KISS Concurrent Engineering Exercise

Jet Propulsion Laboratory

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Jet Propulsion Laboratory, California Institute of Technology

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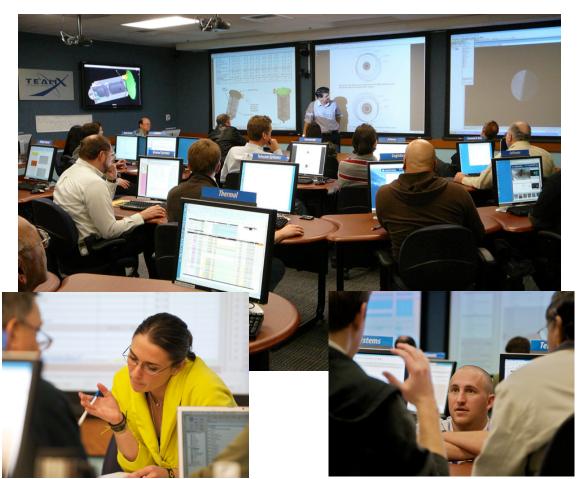


What is Team X?



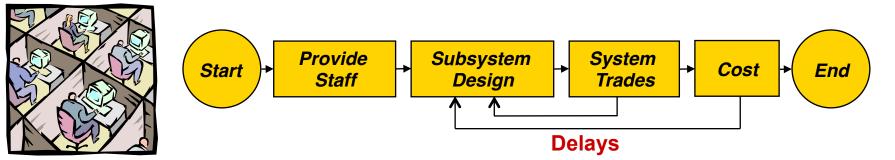
Team X is JPL's concurrent engineering team for rapid design and analysis of space mission concepts

- Developed in 1995 by JPL to reduce study time and cost
- Used for early formulation concept exploration
- Completed over 1100 studies
- Institutionally endorsed
- Emulated by many institutions





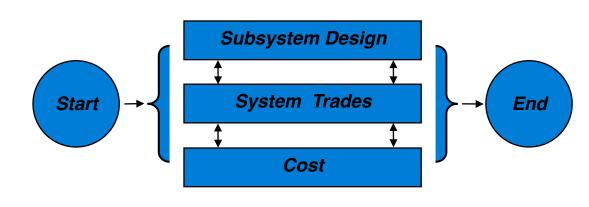
Traditional Method – Serial



Concurrent Engineering – Parallel

• Diverse specialists working in real time, in the same place, with shared data, to yield an integrated design

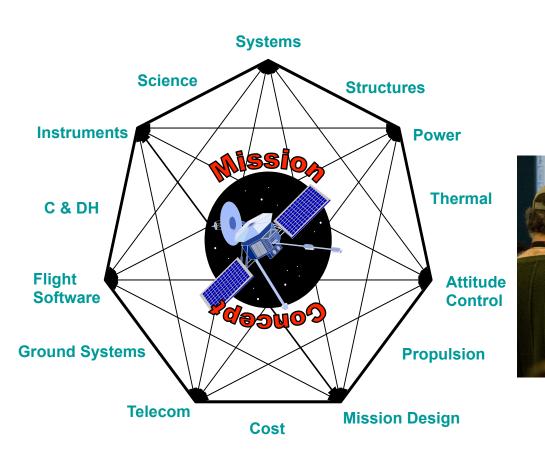






The Mission Team









Plan for Today



🗙 AM – Cubesat

- Team Study Briefing
- Guideline Review
- Review Design Issues
- Subsystem Design Work...Concurrent Design
- Power Estimate Review
- Mass Estimate Review
- Cost Estimate Review
- ▼ PM Relay Satellite
 - Review Design Issues
 - Subsystem Design Work...Concurrent Design
 - Power Estimate Review
 - Mass Estimate Review
 - Cost Estimate Review
- Risk List Review



Cubesat Design Issues



All Subsystems

- Mass, Power and Cost Est.
- Subsystem Risks
- Ops Modes & Scenarios

× ACS

- Inertia
- Torque Calculations
- Wheel and Thruster Sizing, Number and Locations
- Kickoff Stabilization
- Wheel Desaturation

× CDH

- Data Volume/Data Storage
- Bus Interface

Instruments

- Data Volume Calculations
- Orientation/Pointing Req.
- Calibration
- Instrument Operations
- Bus Interface
- Mechanical
 - Deployments
- Mission Design
 - Delta V Budget
 - Cubesat Positions and Orientation



Cubesat Design Issues (Cont'd)



Power

- Battery Sizing
- Solar Array Sizing
- Bus Voltage(s)
- Subsystem Efficiency

Propulsion

- Propellant Load
- Tank Sizing

× Science

- What Spatial Res. Is Req'd?
- Survey or Targeted? Is Tracking Req'd?
- Operations Scenarios
- Data Volume Sizing
- Software
 - Estimate Lines of Code
- × Telecom
 - UHF Link Budget
 - Downlink Data Rate
- × Thermal
 - Internal Operating Temp



Relay Design Issues



All Subsystems

- Mass, Power and Cost Est.
- Subsystem Risks
- Ops Modes & Scenarios

× ACS

- Torque calculations
- Wheel and Thruster Sizing, Number and Location
- Wheel Desaturation

× CDH

- Data Volume/Data Storage
- Bus Interface

- Ground System
 - Downlink Durations
- Payload Dispensers
 - Interfaces
- Mechanical
 - Deployments
 - Mechanisms
 - Packaging
- Mission Design
 - Delta V Budget
 - Launch Vehicle Performance
 - Trajectory



Relay Design Issues (Cont'd)



Power

- Battery Sizing
- Solar Array Sizing
- Bus Voltage(s)
- Power Sys Efficiency
- Propulsion RCS and Primary Systems
 - Prop System Types
 - Propellant Loads
 - Tank Sizing
- × Science
 - Operations Scenarios
- Software
 - Estimate Lines of Code

Telecom

- UHF Multi Link Support
- X-Band Link Budget
- X-Band Downlink Data Rate
- HGA Articulation?
- × Thermal
 - Internal Operating Temp