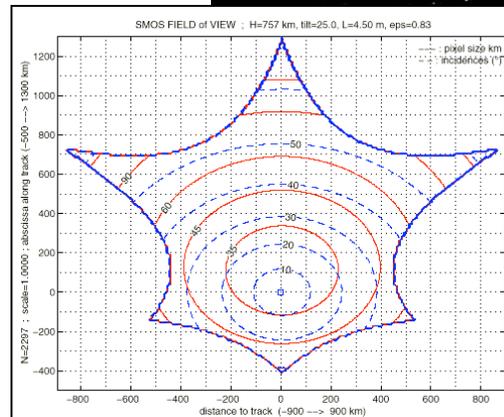
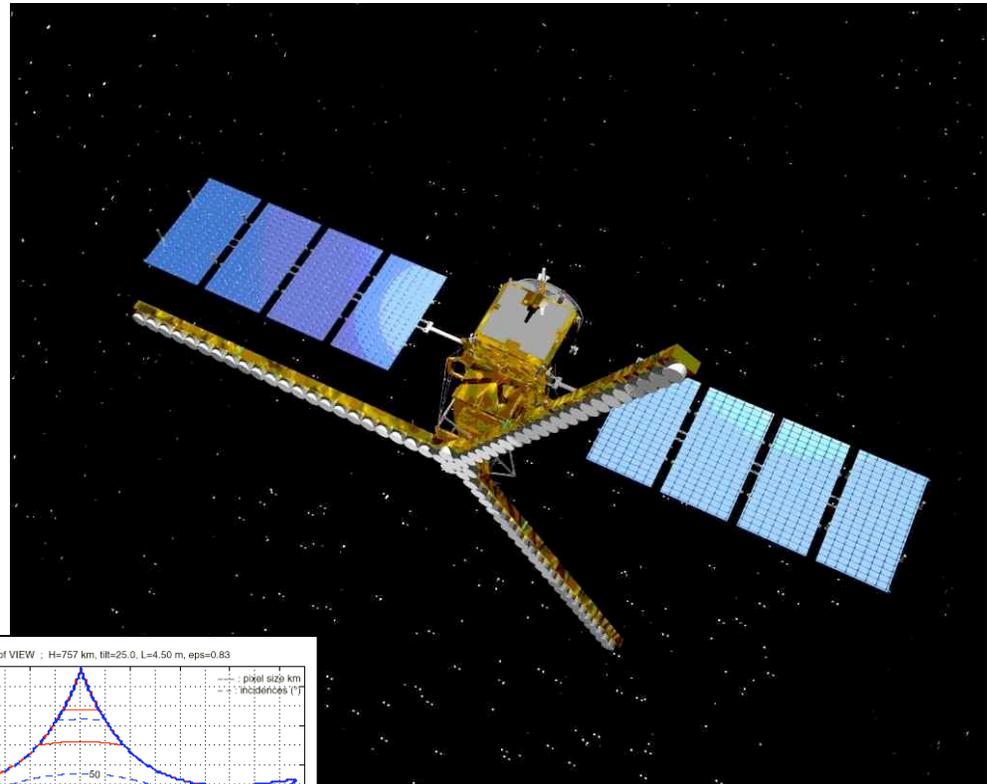


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# Next year: SMOS

- 2-D sparse Y-array @ 1.4 GHz
  - 73 receiver elements
  - 9-meter effective aperture
  - Dual polarization
  - Min. baseline =  $0.9 \lambda$ 
    - ⇒ aliasing limits FOV
      - Scene  $\approx 130^\circ \Leftrightarrow \Delta \approx 0.65 \lambda$
      - $\Delta \approx 0.9 \lambda \Rightarrow 80^\circ$  alias-free
- To be launched by ESA in 2009
  - Soil moisture & ocean salinity
  - Difficult data processing



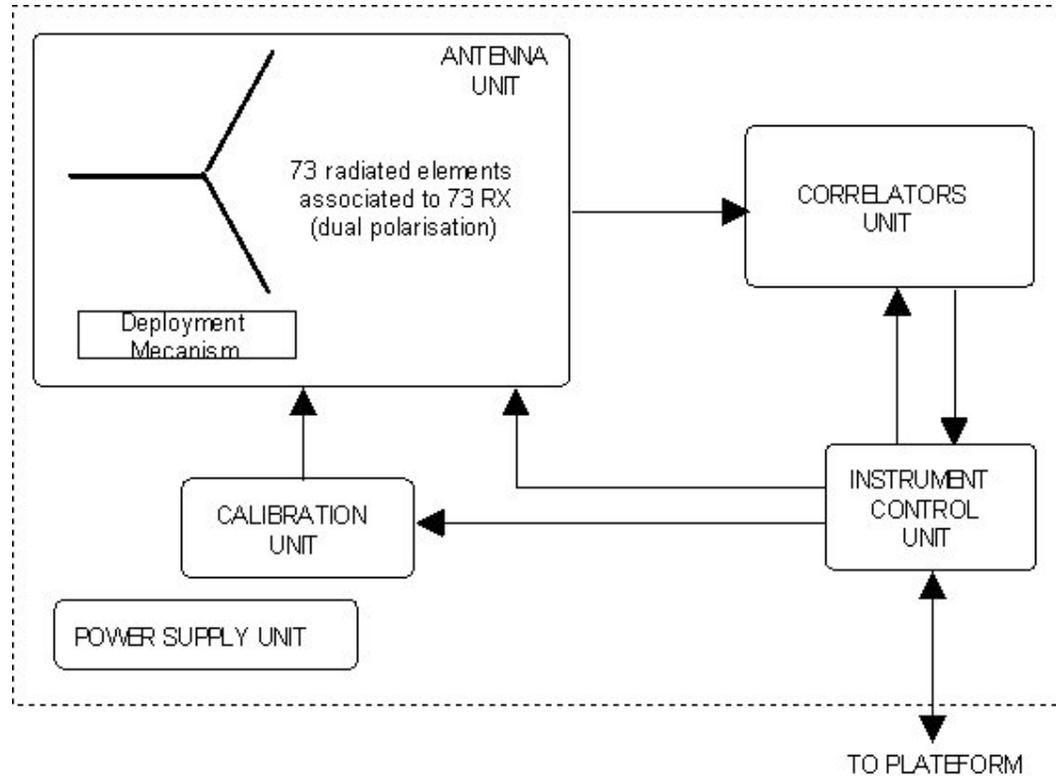
ESA  
<http://www.cesbio.ups-tlse.fr/us/indexsmos.html>



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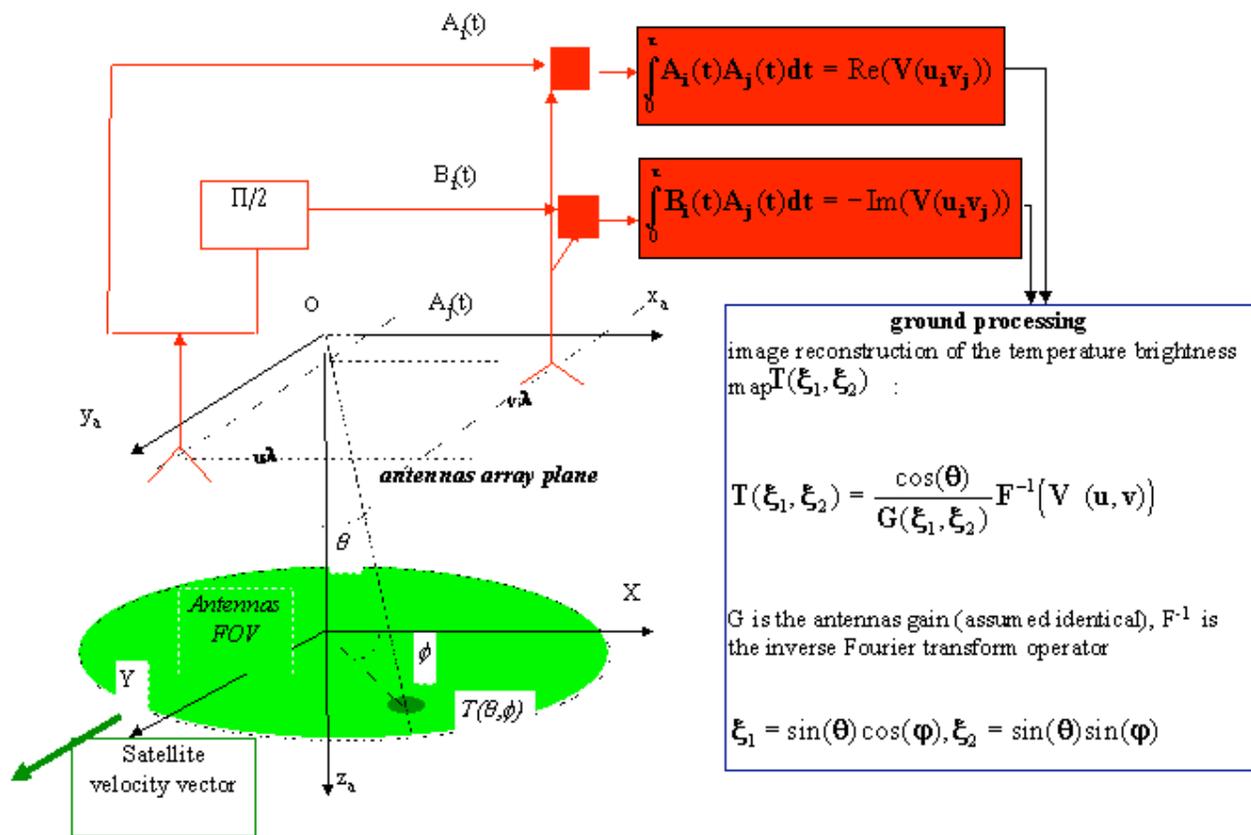
# SMOS Architecture



ESA  
<http://www.cesbio.ups-tlse.fr/us/indexsmos.html>



# 2-D STAR Principle of Operation



ESA  
<http://www.cesbio.ups-tlse.fr/us/indexsmos.html>

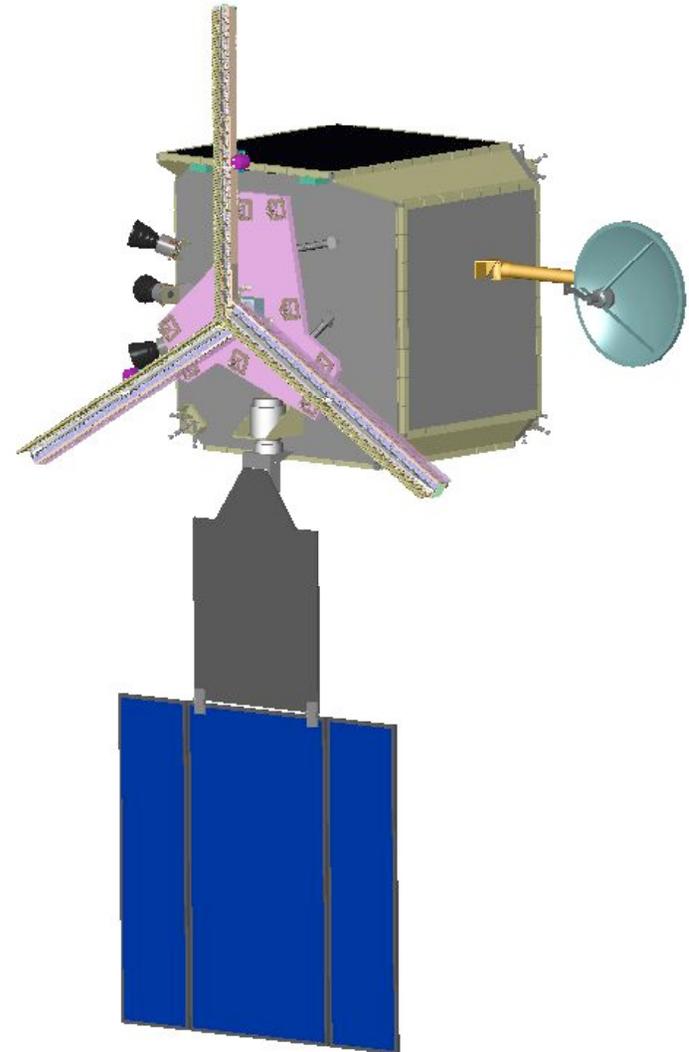
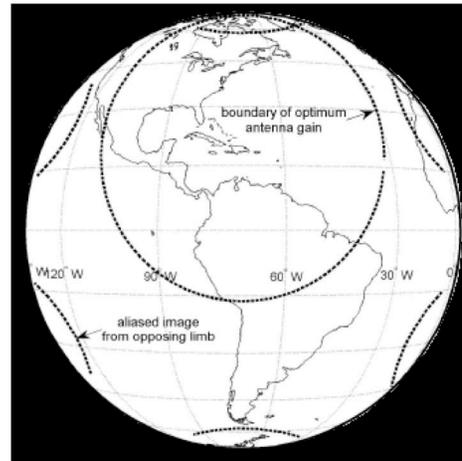


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# Future: GeoSTAR

- Dual 2-D sparse Y-arrays @ 50 GHz & 183 GHz
  - 300 + 600 receiver elements
  - 4-meter & 2-meter effective apertures
  - Single polarization
  - Min. baseline =  $4 \lambda$ 
    - ⇒ largely alias-free FOV
- Planned for “PATH” mission
  - Atmospheric T & q sounding from GEO
  - NRC “Decadal Survey” recommendation
  - Possible payload for GOES





# NRC Decadal Survey

NASA is committed to implementing the Decadal-Survey missions, but current funding of SMD/ESD dictates a stretched out schedule

Decadal Survey Mission	Mission Description	Orbit	Instrument	Rough Cost Estimate
<b>Timeframe: 2010 - 2013. Missions listed by cost</b>				
CLARREO (NASA portion)	Solar radiation, spectrally resolved forcing and response of the climate system	LEO, Precessing	Absolute, spectrally-resolved interferometer	\$200 M
SMAP	Soil moisture and freeze-thaw for weather and water cycle processes	LEO, SSO	L-band radar L-band radiometer	\$300 M
ICESat-II	Ice sheet height changes for climate change diagnosis	LEO, Non-SSO	Laser altimeter	\$300 M
DESDynI	Surface and ice sheet deformation for understanding natural hazards and climate, vegetation structure for ecosystem health	LEO, SSO	L-band INSAR Laser altimeter	\$700 M
<b>Timeframe: 2013 - 2016. Missions listed by cost</b>				
HypIRI	Land surface composition for agriculture and mineral characterization, vegetation types for ecosystem health	LEO, SSO	Hyperspectral spectrometer	\$300 M
ASCENDS	Day night, all-latitude, all-season CO <sub>2</sub> column integral for climate emissions	LEO, SSO	Midfrequency laser	\$400 M
SWOT	Ocean, lake, and river water levels for ocean and inland water dynamics	LEO, SSO	Ka-band wide swath radar C-band radar	\$450 M
GEO-CAPE	Atmospheric gas column for air quality forecasts; ocean color for coastal ecosystem health and climate emissions	GEO	High and low spatial resolution hyperspectral imagers	\$550 M
ACE	Aerosol and cloud profiles for climate and water cycle; ocean color for open ocean biogeochemistry	LEO, SSO	Backscatter lidar Multiple polarimeter Doppler radar	\$800 M
<b>Timeframe: 2016 - 2020. Missions listed by cost</b>				
LIST	Land surface topography for landslide hazards and water runoff	LEO, SSO	Laser altimeter	\$300 M
PATH	High frequency, all-weather temperature and humidity soundings for weather forecasting and SST <sup>a</sup>	GEO	MW array spectrometer	\$450 M
GRACE-II	High temporal resolution gravity fields for tracking large-scale water movement	LEO, SSO	Microwave or laser ranging system	\$450 M
SCLP	Snow accumulation for fresh water availability	LEO, SSO	Ku and X-band radars K and Ka-band radiometers	\$500 M
GRACM	Ozone and related gases for intercontinental air quality and stratospheric ozone layer prediction	LEO, SSO	UV spectrometer IR spectrometer Microwave limb sounder	\$600 M
3D-Winds (Demo)	Tropospheric winds for weather forecasting and pollution transport	LEO, SSO	Doppler lidar	\$650 M

Precipitation and All-weather Temperature and Humidity (PATH)  
Launch: 2016-2020  
Mission Size: Medium




Sea surface temperature



Temperature and humidity profiles



Constraints on models for boundary layer, cloud, and precipitation processes



More accurate, longer-term weather forecasts



Improved storm track and intensification prediction and evacuation planning



Determination of geographic distribution and magnitude of storm surge and rain accumulation

PATH	High frequency, all-weather temperature and humidity soundings for weather forecasting and SST <sup>a</sup>	GEO	MW array spectrometer	\$450 M
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= GeoSTAR!

Note: The NRC panel put PATH in the 3rd group, reflecting their perception of the maturity of the required technology. Recent developments indicate a higher level of readiness, and it may be feasible to implement PATH earlier than thought.



# GeoSTAR System Concept

## Aperture-synthesis concept

- Sparse array employed to synthesize large aperture
- Cross-correlations -> Fourier transform of Tb field
- Inverse Fourier transform on ground -> Tb field

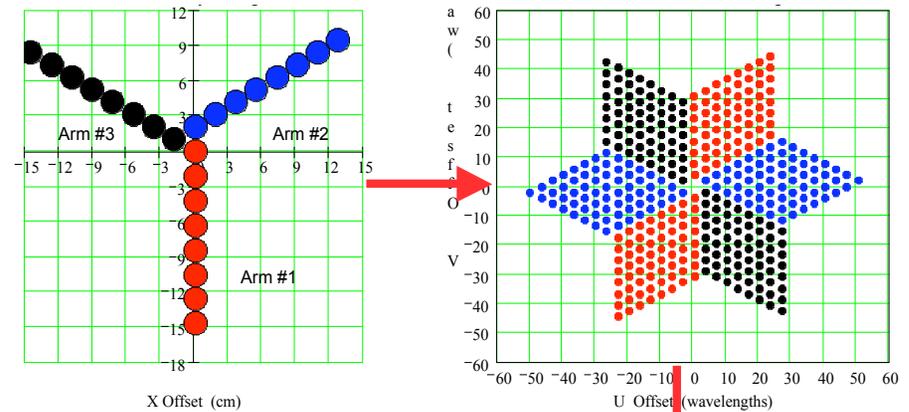
## Array

- Optimal Y-configuration: 3 sticks; N elements
- Each element is one I/Q receiver,  $3.5\lambda$  wide (2.1 cm @ 50 GHz; 6 mm @ 183 GHz!)
- Example:  $N = 100 \Rightarrow$  Pixel =  $0.09^\circ \Rightarrow$  50 km at nadir (nominal)
- One "Y" per band, interleaved

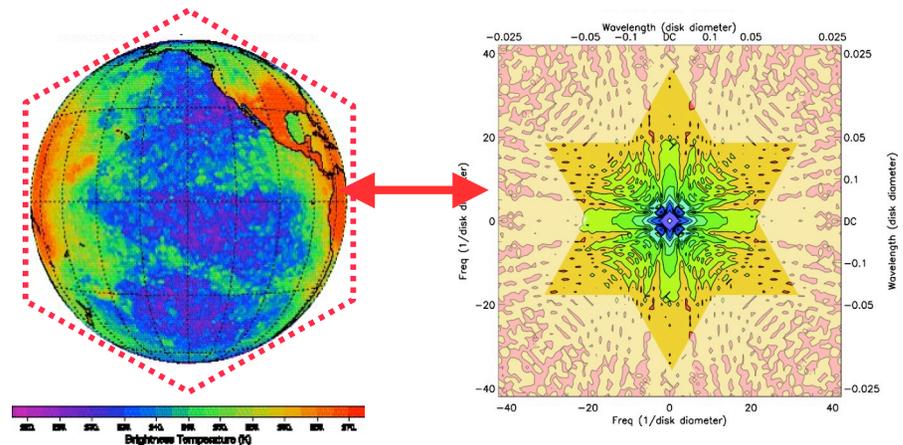
## Other subsystems

- A/D converter; Radiometric power measurements
- Cross-correlator - massively parallel multipliers
- On-board phase calibration
- Controller: accumulator -> low D/L bandwidth

Receiver array & resulting uv samples



Example: AMSU-A ch. 1





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# GeoSTAR Spatial Coverage

- **Fourier imaging features**

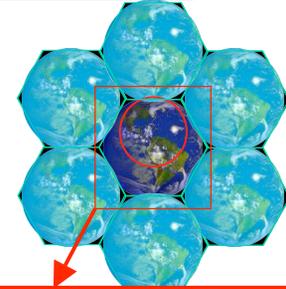
- Basic imaging area is a hexagon
- Periodic nature of Fourier series creates infinite series of secondary imaging hexagons
- Sources in secondary areas are aliased into primary area

- **Basic configuration**

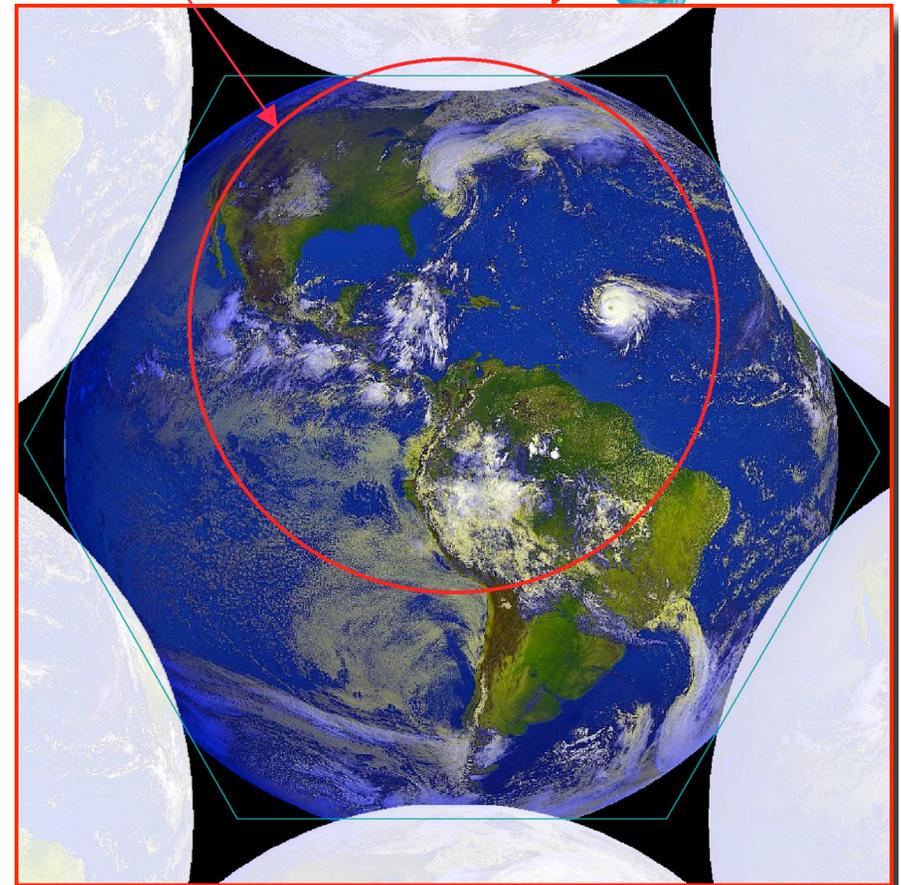
- 4-wavelength element spacing for optimal performance
- Edges of Earth then extend into secondary hexagons
- Those areas are therefore alias contaminated
- This is not a problem: unimportant areas
- Earth's limb visible in 6 sectors
  - Use this for calibration

- **Region Of Interest**

- Maximum performance near center of antenna patterns
- Pitch instrument  $\sim 3^\circ$  N for focus in Caribbean
- ROI is largely free of alias



Region of maximum sensitivity

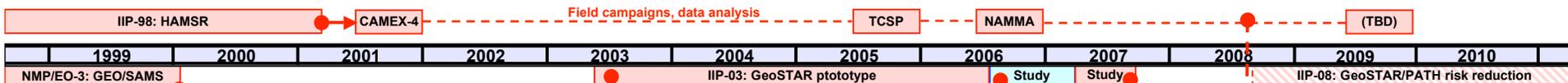




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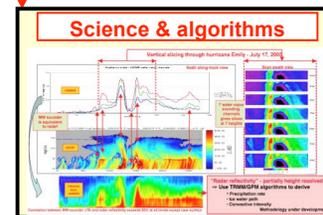
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# GeoSTAR Development History



**NOAA mission**

Year	2003
Agency	NOAA
Instrument	STAR
Frequency	183 GHz
Resolution	1 km
Swath	100 km
Altitude	700 km
Orbit	Polar
Launch	2003
End of Mission	2005



**STAR concept and key technologies developed & tested**

**Compact receivers**

**Innovative array layout**

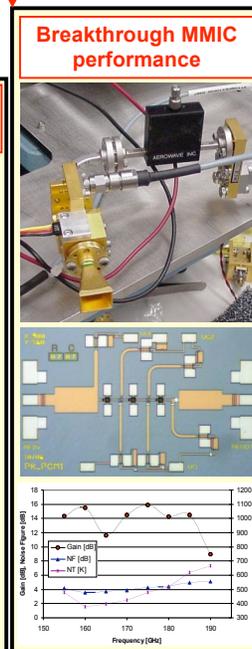
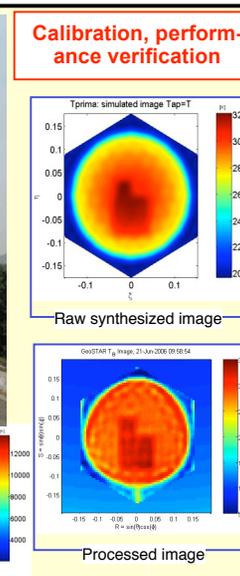
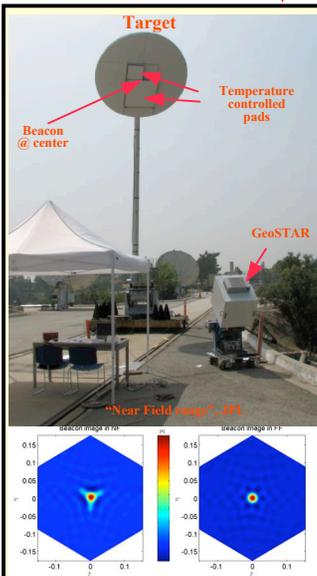
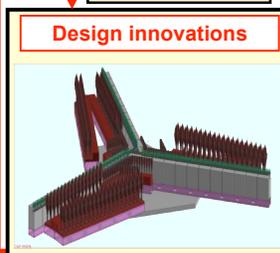
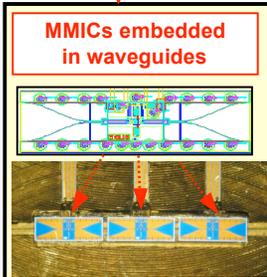
**Low-power MMICs**

**LO phase switching system: Ultrastable operation**

**Correlator:**  
• Efficient  
• Redundant  
• OK for ASICs

**Feedhorns:**  
• Low mutual coupling

**First images at 50 GHz by aperture synthesis**

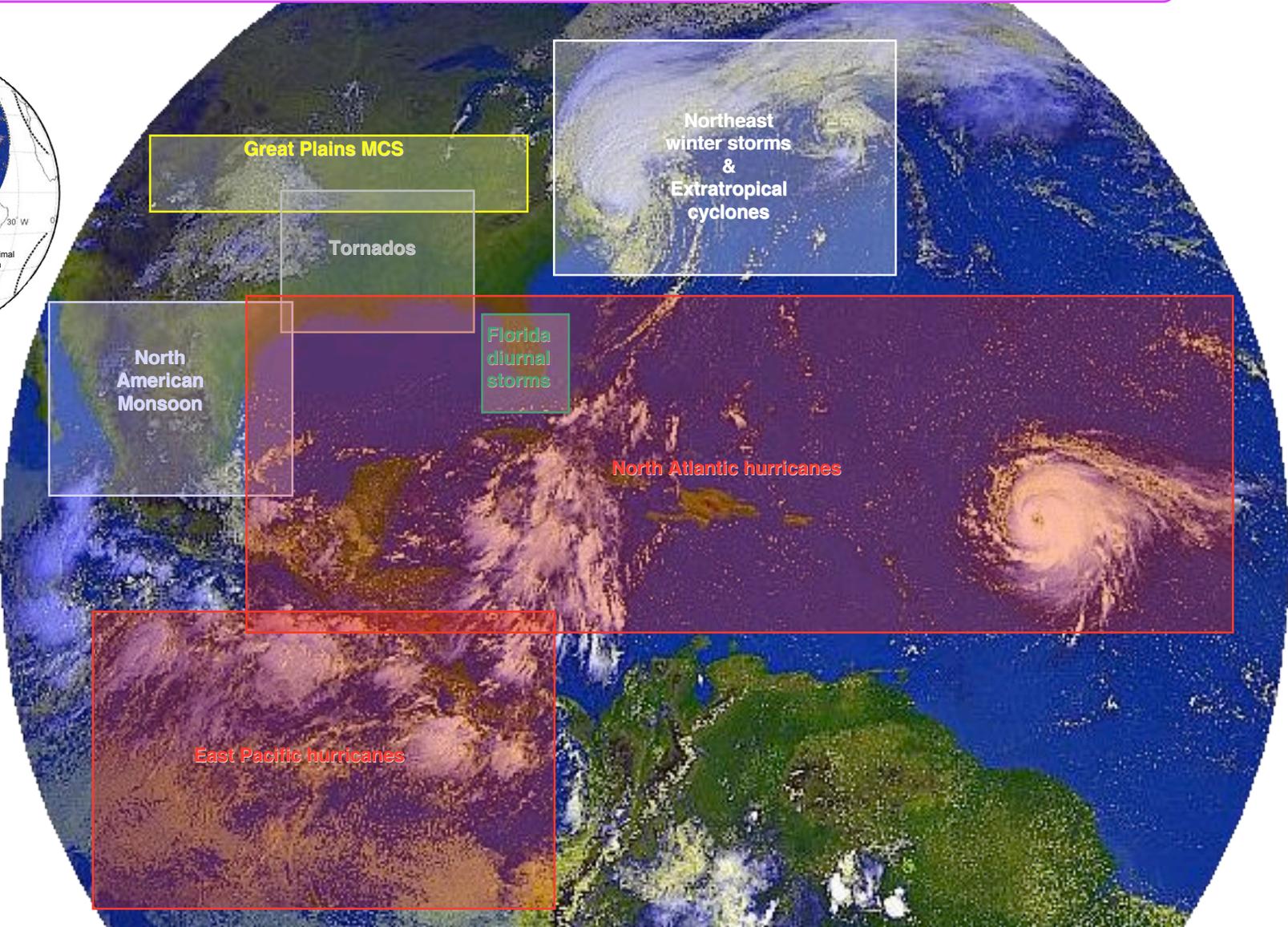
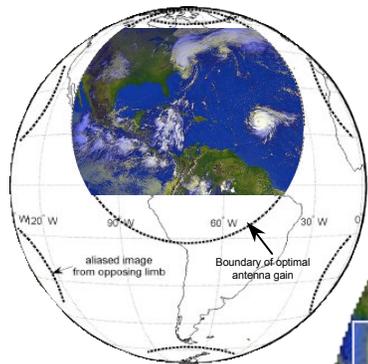




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# GeoSTAR/PATH Science Themes





# Mission Study: Feasibility of GeoSTAR

## Baseline design

*Is it possible to develop a GeoSTAR design that meets all key performance requirements?*

- Dual band, dual collinear array
  - ~ 300 receivers @ 50 GHz/6 channels  $\Rightarrow$  T-sounding @ 50 km
  - ~ 600 receivers @ 183 GHz/4 channels  $\Rightarrow$  q-sounding @ 25 km
- NEDT < 1 K (baseline reference); high-performance: < 1/3 K
- $\Delta t \sim 15$  minutes in central FOV-region

## Cost

*Is the NRC PATH cost estimate reasonable?*

- Mission: ~ \$500M (within 15% of NRC estimate)
- Payload: < \$150M

## Power

*Is it possible to run ~ 900 receivers & 600,000+ correlators with reasonable power?*

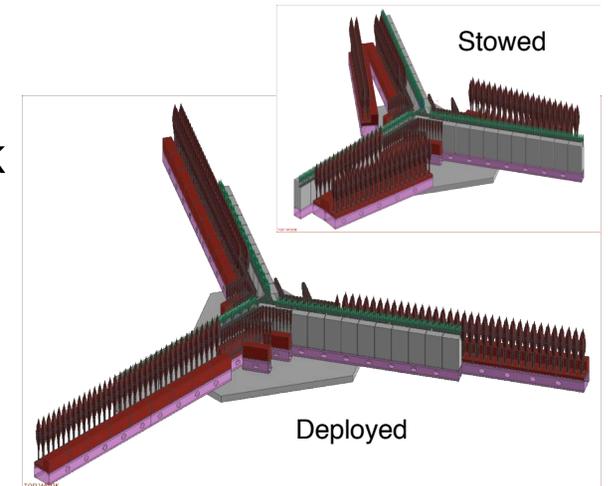
- Receivers: ~ 150 mW each  $\Rightarrow$  less than 140 W total
- Correlators: < 10  $\mu$ W per 1-bit cell (90 nm ASIC @ 100 MHz)  $\Rightarrow$  ~ 5 W total
  - High-performance option: 1.5 bits, 0.5-1 GHz, 65 nm  $\Rightarrow$  NEDT < 1/3 K @ < 50 W
- Total: ~ 250 W (+ 0-90 W solar heat load compensation)

## Mass

*Is total payload mass reasonable?*

- Receivers: ~ 100 mg each  $\Rightarrow$  less than 90 kg
- Total: ~ 230 kg

**CONCLUSION: GeoSTAR/PATH is a feasible medium-class mission**





# Technology Challenges

- **Stressing requirements**
  - Spatial resolution  $\Rightarrow$  Large number of receivers; very large number of digital processors
  - Temporal response  $\Rightarrow$  High radiometric sensitivity; parallel processing
  - Radiometric sensitivity  $\Rightarrow$  Sensitive receivers; fast “processing”
  - Limited resources  $\Rightarrow$  Reduce mass, power, complexity, cost
- **RF front-end (MMIC receivers)**
  - Low noise, reduced size, mass, *power*
  - Suitable for mass production, integration & testing
    - Array element matching
  - Large-scale module integration
    - Multiple-receiver modules integrated with apertures & structure
- **Digital back-end**
  - Digitizers & correlators
    - Large-scale chip integration (rad-hard ASICs)
  - Low power, high bandwidth
- **Other drivers**
  - Signal distribution & cabling
  - Power distribution (low-loss low-voltage DC-DC)